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ON THE CHARACTERIZATION OF JOHNSON-COOK CONSTANTS: NUMERICAL  
AND EXPERIMENTAL STUDY OF HIGH SPEED MACHINING AEROSPACE  
ALLOYS

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# **ON THE CHARACTERIZATION OF JOHNSON-COOK CONSTANTS: NUMERICAL AND EXPERIMENTAL STUDY OF HIGH SPEED MACHINING AEROSPACE ALLOYS**

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## **RÉSUMÉ**

L'industrie aéronautique souhaiterait à terme remplacer l'usinage chimique par l'usinage mécanique lequel est plus précis, plus prévisible et surtout plus écologique. En effet, les rejets issus de l'usinage chimique contiennent notamment du dioxyde de carbone et des solvants qui se dégradent difficilement dans les eaux souterraines. L'usinage mécanique permet aussi d'éviter une disposition importante de matières dangereuses et offre un meilleur recyclage des copeaux. Cependant, la maîtrise de la qualité des pièces produites par usinage mécanique, passe par la prédiction et l'optimisation du processus de coupe du métal. L'outil de simulation le plus utilisé est la modélisation par éléments finis (MÉF). La réussite et la fiabilité des modèles simulés dépendent fortement des lois décrivant le comportement thermomécanique des matériaux usinés. Parmi elles, la plus utilisée est celle de Johnson-Cook (JC), qui combine l'effet de la déformation, de la vitesse de déformation, et de la température. La détermination des paramètres constitutifs de JC pour des conditions d'usinage extrêmes (grande déformation, vitesse de déformation élevée, haute température) a longtemps été un défi majeur, mais une nécessité pour ceux qui appliquent la méthode des éléments finis pour modéliser la coupe à l'échelle de la formation des copeaux.

Cette étude a pour objectif de traiter cette problématique en tentant de mieux comprendre l'effet de la loi de comportement de JC sur la prédiction des paramètres de coupe (les efforts de coupe, les contraintes résiduelles, etc.) pour des alliages d'aluminium. Aussi dans le but de répondre aux besoins de l'industrie aéronautique, nous avons choisi des alliages d'aluminium (Al2024-T3, Al6061-T6, et Al7075-T6) couramment utilisés par celle-ci.

Ce travail de recherche est divisé en trois étapes successives.

Dans un premier lieu nous proposons une nouvelle approche d'identification des paramètres constitutifs de JC pour la coupe de métal. Celle-ci est basée sur la méthode inverse (tests d'usinage orthogonal) et la méthodologie de surface de réponse ce qui permet de générer un grand nombre de conditions de coupe pour une plage fixe de vitesse de coupe et d'avance, et de l'angle de coupe. Grâce à cette approche, nous avons pu analyser la sensibilité des paramètres constitutifs de JC à différents angles de coupe pour les trois alliages. Il a été constaté que, pour ces trois alliages cités, l'un des ensembles de paramètres constitutifs trouvés permet des prédictions plus précises de la contrainte d'écoulement par rapport à ceux rapportés dans la littérature. De plus, une étude par éléments finis en 2D de la coupe orthogonale a également montré une bonne corrélation entre les paramètres de coupe prédits

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(efforts de coupe et épaisseur de copeau) et ceux obtenus expérimentalement lors de l'utilisation des paramètres constitutifs de JC identifiés par l'approche proposée.

En second lieu, nous avons prêté une attention particulière sur l'effet de l'angle de coupe sur les paramètres constitutifs de JC et par conséquent sur la prédiction des paramètres de coupe (les efforts de coupe, la morphologie de copeaux, la longueur de contact outil-copeau). Pour cela, différents ensembles de paramètres constitutifs de JC déterminés à différents angles de coupe ( $-8^\circ$ ,  $-5^\circ$ ,  $0^\circ$ ,  $+5^\circ$ , et  $+8^\circ$ ) ont été utilisés dans un modèle numérique d'éléments finis 2D pour simuler le comportement d'usinage de l'alliage Al2024-T3. Nous avons constaté que l'ensemble de paramètres constitutifs obtenu avec un angle de coupe de  $0^\circ$  donne globalement des prédictions plus précises des paramètres de coupe comparativement aux autres angles de coupe étudiés.

Enfin, la dernière étape de cette thèse est consacrée à la prédiction des contraintes résiduelles générées dans la pièce usinée (Al2024-T3) et des températures dans l'outil de coupe (uncoated carbide). Ainsi cette fois, nous avons décidé de considérer trois ensembles en se basant sur les résultats obtenus lors de l'étape précédente avec les angles de coupe de  $-8^\circ$ ,  $0^\circ$ , et  $+8^\circ$ . Deux modèles numériques basés sur la méthode des éléments finis ont été utilisés: le premier été utilisé pour faire une analyse thermomécanique-2D pour simuler la coupe et le second pour une analyse thermique-3D pour étudier la distribution des températures. Les résultats montrent qu'une meilleure prédiction des contraintes résiduelles est obtenue lors de l'utilisation de JC à  $0^\circ$  tandis que les autres ensembles de JC à  $-8^\circ$  et à  $+8^\circ$  ont tendance à respectivement surestimer ou sous-estimer celles-ci. Concernant la température dans l'outil de coupe, afin d'en évaluer la meilleure prédiction nous avons calculé des moyennes de températures simulées dans les outils de coupe de chaque ensemble de JC étudié. Nous avons remarqué que ces moyennes sont très proches des températures mesurées expérimentalement (environ 5,5% de différence) et nous avons déduit que les ensembles de JC n'influent pas sur la prédiction des températures de coupe dans l'outil.

**Mots-clés:** usinage mécanique; loi de comportement de Johnson-Cook; MÉF; identification; méthode inverse; alliages d'aluminium.

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## **ABSTRACT**

The aerospace industry would eventually replace chemical machining by mechanical machining which is more accurate, more predictable and more ecological. In fact, the discharges in the case of chemical machining contain especially carbon dioxide and solvents that are difficult to degrade in groundwater. The mechanical machining also avoids an important quantity of hazardous substances and provides better chips recycling. However, the control of mechanical machined parts quality goes through the prediction and the optimization of the metal cutting processes. The most attractive computational tool to predict and optimize metal cutting processes is the finite element modeling (FEM). The success and the reliability of any FEM depend strongly on the constitutive laws which describe the thermo-mechanical behavior of the machined materials. The most commonly used one is that of Johnson and Cook (JC) which combines the effect of strains, strain rates, and temperatures. The determination of the material constants of JC under high strains, strain rates, and temperatures during machining conditions has long been a major challenge but a necessity for those who apply finite element modeling techniques in machining processes at the chip formation scale.

This study aims at treating this subject in order to better understand the effect of the JC constitutive law on the prediction of cutting parameters (cutting forces, residual stresses, etc.) for aluminum alloys. In addition, in order to meet the interests of aerospace industry, three aluminum alloys (Al2024-T3, Al6061-T6 and Al7075-T6) commonly used in aircraft applications have been selected.

This research work is divided into three consecutive steps.

Firstly, a new approach to identify the material constants of JC for metal cutting is proposed. The approach is based on the inverse method (orthogonal machining tests) and the response surface methodology which allows generating a large number of cutting conditions within fixed ranges of cutting speed, feed rate, and rake angle. Based on this approach, the sensitivity of the material constants of JC to the rake angle for the three alloys was analysed. It was found that, for these three alloys, one set of the material constants obtained from the proposed approach predicts more accurate values of flow stresses as compared to those reported in the literature. Moreover, a 2D FEM investigation of the orthogonal cutting also showed a good agreement between the predicted cutting parameters (cutting forces and chip

thickness) and experimental ones when using the material constants obtained by the proposed approach.

Secondly, a specific focus was put on the influence of the rake angle on the material constants of JC and hence on the predicted cutting parameters (cutting forces, chip morphology, and tool-chip contact length). To achieve this goal, different sets of JC constants obtained at different rake angles ( $-8^\circ$ ,  $-5^\circ$ ,  $0^\circ$ ,  $+5^\circ$ , and  $+8^\circ$ ) were used in conjunction with a 2D finite element model to simulate the machining behavior of Al2024-T3 alloy. It was found that the material constants set obtained with  $0^\circ$  rake angle gives overall more accurate predictions of the cutting parameters as compared to other studied sets.

Finally, the last step of this study is devoted to the prediction of induced residual stresses within the machined workpiece (Al2024-T3) and the temperature of the cutting tool (uncoated carbide). Three sets of JC based on the results obtained from the previous step with rake angles of  $-8^\circ$ ,  $0^\circ$ , and  $+8^\circ$  were considered. Two finite element models were used; a 2D thermo-mechanical simulation to simulate chip formation and a 3D pure thermal analysis to obtain the temperature distribution. The results show that a better prediction of the residual stresses is obtained with JC at  $0^\circ$  while the other sets of JC at  $-8^\circ$  and  $+8^\circ$  tend to overestimate or underestimate the measured residual stresses, respectively. As far as the temperature of the cutting tool is concerned, the average values of the predicted temperatures of the cutting tool for each studied set of JC was considered in order to evaluate the best prediction. Based on these average values, the effect of the three sets of JC was not significant since the difference between the measured temperatures and the predicted average ones are less than 5.5% with the three cutting conditions.

**Keywords:** mechanical machining; Johnson-Cook constitutive law; FEM; identification, inverse method, aluminum alloys.

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## **LIST OF ABBREVIATIONS**

BUE	Built-up edge
CCD	Central composite design
CMM	Coordinate measuring machine
DOC	Depth of cut
EDM	Electrical Discharge Machine
FE	Finite element
FEA	Finite element analysis
FEM	Finite element modeling
HSM	High speed machining
IABRSM	Inverse approach based on response surface methodology
JC	Johnson-Cook
M&E	Moore and Evans
RSM	Response surface methodology
SHBT	Split-Hopkinson bar technique
XRD	X-ray diffraction



## LIST OF SYMBOLS AND UNITS OF MEASUREMENTS

$a, b, d$	Hole position parameters (mm)
$A$	Yield strength coefficient (MPa)
$B$	Hardening modulus (MPa)
$C$	Strain rate sensitivity coefficient (-)
$\mathbf{C}$	Heat capacity matrix (J/°C)
$C_p$	Specific heat of work material (J/Kg °C)
$D$	Damage parameter (MPa)
$DOC$	Depth of cut (mm)
$E$	Young's modulus (GPa)
$e$	Thermal expansion coefficient (1/°C)
$F_c, F_f$	Tangential and thrust force components (N)
$F_e$	Externally applied force vector (N)
$f$	Feed rate (mm/rev)
$\mathbf{f}$	Nodal point residual force vector (N)
$G$	Residual function (N)
$h$	Thickness of the primary shear zone (mm)
$h_{int}$	Interface heat transfer coefficient (N/s mm °C)
$h_{conv}$	Convection heat transfer coefficient (N/ s mm °C)
$I$	Identity matrix (-)
$K_{th}$	Thermal conductivity (W/m °C)
$K_{chip}$	Shear flow stress in the chip at the tool-chip interface (MPa)
$\mathbf{K}_c$	Heat conduction matrix (W/°C)

$k$	Independent variables
$K$	Stiffness matrix (kg/s <sup>2</sup> )
$K_{ds}$	Correction coefficient at depth “d” for the step “s” (-)
$K_{cor}$	Correction matrix (-)
$\dot{K}_{ds}$	Modified correction matrix (-)
$l_c$	Tool chip contact length (μm)
$m$	Thermal softening coefficient (-)
$m_f$	Shear friction coefficient (-)
$M$	Masse matrix (Kg)
$n$	Hardening coefficient (-)
$N$	Number of data
$\mathbf{Q}$	Heat flux vector (W)
$R^2$	Coefficient of determination (-)
$R_{adj}^2$	Adjusted coefficient of determination (-)
$R_{in}, R_{out}, R_m$	Inner, outer, and actual measurement radius (mm)
$T$	Temperature of the work material (°C)
$\mathbf{T}$	Vector of nodal point temperatures (°C)
$T_{melt}$	Melting point of the work material (°C)
$T_{room}$	Room temperature (°C)
$T_{AB}$	Average temperature on the primary shear plane (°C)
$\dot{\mathbf{T}}$	Vector of nodal temperature rates (°C/s)
$t_c$	Chip thickness (mm)
$u$	Displacement vector (m)

$\ddot{u}$	Acceleration vector (m/s <sup>2</sup> )
$V_c$	Cutting speed (m/min)
$\mathbf{v}_0$	Initial guess velocity (m/s)
$W$	Width of cut (mm)
$x_i, x_j$	Machining parameters (-)
$y$	Response surface (-)
$\alpha$	Tool rake angle (°)
$\beta$	Angle of the inclination of the primary collimator (°)
$\beta_0, \beta_i, \beta_{ii}, \beta_{ij}$	Regression coefficients (-)
$\gamma$	Deceleration coefficient (-)
$2\theta$	Bragg angle (°)
$\bar{\varepsilon}_{AB}$	Plastic equivalent strain on the primary shear plane (-)
$\dot{\bar{\varepsilon}}_{AB}$	Equivalent strain rate on the primary shear plane (s <sup>-1</sup> )
$\dot{\bar{\varepsilon}}_0$	Reference strain rate (s <sup>-1</sup> )
$\varepsilon_f$	Effective strain (-)
$\bar{\sigma}_{AB}$	Equivalent flow stress at the primary shear zone (MPa)
$\sigma_n$	Normal stress at tool-chip interface (MPa)
$\sigma_1$	Maximum principal stress (MPa)
$\sigma_{rc}, \sigma_{\theta c}, \sigma_{zc}$	Corrected stress in radial, tangential, and axial directions (MPa)
$\sigma_{\theta m}, \sigma_{zm}$	Measured stress in tangential and axial directions (MPa)
$\sigma_{m_s}$	Stress measured in the direction of interest on the top of the layer “s” (MPa)
$\sigma_{m_{s+1}}$	Stress measured in the direction of interest on the top of the layer “s+1” (MPa)
$(\sigma_d)_s, (\sigma_d)_{s-1}$	Stress at depth “d” after removing layers “s” and “s-1” (MPa)

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$\sigma c_d$	Residual stress corrected for material removal at depth “d” (MPa)
$\sigma m_d$	Residual stress measured at depth “d” without correction (MPa)
$(\Delta\sigma_d)_s$	Local stress variation at depth “d” after removal step “s” (MPa)
$\sigma c$	Column vectors of the corrected stresses (MPa)
$\sigma m$	Column vectors of the measured stresses (MPa)
$\sigma avg_s$	Average of two measured stresses on both side of the removed layer “s”(MPa)
$\sigma_{AVG}$	Column vector containing all the average measured stresses (MPa)
$\tau_s$	Frictional shear stress at the tool-chip interface (MPa)
$\mu$	Coefficient of friction (-)
$\nu$	Poisson ratio (-)
$\rho$	Density (kg/m <sup>3</sup> )
$\Delta t$	Time step (s)
$\Delta t_{cr}$	Critical time step (s)
$\Delta v$	Velocity correction term (m/s)
$\epsilon$	Experimental error of the observations (-)
$\phi$	Shear angle (°)
$\eta$	Ratio of the heat flowing into the workpiece (-)



## INTRODUCTION

Nowadays, the aeronautical industry is more and more interested in the use of conventional machining rather than the chemical machining in order to comply with the environmental protection laws and regulation and to enhance the functional behavior of the machined structural components.

The use of light weight structural materials with high strength is always in demand from the manufacturing industries. Aluminum alloys such as Al2024-T3, Al6061-T6, and Al7075-T6, which belong to this category, are widely utilized in the aeronautical industry. However, tendency of built-up edge (BUE) formation and unfavorable chips (such tangled and ribbon chips) are often encountered during the machining of these alloys which can affect the surface finish, dimensional tolerances and tool life. In order to overcome these drawbacks, cutting fluids are often used. However, the coolants result in ecological and economic problems, consequently, there is an interest in dry high speed machining (HSM) to make this metal cutting as a green process as possible (Sreejith et Ngoi, 2000). Moreover, HSM has been reported as high material removal rates, enhancement in product quality as well as surface finish, and elimination of BUE and burrs (Fallböhmer *et al.*, 2000; Rao et Shin, 2001).

In fact, machining is one of the most manufacturing processes widely used in industry. Machining is defined as the process in which unwanted material is carried away gradually from a workpiece. Cutting is a term that describes the formation of a thin layer, called chip, via the interaction of a wedge-shaped tool with the surface of the workpiece, given that there is a relative motion between them (Markopoulos, 2012). In most practical operations, the cutting tool is three-dimensional and geometrically complex. For this reason, the two-dimensional orthogonal cutting is used to explain the basic mechanism of metal cutting. In orthogonal cutting, which is the subject of the current research, the cutting edge of the tool is perpendicular to the cutting direction (primary motion), as shown in Figure 0-1. In addition,

orthogonal cutting could be assumed as plane strain condition if the following considerations are respected: (1) the cutting edge is straight and sharp and wider than the width of the machined workpiece. (2) the cutting edge of the tool is perpendicular to the cutting velocity. (3) the width of cut is larger than or equal to 10 times the uncut chip thickness.

Therefore, two cutting forces (cutting force  $F_c$  and thrust force  $F_f$ ) are identified in orthogonal cutting configuration, as shown in Figure 0-2. From the experimental point of view, the orthogonal cutting test can usually be carried out with two set-ups. In the case of turning a disk-shaped workpiece (Figure 0-2 (a)), the straight cutting edge is set parallel to primary rotation axis of the workpiece and is moved linearly towards the center of the disk (feed motion). Since the feed motion results in reduction of disk diameter, the cutting speed is kept constant by increasing the rotation speed. Thin tube turning is also used for orthogonal cutting tests, as shown in Figure 0-2 (b). Here, the cutting speed changes over the cutting edge. By choosing a tube with large diameter and thin wall thickness, the changing in cutting speed could be minimized. The literature review illustrates that the orthogonal cutting tests conducted using a disk-shaped workpiece represents more truly plane strain problem rather than the commonly used thin tube turning experiments due to the fact that chips curl always sideward and out of plane (Ee *et al.*, 2005). Although these two set-ups satisfy the consideration mentioned above, nevertheless, they have two major disadvantages as the residual stresses analysis is considered: the first one is related to the choice of a machined surface zone which is representative of the cutting test and the second one is the effect of cutting passes during the machining tests. Recently (Ducobu *et al.*, 2015) presented a simple set-up to perform orthogonal cutting experiments using a standard milling machine as a planning machine. In this set-up, the workpiece is inserted to the spindle and the tool is mounted on the tool holder. The cutting is achieved by moving the workpiece towards the stationary cutting tool at a cutting speed  $V_c$ , as shown in Figure 0-3. The main drawback of this set-up is that the maximum cutting speed is limited to the maximum feed rate of the machine; therefore, this set-up cannot be used in HSM of aluminum alloy, for example. In addition, great care must be taken to position the cutting tool at a distance sufficient far from the workpiece in order to assure that the required cutting speed is reached before cutting.

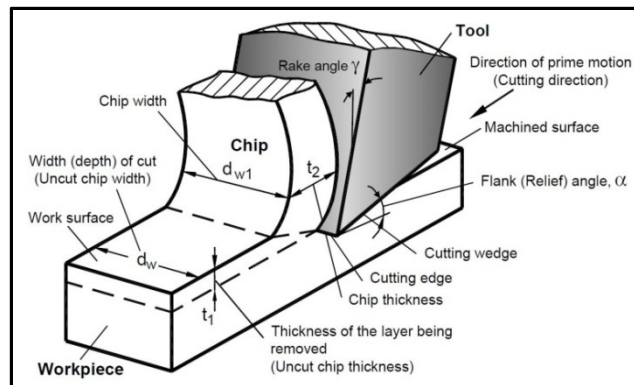


Figure 0-1 Basic terms in orthogonal cutting  
(Astakhov, 2010)

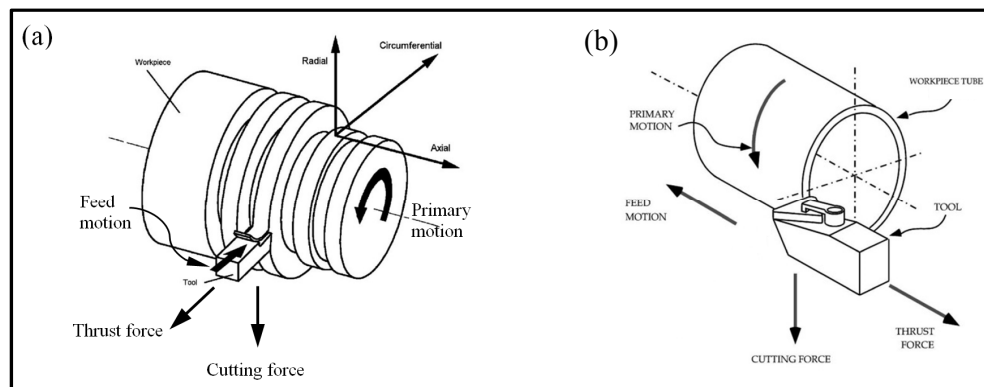


Figure 0-2 Configuration of the orthogonal cutting test and the direction of the cutting forces (a) disk-shaped workpiece (Umbrello *et al.*, 2007b)  
(b) thin tube turning (Özel, 2003)

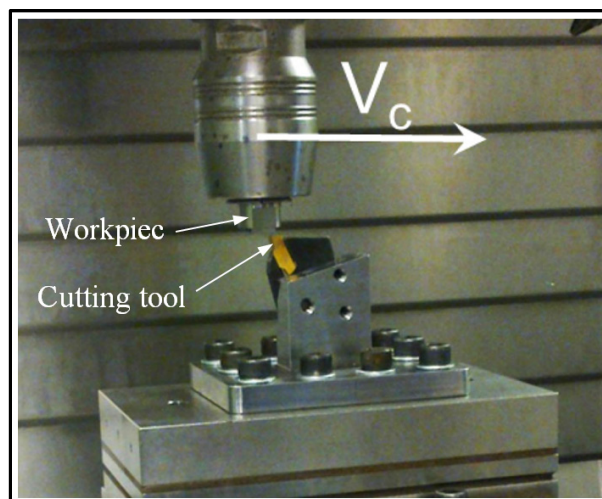


Figure 0-3 Orthogonal cutting configuration  
(Ducobu *et al.*, 2015)

Figure 0-4 shows the chip formation obtained from FEM of orthogonal machining. As the wedge-shaped tool penetrates into the workpiece, the metal ahead of the tool tip undergoes a very high plastic deformation and it is sheared over the primary shear zone to form a chip. This chip (sheared material) slides up the tool face and is partially deformed under high normal stresses and friction resulting in a secondary deformation zone in which high temperature is generated. Tertiary shear zone is created due to the friction between the flank face of the tool and the newly machined surface. This friction area has no effect on the chip formation but it influences the machined surface.

Extreme conditions are encountered during machining tests which lead to different thermo-mechanical loads in the shear zones. Consequently, different types of chip forms are obtained. These chip forms were classified based on their geometrical appearance as depicted in Figure 0-5. Another possible classification was introduced in (Grzesik, 2008). In this case, the chips were classified into continuous, segmented, elemental, and discontinuous chips (see Figure 0-6). This classification is based on material deformation and relevant fracture mechanisms resulting from the interaction between cutting conditions and the workpiece properties. A discontinuous chip formation (Figure 0-6 (a)) happens when fracture occurs before complete chip plastic deformation takes place. An elemental chip (Figure 0-6 (b)) is characterized by variations in chip thickness in periodic manner formed under high speed and hard machining conditions. In segmented chips (Figure 0-6 (c)), the chip is characterized by areas having intense shear deformation (shear bands) separated by other areas with relatively lower deformation. In fact, under certain cutting conditions, the plastic strain rates become high enough to generate considerable heat in the primary shear zone which cannot rapidly be dissipated to the rest of workpiece material. This results in a quasi-adiabatic condition which causes material thermal softening (Xie *et al.*, 1996). As the cutting process continues, the local strain increases until an instantaneous shearing takes place (Jawahir et Van Luttervelt, 1993). However, the explanation of segmented chip formation by adiabatic shear theory is not unanimously accepted and another explanation of segmented chip formation based on fracture theory is reported in the literature (Vyas et Shaw, 1999). Continuous chip formation

(Figure 0-6 (d)) takes place when the chip formation occurs without fracture on the shear plane.

It is worth pointing out that the classification of chip forms is highly important for the cutting machining modeling. Although the existence of various analytical and numerical models of chip formation, it is always difficult to predict an accurate chip shape generated for a set of cutting conditions. This could explain why most research work is limited to the modeling of the continuous chip formation.

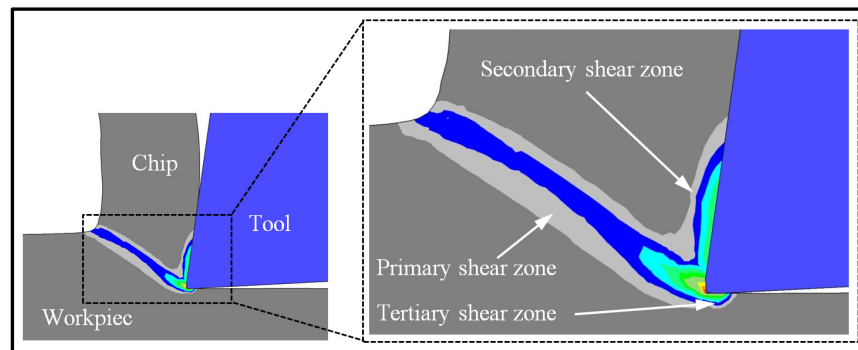


Figure 0-4 Deformation zones in orthogonal machining




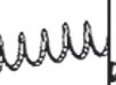










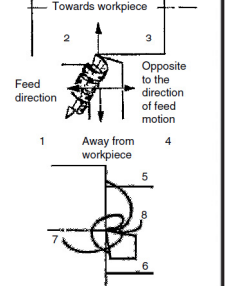




1. RIBBON CHIPS	2. TUBULAR CHIPS	3. SPIRAL CHIPS	4. WASHER-TYPE HELICAL CHIPS	5. CONICAL HELICAL CHIPS	6. ARC CHIPS	7. ELEMENTAL CHIPS	8. NEEDLE CHIPS
							
1.1. Long	2.1. Long	3.1. Flat	4.1. Long	5.1. Long	6.1. Connected		
							
1.2. Short	2.2. Short	3.2. Conical	4.2. Short	5.2. Short	6.2. Loose		
							
1.3. Snarled	2.3. Snarled		4.3. Snarled	5.3. Snarled			

Figure 0-5 Classification of chip forms according to ISO 3685-1977 (E)  
Source: (Grzesik, 2008)

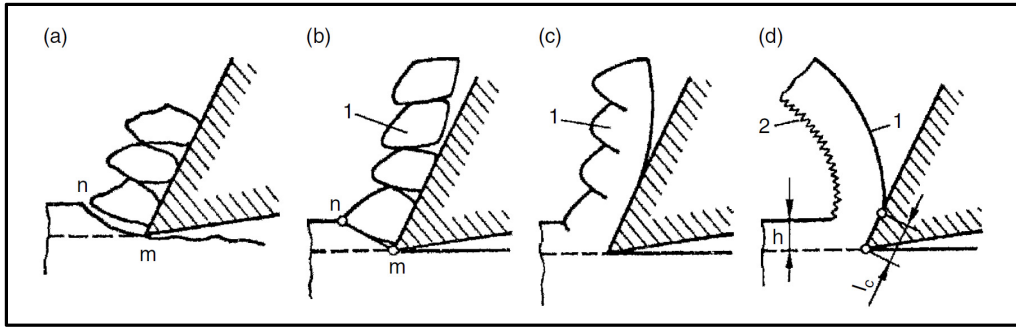


Figure 0-6 Chip formation forms:(a) discontinuous, (b) elemental, (c) segmented, (d) continuous (Grzesik, 2008)

The modeling of machining processes is highly important. It provides an understanding of the physics involved in the chip formation mechanism which in turns help design new geometry cutting tools, develop new machining alloys, and achieve effective optimization. Consequently, the traditional trial and error approach could be avoided.

Over the last decades, several analytical models of chip formation have been proposed by many researchers. The most widely known model for cutting is the shear plane developed by Ernst et Merchant (1941). In this model, the continuous chip formation is generated by a shearing process on a thin plane (plane AB in Figure 0-7), called shear plane. The shear stress is assumed to be independent of the shear angle and is distributed uniformly along the shear plane. The cutting velocity  $V_c$  is instantaneously changed to the chip velocity  $V_F$  across the shear plane. Figure 0-7 shows the condensed force diagram defining the relationship between the cutting force components during cutting. In the shear plane model, cutting force  $F_c$  and thrust force  $F_T$  are determined if the shear angle  $\phi$ , friction angle  $\zeta$ , rake angle  $\alpha$ , shear stress  $\tau_{AB}$ , and uncut ship thickness  $t_1$ , and the depth of cut  $w_1$  are known.

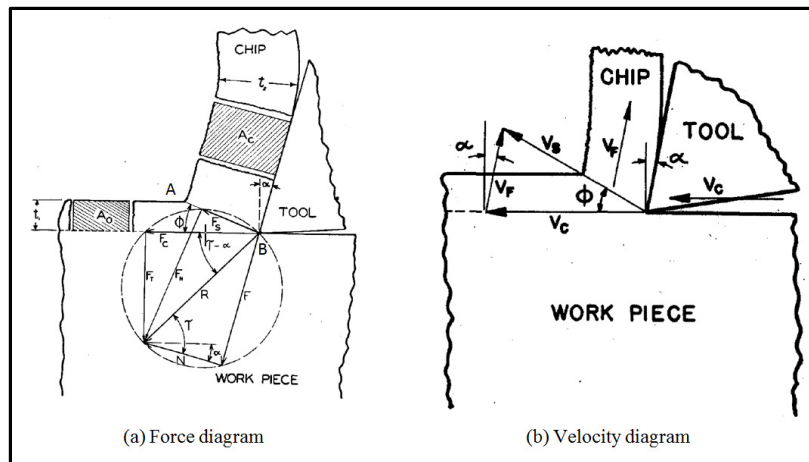


Figure 0-7 Shear plane model  
Source: (Merchant, 1945)

Based on the assumption that the material will choose to shear at an angle that minimizes the required energy, the shear angle, angle between the shear plane and the cutting direction, is given by:

$$\phi = \frac{\pi}{4} + \frac{\alpha}{2} - \frac{\zeta}{2} \quad (0-1)$$

Although there is a lack of agreement with experiment, the shear plane model is considered as a reference for other models that followed.

Lee et Shaffer (1951) developed a more advanced model based on the theory of slip line field to predict cutting forces, chip thickness, and shear angle from tool geometry, the friction coefficient, and the yield stress of the workpiece material. Similar to the shear plane model, the plastic deformation is assumed to take place on the shear plane AB, but the plastic field is extended above this plane to form a triangular plastic zone, as shown in Figure 0-8. The shear angle predicted by this model is given by:

$$\phi = \frac{\pi}{4} + \alpha - \zeta \quad (0-2)$$





line and parallel-sided shear zone theory. The average strain rate is modeled as a function of a constant shear velocity and the length of the shear plane. Oxley model permits the prediction of the cutting forces under the effect of the flow stress of the workpiece. In this model, two plastic zones, namely primary zone and secondary zone, are considered, as depicted in Figure 0-9. The average temperatures in the primary zone and at the tool-chip interface, due to the heat generation by the plastic deformation, were also derived.

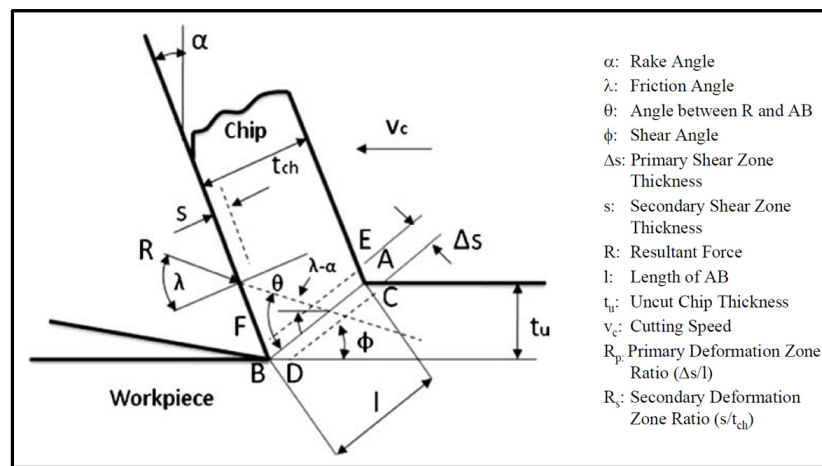


Figure 0-9 Parallel-sided shear zone model  
 Source: (Pittalà et Monno, 2010)

The above mentioned analytical models provide useful insight into the mechanics of the metal cutting.

More promising approach for studying metal cutting is provided by numerical techniques such as the finite element modeling (FEM). The flexibility of the finite element method allows it to deal with large deformation, strain rate effect, tool-chip contact and friction, local heating and temperature effect, different boundary and loading conditions, and other phenomena encountered in metal cutting problems (Shet et Deng, 2003).

Figure 0-10 outlines the main input parameters for FEM machining simulation. One of the most important governing parameter in any cutting simulation is the use of an accurate

constitutive law model which represents the material behaviour especially at the extreme conditions that exist in the shear zones (Childs, 1997; Sartkulvanich *et al.*, 2005a; Shi, 2011).

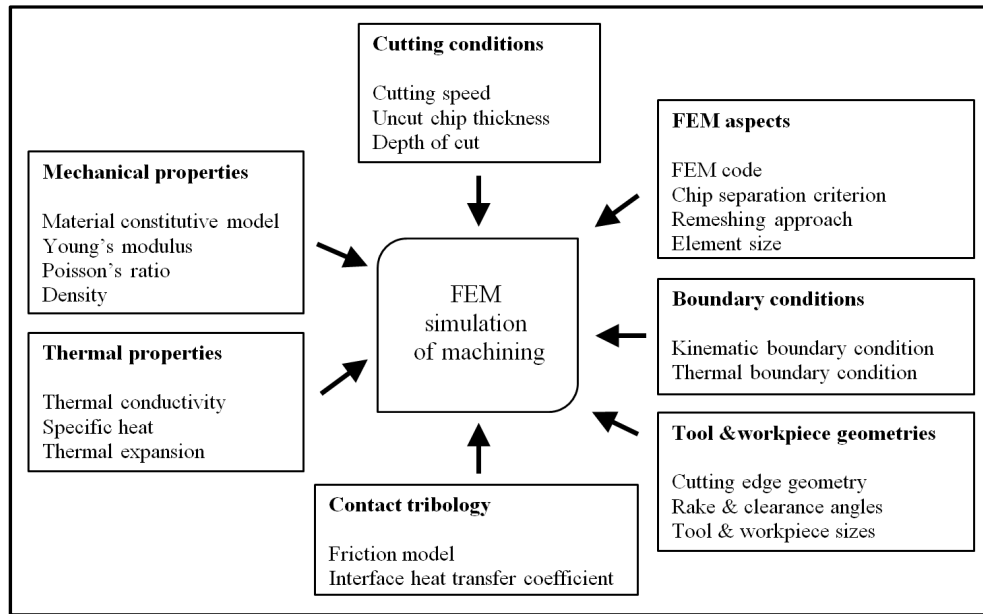


Figure 0-10 Main input parameters for FEM machining simulation

Several constitutive law models that are adopted for machining simulation have been proposed to reproduce the thermo-mechanical effects involved in metal cutting, as listed in Table 0-1. Among these constitutive laws, the Johnson-Cook (JC) model has been widely used for machining simulation because it represents adequately the material flow stress of several metallic materials in terms of their strains, strain rates, and temperatures. Moreover, this constitutive law, available in many finite element codes, has been successfully used with aluminum alloy to predict the flow stress in conditions similar to metal cutting (Jaspers et Dautzenberg, 2002).

However, different constitutive model constants for the same material could be found in the literature which can affect significantly the predicted results of the machining process such as cutting forces, chip morphology, temperatures, tool wear, and residual stresses. These discrepancies could be attributed, principally, to the different methods used for the determination of the constitutive model constants. The literature review illustrates that most

common experimental methods used to identify constitutive law models are static tests (tensile, compression), dynamic tests (Split-Hopkinson bar technique; Taylor test), and inverse method (machining test). Sartkulvanich *et al.* (2005b) have attested that, in metal cutting simulation, material constants should be obtained at high strain rates (up to  $10^6 \text{ s}^{-1}$ ) temperatures (up to  $1000 \text{ }^\circ\text{C}$ ), and strains (up to 4); therefore, the inverse method (machining test) which is conducted under these conditions has proven to be effective as a characterization test (Umbrello *et al.*, 2007b).

As a result, a new approach to identify the material constants of the constitutive law based on orthogonal machining tests is investigated and aimed to provide more reliable material constants that can be used in FEM machining simulation.

Table 0-1 Constitutive law models for machining simulation

Constitutive law models	Constitutive law equations	constants	References
Johnson-Cook	$\bar{\sigma} = [A + B (\bar{\epsilon})^n] \left[ 1 + C \ln \left( \frac{\dot{\bar{\epsilon}}}{\dot{\bar{\epsilon}}_0} \right) \right] \left[ 1 - \left( \frac{T - T_{room}}{T_{melt} - T_{room}} \right)^m \right]$	A, B, C, m, n	(Johnson et Cook, 1983)
Power law	$\bar{\sigma} = \sigma_0 \left( \frac{\bar{\epsilon}}{\bar{\epsilon}_0} \right)^n \left( \frac{\dot{\bar{\epsilon}}}{\dot{\bar{\epsilon}}_0} \right)^m \left( \frac{T}{T_0} \right)^{-\nu}$	$\sigma_0$ , m, n, $\nu$	(Shi et Liu, 2004)
Vinh	$\bar{\sigma} = \sigma_0 (\bar{\epsilon})^n \left( \frac{\dot{\bar{\epsilon}}}{\dot{\bar{\epsilon}}_0} \right)^m \exp \left( \frac{G}{T} \right)$	$\sigma_0$ , m, n, G	(Vinh <i>et al.</i> , 1979)
Zerilli-Armstrong	For b.c.c. metal $\bar{\sigma} = C_0 + C_1 \exp \left( -C_3 T + C_4 T \ln \left( \frac{\dot{\bar{\epsilon}}}{\dot{\bar{\epsilon}}_0} \right) \right) + C_5 (\bar{\epsilon})^n$ For f.c.c. metal $\bar{\sigma} = C_0 + C_2 (\bar{\epsilon})^{1/2} \exp \left( -C_3 T + C_4 T \ln \left( \frac{\dot{\bar{\epsilon}}}{\dot{\bar{\epsilon}}_0} \right) \right)$	$C_0, C_1, C_2, C_3, C_4$	(Zerilli et Armstrong, 1987)
Oxley	$\bar{\sigma} = \sigma_0(T, \dot{\bar{\epsilon}}) (\bar{\epsilon})^{n(T, \dot{\bar{\epsilon}})}$	$\sigma_0(T, \dot{\bar{\epsilon}})$ , $n(T, \dot{\bar{\epsilon}})$	(Oxley et Young, 1989)
Marusich	$\left( 1 + \frac{\dot{\bar{\epsilon}}}{\dot{\bar{\epsilon}}_0} \right) = \left( \frac{\bar{\sigma}}{g(\bar{\epsilon}_p)} \right)^{m_1} \quad \text{if } \dot{\bar{\epsilon}}_p < \dot{\bar{\epsilon}}_t$ $\left( 1 + \frac{\dot{\bar{\epsilon}}}{\dot{\bar{\epsilon}}_0} \right) \left( 1 + \frac{\dot{\bar{\epsilon}}_t}{\dot{\bar{\epsilon}}_0} \right)^{\frac{m_2-1}{m_1}} = \left( \frac{\bar{\sigma}}{g(\bar{\epsilon}_p)} \right)^{m_2} \quad \text{if } \dot{\bar{\epsilon}}_p > \dot{\bar{\epsilon}}_t$ $g(\bar{\epsilon}_p) = [1 - \alpha(T - T_0)] \sigma_0 \left( 1 + \frac{\bar{\epsilon}_p}{\bar{\epsilon}_0} \right)^{\frac{1}{n}}$	$\sigma_0$ , n, $m_1$ , $m_2$ , $\alpha$	(Marusich et Ortiz, 1995)



## **CHAPTER 1**

### **RESEARCH OBJECTIVES AND THESIS OUTLINE**

#### **Research objectives**

The determination of material flow stress constants under high strains, strain rates, and temperatures during machining conditions has long been a major challenge but a necessity for those who apply the finite element simulation modeling techniques for machining process development. As mentioned above, the different material constants provided in the literature for the same material are not reliable since they significantly affect the predicted results. It was shown that most of these parameters are reasonably predicted when using material constants obtained from machining tests. Unfortunately, a reduced number of machining experiments for the identification of the constitutive law have been used which can affect the optimization procedure. In addition, different rake angles were used during these machining tests to obtain the material constants. In fact, the chip formation mechanism could easily change from continuous to segmented chip when the rake angle changes from positive to negative values. These two mechanisms lead to different thermo-mechanical loads in the cutting zone. Thus, the rake angle appears to have a significant effect on the constitutive models when the inverse method is considered.

The overall objective of this research work is to cover this issue to comprehensively understand the effect of the material constitutive law on the predicted machining results.

In particular, the specific objectives of the proposed research are to:

- i) Develop an experimental approach to identify the material constants of the JC constitutive law model for finite element modeling simulation of high speed machining in order to improve the existing inverse method;

- ii) Investigate the effect of the rake angle on the material constants of the JC constitutive law model;
- iii) Conduct a sensitivity analysis on the effect of the different sets of JC constitutive law material constants, identified at different rake angles, on the numerically predicted machining results in order to standardize the existing inverse method.

### **Thesis outline**

This research work is presented as a thesis by publication and is divided into five chapters.

Chapter 2 provides an overview of the relevant works that have been achieved in the literature and it ends by a summary and a literature review in order to highlight the problem defining the scope of the present research.

Experimental and finite element details are presented in chapter 3.

Chapter 4 presents the first published journal article. In this research work, an inverse approach based on response surface methodology was developed to determine the material constants of Johnson-Cook. Three aluminum alloys (Al2024-T3, Al6061-T6, and Al7075-T6) were considered in order to cover a wide range of commercial aluminum alloys commonly used in aircraft applications. In addition, a particular focus was made to study the effect of the rake angle on the identification of the constitutive law. Finally, a FEM investigation was carried out to validate the obtained material constants.

From the above investigation, it was concluded that the rake angle has a significant effect on the constitutive model when the inverse method is considered. In this context, five sets of JC constitutive law determined at five different rake angles and obtained in the first article were employed to simulate the machining behavior of Al2024-T3 alloy using FEM. Therefore, the effects of these sets of JC constants on the numerically predicted cutting forces, chip

morphology, and tool-chip contact length were the subject of a comparative investigation of the second published journal article presented in chapter 5.

Finally, chapter 6 presents the third submitted journal article. In this work, the effect of different sets of JC constants on the numerically predicted residual stresses in the machined components of Al2024-T3 and cutting temperatures for the uncoated carbide tool were investigated. In this context, two different approaches are considered in this study. The former is a thermo-mechanical analysis using Deform-2D finite element software in order to predict the residual stresses induced in the workpiece. The latter is a pure thermal simulation using Deform-3D software to obtain the temperature distribution in the cutting tool.

The thesis conclusion drawn from the current research work and recommendations for future work are provided at the end of this thesis.





## **CHAPTER 2**

### **LITERATURE REVIEW**

#### **2.1 Introduction**

This chapter summarises the most important issues relevant to the current works. The first section is devoted to the machining-induced residual stresses: their importance, their definition, their possible sources, their classifications, the different experimental techniques used to measure them, and the subsurface residual stress measurements as well as the different techniques used to correct these measurements. The second section presents the cutting temperature and the most commonly experimental techniques for temperature measurements. The third section describes the main FEM aspects for metal cutting processes. This includes the presentation of different finite element formulations, the time integration methods for solving non-linear problems, the existing chip separation methods, and the modeling of both the workpiece material and the friction at tool-chip interface. The fifth section is devoted to the FEM of the metal cutting, underlying the main investigations that have been carried out in this regard. Finally, this chapter ends by providing the main finding in the literature including a review in order to highlight the problem defining the scope of the present research.

#### **2.2 Residual stresses induced by the machining process**

The functional behavior of a structural component is heavily influenced by the residual stress distribution caused by the machining process. It is known that fatigue life, deformation, static strength, chemical resistance, and electrical properties are directly influenced by the residual stresses (Brinksmeier *et al.*, 1982; Capello, 2005; El-Axir, 2002; Young, 2005). It is therefore necessary to understand and control the residual stresses for the functionality and longevity of engineering structures.

The residual stresses are defined as those stresses that remain in the machined workpiece after machining is completed and a return to the initial state of temperature and loading is achieved. During machining, the formation of residual stresses is induced under the action of the three following mechanisms (Guo et Liu, 2002b):

- Mechanical deformation: non-uniform plastic deformation due to cutting forces;
- Thermal deformation: non-uniform plastic deformation induced as a result of thermal gradient;
- Metallurgical alterations: specific volume variation resulted from phase transformation.

It is worth mentioning that the first two mechanisms are always present and occur simultaneously in most cutting processes while the third one depends on the amount of heat generated during the cutting, as well as the cooling rate.

Mechanical deformation induced in superficial layer material due to the applied mechanical load may produce both tensile and compressive residual stresses (El-Wardany *et al.*, 2000). In fact, the superficial layer material of the workpiece is subjected to loading cycles of stress versus strain along the cutting direction, as shown in Figure 2-1. The material element first experiences compressive plastic deformation ahead of the advancing cutting tool and then tensile plastic deformation behind it. As a result, this region is the seat of two consecutive modes of deformation and the predominant one, after cutting, determines the final state of residual stresses (Wu et Matsumoto, 1990). Since the superficial layer material is constrained by the bulk material beneath, surface compressive residual stress will be produced, after relaxation, if the loading cycles give rise to a tensile plastic deformation, and vice versa, as shown in Figure 2-1.

In dry cutting, the superficial layer material of the workpiece absorbs more heat and tends to elongate more than the bulk material beneath. After cutting, the hot superficial layer remains hot because the cooling starts mainly from the bulk material by conduction. Since its thermal

expansion is constrained by the bulk material beneath, large compressive stress is generated in the surface material. If this compressive stress exceeds the yield strength of the surface material, the superficial layer will plastically deformed under compression stress, which in turn, surface tensile residual stress is produced after cooling (El-Wardany *et al.*, 2000; Shi et Liu, 2004).

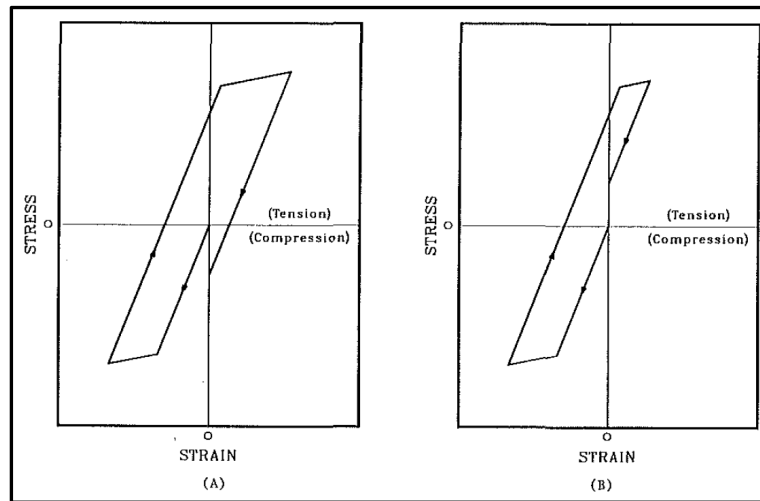


Figure 2-1 (A) predominantly tensile plastic deformation  
(B) predominantly compressive plastic deformation  
(Wu et Matsumoto, 1990)

As mentioned above, if the temperature induced during cutting and the cooling rate are both high enough, phase transformation occurs on the newly generated surface which results in alteration in grain structure that, in turn, can modify the physical and mechanical properties of the workpiece material (Oh et Altan, 1989). These non-uniform alterations between the surface layer and the bulk material can induce residual stresses. The type of residual stresses is related to grain size in the new phase. If the new phase induces larger grain size, the surface layer tends to expand. Since the expansion is constrained by the underlying material, that has no phase transformation, compressive residual stresses are generated in the surface layer and tension stresses are generated in the underlying material (Brinksmeier *et al.*, 1982).

The residual stresses can also be classified into three types according to the length scales over which they act (Lu, 1996):

- The macrostresses (Type I) vary over large distance (several grains or more);
- The microstresses (Type II) vary on the length scale of grains. They are produced by variations between different phases or between inclusions and matrix;
- The stresses at sub-grain scale (Type III) vary over several atomic distances within the grain. They arise from defects, dislocations and precipitates.

### **2.2.1 Residual stress measurement techniques**

It is important to keep in mind that residual stresses are not directly measurable; instead the stress determination require measurement of intrinsic properties, such as elastic strain, displacement or some secondary quantities, such as speed of sound, or magnetic signature that can be related to the stress (Withers *et al.*, 2008).

Residual stress measurement methods could be classified as direct and indirect methods. Indirect methods rely on deformation measurement due to the disequilibrium of the residual stresses, which are relaxed by removing a thin layer of stressed material from the workpiece. By measuring these deformations, the residual stresses that exist in the removed layer could be found. On the contrary, the direct method is based on the measurement of physical quantities that are related to the existing stresses (Brinksmeier *et al.*, 1982). Another classification of residual stress measurement methods namely destructive and non-destructive methods is also proposed in the literature.

### **2.2.2 Indirect methods**

The deflection method is the most useful technique for practical application as reported in (Brinksmeier *et al.*, 1982). In this technique, layers of material are removed from the surface of the workpiece in order to cause disequilibrium of residual stresses existing in the

workpiece. This disequilibrium results in deflections in the remaining workpiece. These deflections are then measured and the residual stresses are finally calculated using elasticity theory (Lu, 1996). Hole drilling is another commonly used method which involves local material removal and measurement of deformations that occur in the neighborhood of the hole due to the residual stresses relaxation. These deformations are usually measured by strain gages. Based on deformation values, Young's modulus, Poisson's ratio, and calibration coefficients or calibration function of the workpiece, residual stresses could be found (Lu, 1996).

### **2.2.3 Direct methods**

X-ray diffraction (XRD) has been widely used for measuring residual stresses of machined crystalline materials (Jawahir *et al.*, 2011). The basic idea of this technique relies on the measurement of the variation in the lattice spacing. This altered spacing caused by stress can be found by measuring the angular position of the diffracted X-ray beam. As a result, the variation of lattice spacing represents strain from which the residual stress can be determined (Lu, 1996). The main advantages of using XRD technique are:

- Non-destructive technique when the surface measurements are concerned, however, removing successive layers of surface material layers is required to determine the in-depth residual stresses;
- Variable measuring area with the possibility to repeat measurements.

On the contrary, the main drawback of XRD technique is that it is a time consuming procedure when the in-depth residual stress is required. In this case, the electro-polishing technique is generally used for removing successive layers of surface material without generating additional residual stresses (Brinksmeier *et al.*, 1982).

It is worth noting that an X-ray beam has a variable measuring area from square centimeters to square millimeters, and a penetration depth of about 10-30  $\mu\text{m}$ , depending on the material

and X-ray source, which means that the measured residual stress represents the arithmetic average stress (Prevey, 1986). However, some error sources are inherent when using XRD technique. The measurement of the thickness of the etched layer, the material homogeneity, and the presence of large grains are to name a few.

The neutron diffraction method relies on the same physical principles as the XRD technique but with the use of neutron sources. Neutron diffraction provides the measurement of the in-depth residual stress without the need for layers removal because of its larger penetration depth (Prime, 1999; Sharpe, 2008). Nevertheless, the volume resolution of the material over which a measurement is made is usually greater than 1 mm<sup>3</sup> (Lu, 1996). As a result, the neutron diffraction technique is not appropriate to provide in-depth residual stress profiles induced by machining process that are limited to few hundreds of micrometers.

Another promising non-destructive technique for residual stress measurement is the ultrasonic technique. This technique is based on the fact that the speed of propagation of ultrasonic waves in a solid is related to its state of strain. Similarly, the electromagnetic technique is based on the interaction between magnetization and the state of strain in ferromagnetic materials (Brinksmeier *et al.*, 1982; Lu, 1996). Ultrasonic and electromagnetic techniques offer fast measuring time (few seconds); however, the structure, composition, hardness, texture, density as well as electric and magnetic properties of the sample can affect the stress measurements (Brinksmeier *et al.*, 1982).

In the current work, the residual stress state of the machined surface and sub-surface was measured by means of X-ray diffraction technique using the  $\sin^2\psi$  method.

#### **2.2.4 In-depth residual stress measurement by X-ray**

It is known that the penetration depth of X-ray in the material is about few microns; therefore, it is necessary to remove thin layers of material after each measurement by X-ray in order to expose the subsurface layers. The removing material is commonly performed by

electro-polishing technique. This technique allows dissolving gradually and locally the sample surface without generating additional residual stresses (Prevey, 1986). However, layer removal results in a new equilibrium state characterized by a change in the residual stress distribution. As a result, the measured in-depth stress profile does not correspond to the real one initially existing in the structure and a correction of the measured stress is needed. However, in some situations a correction is necessary while in other situations it is negligible (Lu, 1996; Prevey, 1986). Two corrections methods are available; analytical model developed by (Moore et Evans, 1958) and a numerical method based on finite element analysis (Prevey, 1996; Savaria *et al.*, 2012) are presented in the next two sections.

#### 2.2.4.1 Analytical correction method of residual stresses

The analytical model proposed by Moore et Evans (1958) is one of the most commonly used correction method in industry. It allows correction for stress redistribution in simple geometries such as solid and hollow cylinders and the flat plates. In the case of a hollow cylinder geometry, the correction equations are given as follows:

$$\sigma_{rc}(R_m) = - \left( 1 - \frac{R_{in}^2}{R_m^2} \right) \int_{R_m}^{R_{out}} \left( \frac{r \sigma_{\theta m}(r)}{r^2 - R_{in}^2} \right) dr \quad (2-1)$$

$$\sigma_{\theta c}(R_m) = \sigma_{\theta m}(R_m) + \left( \frac{R_m^2 + R_{in}^2}{R_m^2 - R_{in}^2} \right) \sigma_{rc}(R_m) \quad (2-2)$$

$$\sigma_{zc}(R_m) = \sigma_{zm}(R_m) - 2 \int_{R_m}^{R_{out}} \left( \frac{r \sigma_{zm}(r)}{r^2 - R_{in}^2} \right) dr \quad (2-3)$$

where  $\sigma_{rc}$ ,  $\sigma_{\theta c}$ , and  $\sigma_{zc}$  are the corrected stresses in radial, tangential, and axial directions, respectively.  $\sigma_{\theta m}$  and  $\sigma_{zm}$  are the measured stresses,  $R_{in}$ ,  $R_{out}$ ,  $R_m$  are the inner, outer, and actual measurement radius, respectively. The main advantage of the Moore and Evans correction solution is the determination of the radial stress in order to better estimate the

hydrostatic stress used in multiaxial fatigue criteria (Coupard *et al.*, 2008). The Moore and Evans correction equations, mentioned above, are based on the following assumptions:

- The specimen must be long enough and the measurements are taken in a point sufficiently distant from the edges;
- Material removal layers are conducted under circumferential polishing;
- The material behaviour remains elastic during the material removal;
- The stress field is assumed to be either rotationally symmetric or symmetric about a plane.

An analysis of these assumptions shows that it is difficult to fulfill all of these assumptions especially when using a thin disk-shaped workpiece where edges effects become important. Additionally, the circumferential polishing appears to be a laborious task and results in an error in the depth measurement due to the loss of a reference point (Coupard *et al.*, 2008; England, 1997). As a result, the Moore and Evans method appear to produce significant errors in the residual stress corrections (Hornbach *et al.*, 1995; Savaria *et al.*, 2012).

#### **2.2.4.2 Finite element correction method**

An alternative method known as FEA matrix relaxation correction method has been recently developed based on FE simulation (Prevey, 1996; Savaria *et al.*, 2012). The main advantage of this method is its ability to take into account the complexity of the geometry of both the part and the local material removal zone. The main assumptions of this method as reported by (Savaria *et al.*, 2012) are:

- The redistribution of residual stresses is assumed to remain elastic after the material removal;
- Removing successive layers of surface material is done without generating additional residual stresses;



- The residual stresses are assumed to be uniform over the surface of the removed layer;
- The residual stresses measured by XRD technique at a surface layer are assumed to be constant over the thickness of a removed layer;
- The geometry of the part and the material removal zone formed by polishing are identical in both the experimental layer removal and the numerical simulation;
- The stresses in each direction are independent, i.e. the stress redistribution in one direction is not influenced by the stress redistribution in other directions.

The basic idea behind this method is that the total correction of residual stress at any depth and in the direction of interest depends only on the geometry (part and material removal zone) and the values of the previously released stresses in that direction. As a result, the local stress variation  $(\Delta\sigma_d)_s$  observed at a depth of “d” after removing a layer “s” can be given as follows (see Figure 2-2):

$$(\Delta\sigma_d)_s = (\sigma_d)_s - (\sigma_d)_{s-1} = -K_{ds} \sigma m_s \quad (2-4)$$

where  $K_{ds}$  is the correction coefficient applied for the stress direction of interest and at depth “d” and after removing the layer “s”.  $(\sigma_d)_s$  and  $(\sigma_d)_{s-1}$  are the stresses at depth “d” after removing layers “s” and “s-1”, respectively.  $\sigma m_s$  is the stress measured in the same direction of interest and at the top surface of the layer “s” before its removal.

The dependence of the correction coefficients  $K_{ds}$  on the geometry only, allows these coefficients to be determined numerically from any known residual stress profile induced in the part and can then be employed to correct any residual stress profile measured experimentally in the same direction, location, and geometry using the same layer removal zone dimensions (Prevey, 1996). Since the first point is measured by XRD without any layer removal, no correction will be applied for this measurement but for all others points in depth, for example at depth “d”, the corrected stress  $\sigma c_d$  is calculated from the measured one  $\sigma m_d$  at this depth and stresses measured in all previous steps  $\sigma m_s$  as follows:

$$\sigma c_d = \sigma m_d + \sum_{s=1}^{d-1} K_{ds} \sigma m_s \quad (2-5)$$

Equation (2-5) can be written in a matrix form as follows:

$$\sigma c = [I + K_{cor}] \sigma m \quad (2-6)$$

where  $\sigma c$  and  $\sigma m$  are column vectors of the corrected and measured stresses at  $n$  points in depth.  $I$  and  $K_{cor}$  are the identity and the correction matrixes of dimension  $n \times n$ . It is worth noting that the  $K$  matrix is different from one stress direction to another.

The next steps can be followed to determine the  $K_{cor}$  matrix:

- Create a 2D or 3D finite element model of the part of interest;
- Use artificial thermal gradients to generate the residual stresses in the model;
- Collect the generated residual stresses in subsurface layers of interest before removing any layer;
- Start simulating the polishing (layer by layer) and collect the residual stresses in the remaining subsurface layers after each removal of overlying layers;
- Use Equation (2-4) to calculate the correction coefficients in order to generate the  $K$  matrix in which each column is generated after each removal of overlying layers;
- Use Equation (2-6) to correct any residual stress profile induced in the same part geometry and using the same layer removal zone dimensions.

Savaria *et al.* (2012) introduced a modification to this method by considering the average of the two stress values,  $\sigma m_s$  and  $\sigma m_{s+1}$ , measured on both sides of each removed layer  $s$ , which can be useful in the case of high gradient residual stress distributions. This average stress is defined as follows:

$$\sigma_{avg_s} = \frac{\sigma_{m_s} + \sigma_{m_{s+1}}}{2} \quad (2-7)$$

As a result, the new equations for residual stress corrections become:

$$(\Delta\sigma_d)_s = -\dot{K}_{ds} \sigma_{avg_s} \quad (2-8)$$

$$\sigma_{c_d} = \sigma_{m_d} + \sum_{s=1}^{d-1} \dot{K}_{ds} \sigma_{avg_s} \quad (2-9)$$

$$\sigma_c = \sigma_m + \dot{K}_{ds} \sigma_{AVG} \quad (2-10)$$

where  $\sigma_{AVG}$  represents a column vector containing all the average stresses calculated individually for each layer using Equation (2-7) and placed in order of depth. It is worth noting that Equations (2-7) to (2-10) should be performed for each stress direction.

In the present work, corrections to the residual stress measurements due to the removed volume of material were done by FEA matrix relaxation correction method and presented in APPENDIX I.

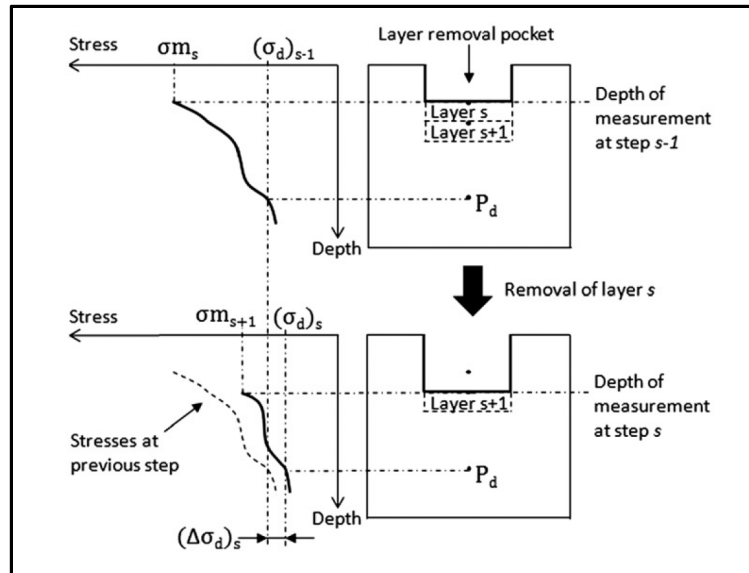


Figure 2-2 Schematic drawing of a layer removal process and a visualisation of the stress redistribution (Savaria *et al.*, 2012)

### 2.3 Cutting temperatures

As mentioned earlier, the high plastic deformation in shearing zones and the friction along the tool-chip interface are the two main heat sources in metal cutting. The knowledge of the temperature distribution in metal cutting is highly important for the following reasons:

- Temperature generation substantially influences the mechanical properties of the workpiece material and hence the cutting forces (Outeiro *et al.*, 2002);
- Temperature generation at the tool-chip interface affects the contact mechanism by altering the friction conditions, which in turn alters the maximum temperature location, heat partition, and the diffusion of the tool material into the chip (Abukhshim *et al.*, 2006).

Based on these considerations, the temperature distributions within the machined aircraft components are highly important for aerospace industries, especially for those who use mechanical machining instead of chemical machining. This is because the machining residual stresses induced from temperature rise result in geometric distortions. On the contrary, the

cutting tool temperature distributions, especially at tool tip, are highly important for tool manufacturers who are interested in tool wear. As a result, a number of techniques have been developed to measure cutting temperatures which are pointed out in the next section.

The literature review illustrates that the most common experimental techniques used for cutting temperature measurements can be classified as: direct conduction, indirect radiation, and metallographic (Ay et Yang, 1998; Komanduri et Hou, 2001; O'sullivan et Cotterell, 2001; Sutter *et al.*, 2003). These techniques involve: Thermocouples (tool-work thermocouples and embedded thermocouple), radiation techniques (infrared thermography and infrared pyrometer), powders of known constant melting point, and metallographic techniques.

The temperature measurement by a thermocouple is based on the principle that when two dissimilar metals are connected together to form two junctions, namely hot and cold junctions, and if these two junctions are kept at two different temperatures, an electromotive force (EMF) will be generated through these two junctions. The EMF produced is a function of the materials used for the thermocouple as well as the temperatures of the junctions (Astakhov, 1998).

The tool-work thermocouple technique is based on the fact that the tool and workpiece materials are different; therefore, they are used to form a natural thermocouple. The hot junction of this thermocouple is the tool-chip interface while the tool or the workpiece materials form the cold junction (Lezanski et Shaw, 1990; Stephenson et Ali, 1992). Though this technique is relatively easy to apply, it has the disadvantage that it measures the mean temperature over the entire tool-chip interface. Additionally, both tool and workpiece should be electrical conductors.

Embedded thermocouple inserted inside the cutting tool (or workpiece) was widely used to measure the temperature either at a single point or at different points in order to determine the temperature distribution (Chen *et al.*, 1997; Filice *et al.*, 2007a; Filice *et al.*, 2006;

Kitagawa *et al.*, 1997; O'sullivan et Cotterell, 2001; Umbrello *et al.*, 2007a). By using this technique, several fine deep holes are drilled in the cutting tool (or workpiece) and the thermocouples are then inserted inside them. In order to provide an accurate measurement, the depth of holes needs to be as close as possible to the surface of interest. The main limitation of this technique is when more than one hole are drilled in the cutting tool which lead to a limitation in the tool strength and changing in the heat conduction of the tool. Difficulties to drilling holes in hard tool materials, increasing the response time with an increasing diameter of the thermocouple are other limitations.

The radiation techniques are also employed to determine the cutting temperature based on the heat radiation emitted from the workpiece, the chip, and the cutting tool. If temperature field is needed, infrared thermography technique could be adopted using infrared cameras or photo cameras with films reactive to infrared radiation (Boothroyd, 1961; Dewes *et al.*, 1999; Filice *et al.*, 2006; O'sullivan et Cotterell, 2001; Young, 1996). Otherwise, infrared pyrometer could be used in the case of point measurement (Kottenstette, 1986; Lin *et al.*, 1992; Ming *et al.*, 2003; Müller et Renz, 2003). Even though these techniques are nonintrusive and hence do not disturb the measured phenomenon, fast with short response time, and giving access to difficult-to-measure machining zones, they require a long and complex calibration. Moreover, the measurements are highly influenced by many factors such as the surface conditions of examined object and environmental absorption and reflection of infrared radiations (Filice *et al.*, 2006).

The fine powder technique is also used to measure the temperature in the cutting tool (Casto *et al.*, 1989; Kato *et al.*, 1976). In this technique the temperature distribution is obtained by observing the boundary line formed by melted and non-melted powder dispersed on the surface of interest. The main limitation of this technique is the long time needed to entirely melt the powder (Abukhshim *et al.*, 2006).

In temperature measurement using metallographic techniques, the temperature gradient within the cutting tool is determined by performing metallographic examinations and/or by

measuring micro-hardness tests of the affected zone after cutting experiments (Dearnley, 1983; Wright, 1978). These techniques suffer from some of drawbacks such as measurement uncertainty (about  $\pm 25^{\circ}\text{C}$ ), limited only to tool materials that experience a change in microstructure with temperature, calibration difficulties, and no temperature gradient could be obtained as a function of time.

Since the aim of the present work is to analyse the effect of different sets of the material constants on the predicted cutting temperature in the cutting tool, it is more convenient to have experimental measurements at a single point; therefore, the cutting temperature in the cutting tool was measured by means of an embedded thermocouple (type K).

## **2.4 Finite element modeling considerations in metal cutting simulations**

Numerical modeling techniques such as FEM are widely employed in metal cutting to better understand the chip formation mechanism. In addition, it gives access to many difficult-to-measure machining process variables and it has proven to be highly efficient (Vaz Jr *et al.*, 2007). Moreover, FEM appears to be the most suitable method to design and develop the machining processes and cutting tool as compared to the experimental procedures which are costly and time consuming. Consequently, the experimental trial and error approach could be avoided.

In this section, some essential aspects related to FEM of metal cutting are presented, such as finite element formulations, time integration methods, chip separation methods, constitutive law models representing the flow stress for machining, and friction models.

### **2.4.1 Finite element formulations**

There are primarily two different types of finite element formulations in FEM of metal cutting process. These are Eulerian and Lagrangian formulations.

In the Eulerian formulation the finite element mesh is totally fixed in space and the material flows through it. While the Lagrangian formulation assumes that the finite element mesh is totally embedded in material and follows its deformation. Therefore, the material points do not stay coincident with element integration points but they change their positions with time in the Eulerian meshes, whereas in Lagrangian meshes the material points stay coincident with element integration points (Belytschko *et al.*, 2013). Since the material points do not coincide with the same integration points during the entire motion in Eulerian formulation, the material time derivatives contain convective terms. In fact, the presence of convective terms makes the material time derivative much more difficult to handle. History-dependent material properties, such as stresses and strains, have to be interpolated in an approximate way and material points have to be traced back to the fixed mesh points (Movahhedy *et al.*, 2000a); consequently, Lagrangian formulation is more prevalent in the history-dependent analysis (Belytschko *et al.*, 2013).

The literature review illustrates that the Eulerian formulation is used in FEM of metal cutting process (Carroll et Strenkowski, 1988; Dirikolu *et al.*, 2001; Raczy *et al.*, 2004; Tay *et al.*, 1974). As the mesh is spatially fixed, this approach offers the advantage to eliminate the problems associated with high element distortion due to the large deformation in the shear zones and consequently, no remeshing is needed. Therefore, the Eulerian formulation is more suitable for modelling the zone around the tool tip, at least for ductile material. (Movahhedy *et al.*, 2000a). The major drawback of this formulation is that the chip geometry has to be assumed a priori and it cannot be predicted as an outcome of the cutting process. It was argued that the chip thickness is the major output parameter that affects all other cutting parameters so it cannot be assumed physically (Astakhov et Outeiro, 2008). Consequently, the predictive nature of such approach appears to be limited and cannot represent the real deformation process encountered in real machining process.

When Lagrangian formulation is used in FEM of metal cutting process, the material will be deformed under the action of the cutting tool. Therefore, the chip geometry is the outcome of simulation and is based on the cutting conditions. In addition, this provides simpler scheme



to simulate the chip formation from transient to steady state conditions and modeling the continuous chip and segmented one as well as its ability to predict residual stresses (Ceretti *et al.*, 1999; Komvopoulos et Erpenbeck, 1991; Outeiro *et al.*, 2008; Özel, 2006; Salvatore *et al.*, 2012; Strenkowski et Carroll, 1985). There are, however, two main problems associated to the use of Lagrangian formulation: The first one is related to high plastic deformation encountered in shear zones and, the second one is related to the formation of the chip. High plastic deformation localized in front of tool tip results in severe distortions of the mesh and involves the use of pre-distorted mesh (Shih, 1995) or remeshing technique (Marusich et Ortiz, 1995). In addition, for the chip formation, many chip separation criterions were adopted in the literature and this approach is generally thorny as it is discussed in next paragraph. Although the aforementioned disadvantages, the Lagrangian formulation is still more attractive than the Eulerian formulation in the metal cutting simulation (Movahhedy *et al.*, 2000a), and is used in the current study.

## 2.4.2 Time integration methods

### 2.4.2.1 Mechanical analysis

The dynamic equilibrium condition with no damping consideration is given as follows:

$$M \ddot{u} + K u = F_e \quad (2-11)$$

where  $M$  is the mass matrix,  $K$  is the stiffness matrix,  $u$  is displacement vector,  $\ddot{u}$  is acceleration vector, and  $F_e$  is the externally applied force vector. To solve Equation (2-11) using FEM, implicit and explicit time integration methods can be used to determine the final dynamic equilibrium condition due to  $F_e$ .

In implicit integration, the determination of the displacement vector at the end of the first time step ( $u^{(1)}$ ) is essential to start the solution and Newmark  $\beta$ -method is used as follows:

$$u^{(1)} = u^{(0)} + \Delta t \dot{u}^{(0)} + 0.5 \Delta t^2 (1 - 2\beta) \ddot{u}^{(0)} \quad (2-12)$$

The initial acceleration ( $\ddot{u}^{(0)}$ ) is given as follows:

$$\ddot{u}^{(0)} = [M_{sys}^{(0)}]^{-1} \times \{F_e^{(0)} - K_{sys}^{(0)} \times u^{(0)}\} \quad (2-13)$$

Then, the Newton-Raphson method is usually utilized to enforce equilibrium of the internal structure forces with the externally applied ones by performing iterations until a convergence criterion (residual function  $G$ ) is satisfied for each time step  $\Delta t$ . The iterative solution is given by the following equations:

$$\ddot{u}^{(t)} = \frac{u^{(t)} - 2u^{(t-\Delta t)} + u^{(t-2\Delta t)}}{\Delta t^2} \quad (2-14)$$

$$G(u^{(t)}) = M_{sys}^{(t)} \ddot{u}^{(t)} + K_{sys}^{(t)} u^{(t)} - F_e^{(t)} \quad (2-15)$$

$$\Delta u^{(t)} = -[K_{sys}^{(t)}]^{-1} \times G(u^{(t)}) \quad (2-16)$$

$$u^{(t+\Delta t)} = u^{(t)} + \Delta u^{(t)} \quad (2-17)$$

Implicit time integration has the advantage of being unconditionally stable for linear problems and it allows the use of large time step in nonlinear problems. It is worth pointing out that in the case of nonlinear problems with large deformation, as in metal cutting, the mass matrix ( $M_{sys}^{(t)}$ ) and the stiffness matrix ( $K_{sys}^{(t)}$ ) of the whole structure will be calculated and inverted for each time step  $\Delta t$ , therefore, implicit time integration is regarded as CPU time consuming (Sun *et al.*, 2000).

In explicit time integration, the central difference time integration rule is utilized to link the displacement, the velocity, and the acceleration in the following way:

$$\ddot{u}^{(t)} = [M^{(t)}]^{-1} \times \{F_e^{(t)} - K^{(t)} \times u^{(t)}\} \quad (2-18)$$

$$\dot{u}^{(t+\Delta t/2)} = \dot{u}^{(t-\Delta t/2)} + \Delta t \times \ddot{u}^{(t)} \quad (2-19)$$

$$u^{(t+\Delta t)} = u^{(t)} + \Delta t \times \dot{u}^{(t+\Delta t/2)} \quad (2-20)$$

Unlike implicit integration, the solution is advanced using known values of velocities ( $\dot{u}^{(t-\Delta t/2)}$ ) and accelerations ( $\ddot{u}^{(t)}$ ) from the previous time step, consequently, there is no need for iteration procedure. The advantage of using explicit integration is there is no need to compute stiffness and mass matrices for the whole system; instead the solution is solved based on the element level which is not computationally expensive. The drawback of this method is that it is conditionally stable and the time step  $\Delta t$  must be carefully chosen and has to be smaller than the critical time step  $\Delta t_{cr}$ , (Guo et Liu, 2002b).

It is worth mentioning that implicit integration method has been used in the current study which is widely used in the literature (Abboud *et al.*, 2013; Bäker, 2005; Hua *et al.*, 2006; Outeiro *et al.*, 2006; Umbrello, 2008; Yen *et al.*, 2004b); however, other studies could be found that used explicit integration (Guo et Liu, 2002b; Mabrouki *et al.*, 2008; Nasr *et al.*, 2007b; Shi et Liu, 2004; Shrot et Bäker, 2012)

Quasi-static implicit time integration is used in DEFORM software. The quasi-static assumption is based on the fact that the inertia forces can be ignored as compared to those generated by deformation. The nonlinear set of stiffness equations can be solved iteratively by using the Newton-Raphson method. This method is based on linearization and application of convergence criteria to obtain the final solution (velocity). Linearization is obtained by

Taylor expansion near an assumed solution point (initial guess) which can be written in the form

$$K \Delta \mathbf{v} = \mathbf{f} \quad (2-21)$$

where  $K$  is the stiffness matrix and  $\mathbf{f}$  is the residual of the nodal point force vector.

Once Equation (2-21) is solved for velocity correction term  $\Delta \mathbf{v}$ , the initial guess velocity  $\mathbf{v}_0$  is updated as follows:

$$\mathbf{v} = \mathbf{v}_0 + \gamma \Delta \mathbf{v} \quad (2-22)$$

where  $\gamma$  is deceleration coefficient which is a constant between 0 and 1.

Iteration solution is continued until the velocity correction term  $\Delta \mathbf{v}$  becomes small enough and negligible. The Newton-Raphson iteration procedure is given schematically in Figure 2-3. More detailed information can be found in (Kobayashi *et al.*, 1989).

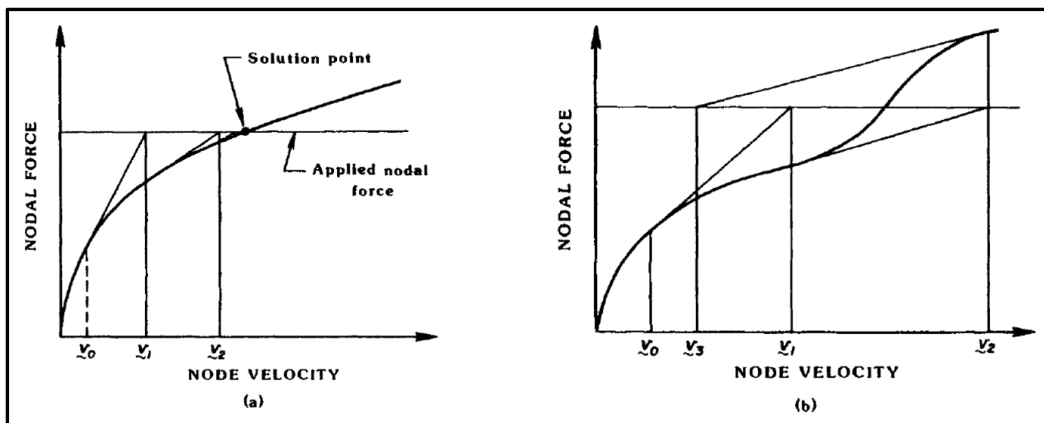


Figure 2-3 Representation of the Newton-Raphson method:  
(a) convergence (b) divergence, (Kobayashi *et al.*, 1989)

### 2.4.2.2 Thermal analysis

Finite element discretization for heat balance equation can be expressed in the following form

$$\mathbf{C} \dot{\mathbf{T}} + \mathbf{K}_c \mathbf{T} = \mathbf{Q} \quad (2-23)$$

where  $\mathbf{C}$  is the heat capacity matrix,  $\mathbf{K}_c$  is the heat conduction matrix,  $\mathbf{Q}$  is the heat flux vector,  $\mathbf{T}$  is the vector of nodal point temperatures, and  $\dot{\mathbf{T}}$  is the vector of nodal temperature rates.

The needed theory to integrate Equation (2-23) can be found in numerical method books (Dahlquist et Bjorck, 1974). The convergence of a discretization problem can be bounded in terms of its consistency and stability. Consistency is satisfied by an approximation of the type

$$\mathbf{T}^{(t+\Delta t)} = \mathbf{T}^{(t)} + \Delta t[(1 - \beta)\dot{\mathbf{T}}^{(t)} + \beta\dot{\mathbf{T}}^{(t+\Delta t)}] \quad (2-24)$$

where  $\beta$  is a parameter varying between 0 and 1, and  $t$  represents time.  $\beta$  should be greater than 0.5 for unconditional stability. A value of 0.75 is usually selected for stability and reasonable convergence time (Hua *et al.*, 2006).

It is worth noting that metal cutting is known as a coupled thermo-mechanical process. Therefore, both the thermal and mechanical analysis should be solved at the same time. However, in an incremental analysis, they are usually solved in a staggered procedure. The basic idea of this procedure is that during a time increment ( $\Delta t$ ), the stress analysis is first solved as an isothermal process with temperature distribution is that of the beginning of this increment. The thermal analysis is then solved using the heat generated in the current

increment and finally the temperatures are updated at the end of ( $\Delta t$ ) (Movahhedy *et al.*, 2000b).

### 2.4.3 Chip separation methods

The analysis of metal cutting using FEM was first carried out in 1970s to better study the mechanics of the cutting process in greater detail than possible in experimental and analytical methods. Different FEM codes have been developed in order to analyze metal cutting processes during the early 1980s-1990s. In these works, two main chip separation criteria based on either geometrical or physical considerations were introduced which allow the chip to separate from the workpiece along a predefined parting line (or plane). In the work of Usui *et al.* (1982), a chip separation criterion based on geometrical consideration have been proposed. The basic idea about this type of criterion is that chip separation occurs when the distance between the nearest node along the predefined parting line and the tool tip becomes less or equal to a critical distance  $d_{cr}$ , as shown in Figure 2-4 (a). The physical consideration is based on physical meaning such as ductile fracture concepts (Iwata *et al.*, 1984), equivalent plastic strain (Strenkowski *et al.*, 1985), and total strain energy density (Lin *et al.*, 1992; Lin *et al.*, 1993). In this case, the chip is said to separate when the considered physical parameter calculated at the nearest node from the cutting edge reaches a critical value, as illustrated in Figure 2-4 (b). However, a sensitivity analysis using different types and magnitudes of chip separation criteria were conducted by (Huang *et al.*, 1996). The authors attested that the magnitude has a great effect on the predicted cutting parameters in the machined surface.

An alternative approach known as automatic remeshing method was developed to overcome the drawbacks of the chip separation criteria and to model the material flow around the cutting tool tip. This remeshing procedure starts by detecting mesh distortion, partitioning the contact boundary, generating new internal nodes, generating new mesh, interpolating the solution information from the old mesh onto the new one, and then the simulation continues. Marusich *et al.* (1995) developed a FEM in which the elements with plastic power

contents exceeding a critical value are detected for refinement. The remeshing technique was also adopted in (Ceretti *et al.*, 1996; Kumar *et al.*, 1997; Özel et Altan, 2000) in order to avoid the need for a chip separation criterion. Commercial FEM code Deform-2D was used in their studies. The disadvantage of this technique is its time consuming and too many repetitive remeshing leading to excessive interpolation error.

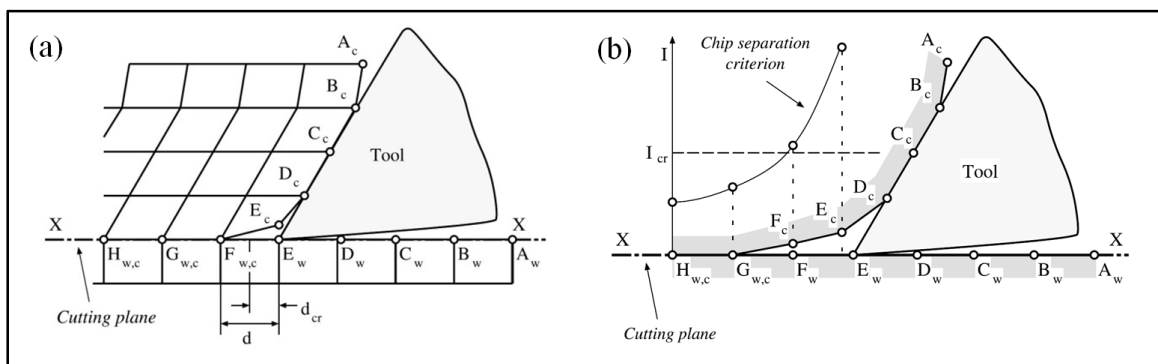


Figure 2-4 Chip separation based on: (a) geometrical criterion (b) physical criterion (Vaz Jr *et al.*, 2007)

#### 2.4.4 Constitutive law models representing the flow stress for machining

FEM of the cutting process has been always a real challenge due to the extreme conditions encountered during the metal cutting as compared to other production processes, (Kalpakjian *et al.*, 2008). The use of an accurate constitutive law model to represent completely the thermo-visco-plastic behavior of the machined material at high strains, high strain rates, and high temperature is difficult to develop. However, one can find in the literature several constitutive law models that are adopted for machining simulation, some of them could be found in (Fang, 2005; Sartkulvanich *et al.*, 2005b; Shi et Liu, 2004). An empirical model proposed by Johnson and Cook (Johnson et Cook, 1983) and a dislocation-mechanics-based constitutive model proposed by Zerilli and Armstrong (Zerilli et Armstrong, 1987) are two of the earliest reliable material models reported in the literature. It is worth pointing out that a successful machining modeling takes place in two steps: first, the selection of an accurate constitutive model and second, the determination of its constants.

The Johnson-Cook constitutive model has been widely used in metal cutting simulation (Arrazola *et al.*, 2008; Mabrouki *et al.*, 2008; Miguélez *et al.*, 2013; Miguélez *et al.*, 2009; Nasr *et al.*, 2007a; Nasr *et al.*, 2007b; 2007c; Outeiro *et al.*, 2008; Umbrello, 2008). Furthermore, it was proved to be appropriate in modelling the cutting process (Adibi-Sedeh *et al.*, 2003; Chen *et al.*, 2014; Huang et Liang, 2003; Karpas et Özel, 2006; Lalwani *et al.*, 2009; Lee, 2011; Li *et al.*, 2011; Long et Huang, 2005; Özel et Zeren, 2004).

The JC constitutive model, like other models, depends heavily on its material model parameters and the method used to determine them. The literature review illustrates that the most common methods used to identify the JC constitutive law constants are (1) dynamic tests (torsion tests, Split-Hopkinson bar technique (SHBT), and Taylor test), (2) analytical method in conjunction with cutting tests, (3) FEM in conjunction with cutting tests, (4) inverse method (machining test).

Johnson et Cook (1983) have used torsion tests over a wide range of strain rates, Split-Hopkinson bar tensile tests over a range of temperature, and static tensile tests in order to determine the material constants of JC constitutive model for various metals. Lesuer (2001b) has determined the first three parameters of the JC model using Split-Hopkinson bar technique (SHBT) at high strain rates ( $10^3$  to  $10^4$  s<sup>-1</sup>). Adibi-Sedeh *et al.* (2003) used data obtained from SHBT at high strains to determine the first three parameters of JC constitutive model. Dannemann (2001) used SHBT at high strain rates in combination with other quasi-static tests, to obtain the JC constitutive constants for two aluminum alloys. Taylor impact test was also used by Rule (1997) to extract the material constants of JC constitutive law at high strain rates up to  $10^5$  s<sup>-1</sup>. Even though these strains and strain rates achieved by these dynamic tests are high, they are still far from those encountered in machining (Li *et al.*, 2011). Moreover, these techniques are costly, complex, and difficult to run (Panov, 2006).

Shatla *et al.* (2001a) have proposed an analytical method in conjunction with orthogonal slot milling tests to determine the material constants of the JC constitutive equation. This method is focused on the minimization of the error between the measured cutting forces and those



predicted analytically using a computer code (OXCUT) developed by the authors. Although the shape of the chip was found serrated for the cutting conditions used to verify the determined flow stress data, the agreement between the predicted and measured cutting forces was still good for the three studied alloys.

The FEM in conjunction with cutting tests has been also proposed in the literature. In this approach, the material constants are determined by matching the cutting force obtained by FEM with the measured one (Yang *et al.*, 2011).

However, the above mentioned methods cannot give a unique solution since they are affected by the secondary shear zone and by the friction model at tool-chip interface (Tounsi *et al.*, 2002). Recently, Shrot et Bäker (2012) has also shown that based on the chip shape and the cutting force, it was possible to find different sets of material constants that lead to indistinguishable results such as chip morphology and cutting force for the same cutting condition.

An alternative method known as the inverse method has also been proposed for characterization using machining tests (Guo, 2003; Laakso et Niemi, 2015; Limido, 2008; Ozel *et al.*, 2006; Tounsi *et al.*, 2002). In this method, the experimental data such as cutting forces, thrust forces and chip geometry were converted to physical quantities namely average stresses, strains, strain rates and temperatures in the primary shear zone using analytical and empirical models. Then, the material constants were obtained using a nonlinear regression solution. The main advantage of such an approach is that extreme conditions such as strain rates up to  $10^6 \text{ s}^{-1}$ , temperature up to 1000 °C and strains up to 4 are achieved directly with machining tests. In addition, it was shown that the predicted results such as cutting forces, chip morphology, temperatures, and residual stresses are reasonably predicted when using material constants obtained from machining tests (Umbrello *et al.*, 2007b). The disadvantage of this method is its circular nature where machining experiments are used to provide the flow stress data that are then employed for making machining predictions (Kristyanto *et al.*, 2002).

#### 2.4.5 Friction models

Friction between chip and cutting tool is considered as important and complicated issues in metal cutting processes. It can determine surface integrity of a machined layer, tool wear, cutting forces, and specific power (Vaz Jr *et al.*, 2007). Therefore, many research works focused on friction modeling and hence several models were proposed in the literature.

The basic Coulomb friction model has been used for modeling the friction at tool-chip interface in the several FE analyses (Carroll et Strenkowski, 1988; Ee *et al.*, 2005; Klocke *et al.*, 2001; Lin et Lin, 1999; Mabrouki *et al.*, 2008; Miguélez *et al.*, 2009; Nasr *et al.*, 2007b). In this model, the coefficient of friction  $\mu$  represents the relation between frictional stress  $\tau_s$  and normal one  $\sigma_n$  as follows:

$$\tau_s = \mu \sigma_n \quad (2-25)$$

However, when the product of interface pressure and coefficient of friction surpasses the shear yield strength of the material, Equation (2-25) fails to give accurate prediction since there is no relative sliding at the interface tool-chip (Schey, 2000).

Alternatively, another friction model known as constant shear model assumes that the frictional stress on the rake face of the tool can be denoted as a proportional to the shear yield stress of the chip material.

$$\tau_s = m_f \times K_{chip} \quad (2-26)$$

where  $m_f$  is the shear friction coefficient ( $0 < m_f < 1$ ) and  $K_{chip}$  is the shear flow stress in the chip at the tool-chip interface. This approach has been widely used in FE simulations in the literature due to its simplicity (Ceretti *et al.*, 1999; Filice *et al.*, 2006; Outeiro *et al.*, 2008; Outeiro *et al.*, 2006; Sartkulvanich *et al.*, 2005a; Shi *et al.*, 2010b; Umbrello *et al.*, 2007a; Umbrello *et al.*, 2007b; Umbrello, 2008; Yen *et al.*, 2004a; Yen *et al.*, 2004b; Yen *et al.*,

2004c). The main shortcoming associated with this model is the omission of the sliding region which may affect the model predictions.

A more realistic model based on experimental observation has been presented by (Zorev, 1963). According to this model, the distribution of the normal and the frictional stresses is depicted in Figure 2-5. Zorev's model assumes the existence of two distinct regions, namely the sticking and sliding regions. The sticking region is located close to the tool tip where the normal stress is very high which results in a high plastic deformation. Consequently the frictional stress is assumed to be equal to the shear yield stress of the chip material  $K_{chip}$ . The sliding region, located over the remaining tool-chip length, is defined by relatively low normal stresses and small plastic deformation. Therefore, the Coulomb model could be properly applied to predict the frictional stress. Zorev's model could be represented by means of the following equation:

$$\begin{aligned}\tau_s &= K_{chip} \text{ when } 0 < x \leq l_p \text{ (} \mu \sigma_n \geq K_{chip} \text{)} \rightarrow \text{sticking} \\ \tau_s &= \mu \sigma_n \text{ when } l_p < x < l_c \text{ (} \mu \sigma_n < K_{chip} \text{)} \rightarrow \text{sliding}\end{aligned}\tag{2-27}$$

This model was used in two ways. In some research works such as (Fang et Zeng, 2005; Li *et al.*, 2002; Ng *et al.*, 1999; Özel et Zeren, 2007; Wen *et al.*, 2006; Zhang et Bagchi, 1994), the shear flow stress ( $K_{chip}$ ) and the coefficient of friction ( $\mu$ ) are defined and then implemented in the FEA software. In other research works, (Filice *et al.*, 2007a; Özel et Altan, 2000; Shatla *et al.*, 2001b), this model is implemented by determining an arbitrary length of the sticking region. As a result, the modeling errors encountered in FE simulation would be higher as compared to the two models mentioned above.

In the current work, the constant shear model is considered due to the above mentioned reasons.

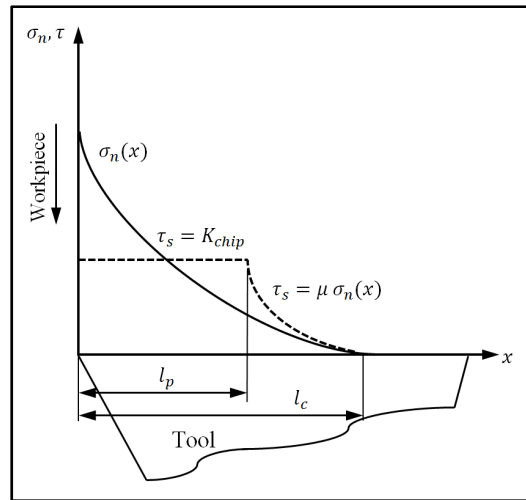


Figure 2-5 Normal and frictional stress distribution according to (Zorev, 1963)

## 2.5 Applications of FEM in simulation of metal cutting

With the high performance of advanced computers and robust finite element codes, FEM of chip formation has recently become one of the most reliable tools for prediction and optimization of machining processes. Consequently, some researchers focused on the use of FEM to predict cutting forces, chip morphology, temperature distributions, residual stresses, etc...for machining of steels, aluminum, and titanium alloys.

Shih (1995) developed a 2D plane strain FE model to study the rake angles effects in the orthogonal cutting of annealed AISI 1020 carbon steel with continuous chip formation. In this study, the effects of elasticity, viscoplasticity, temperature, large strain and high strain rate were considered to model the behavior of the machined material. The chip separation is based on distance. It was shown that higher cutting forces, thicker chip, smaller shear angle, and longer tool-chip contact length are obtained when the rake angle changes from positive to negative values. Cutting with negative rake angle experiences higher temperature distribution, large plastic strain, and higher effective stresses near the tool tip.

Liu et Guo (2000) used the commercial explicit finite element software ABAQUS to develop a thermal-elastic-viscoplastic FE model of annealed 304 stainless steel in order to investigate the effects of sequential orthogonal cuts and tool-chip friction on residual stresses in a machined layer, as well as, the effect of sequential cuts on chip formation, cutting forces and temperature.

Later, using the same workpiece material, Guo et Liu (2002b) carried out a similar investigation on cutting parameters (cutting forces and plastic deformation and temperature) and residual stresses in sequential cuts. They found that a compressive residual stress distribution can be obtained by decreasing the uncut chip thickness below a critical value in the second sequential cut.

Shi et Liu (2004) investigated the effect of several constitutive law models namely, Litonski-Batra, power law, Johnson-Cook, and Bodner-Partom on the predicted results during orthogonal cutting of HY-100 steel with continuous chip formation. ABAQUS software was used to conduct a fully coupled thermal-stress analysis. The equivalent plastic strain criterion was adopted to simulate the chip separation and with Coulomb friction model at tool-chip interface. Except for Litonski-Batra model, the predicted cutting forces are within 13% of the experimental values. For the four models, the predicted chip thickness and the shear angle do not deviate from the measured ones by more than 9%. The influence on the residual stresses was more pronounced. However, this last observation is not conclusive because of the used chip separation criterion.

Yen *et al.* (2004b) analyzed the effect of different tool edge geometries (honed and chamfered edges) on process parameters by using DEFORM-2D software to simulate orthogonal cutting of 0.2% carbon steel. They showed that there is a direct relationship between the investigated tool geometries, which could be considered as tool wears, and cutting temperature, tool stresses, and chip sliding velocity.

Hua *et al.* (2005) proposed a FE model to predict residual stresses under the effect of cutting tool edge geometry, workpiece hardness, feed rate, and cutting speed during hard turning of AISI 52100. A hardness-based flow stress model is implemented in an elastic-visoplastic FE model using DEFORM-2D. The numerically predicted residual stresses were then compared to experimental ones using a chamfer with hone cutting edge under different cutting conditions and material hardness. It was concluded that using a chamfer with medium hone radius is appropriate in order to obtain desired residual stress distribution with low level of temperature and cutting force.

Filice *et al.* (2006) proposed an approach including analytical models and a pure thermal FE simulation using DEFORM-3D software to obtain a satisfactory temperature prediction in the cutting tool. The predicted temperature is found to be in good agreement with experimental one.

Outeiro *et al.* (2006) achieved an experimental and numerical investigation in order to study the effect of tool geometry, tool coating and cutting parameters (cutting speed, feed and depth of cut) on residual stress distribution induced by orthogonal cutting of AISI 316L steel. In this study, DEFORM-2D was used as finite element software and the Johnson-Cook constitutive model was employed to model the thermo-mechanical behavior of the workpiece. The authors showed that residual stresses are more sensitive to uncut chip thickness and the sequential cuts.

Filice *et al.* (2007a) achieved a rigorous investigation about the effect of friction models on the numerically predicted results such as cutting forces, tool-chip contact length, chip thickness, shear angle, and temperature distribution in the cutting tool using DEFORM-2D and 3D. In this work, Oxley model was used to represent the material behavior of the AISI 1045. The authors concluded that all the investigated parameters are almost not sensitive to friction model.

Umbrello (2008) presented a FE analysis to investigate the behavior of titanium alloy Ti6Al4V during orthogonal machining process for conventional and high cutting speed conditions. Three sets of Johnson-Cook constitutive model determined by several methods were implemented in DEFORM-2D software. It was concluded that a reliable prediction of cutting force and chip morphology can be achieved only if the material constants of Johnson-Cook are obtained at large strains, high strain rates, and high temperatures similar to those induced by machining test.

Davim *et al.* (2008) used AdvantEdge software, which is an update Lagrangian that employs explicit integration method, to investigate the effect of two different cutting tool materials (polycrystalline diamond and cemented carbide) on the thermal and mechanical behavior during orthogonal cutting simulation of aluminum alloys (Al 7075-0). The Johnson-Cook constitutive model was used to represent the material behavior of workpiece in this simulation. They concluded that the polycrystalline tool comparatively appear to be more reliable in terms of cutting and feed forces and temperature.

Mabrouki *et al.* (2008) achieved a numerical simulation using ABAQUS software in its explicit integration scheme in order to better understand the physical phenomena accompanying segmented chip formation during orthogonal cutting of Al2024-T351 alloy. The Johnson-Cook constitutive model was adopted to represent the material flow stress. In addition, an energy-based ductile failure criterion is adopted to simulate the chip formation.

Davim *et al.* (2009) achieved FEM simulation in order to compare cutting forces, cutting temperature, and plastic strain with those obtained experimentally and analytically during radial turning of AISI D2 steel using an uncoated carbide tool. In this study, the simulations were performed using AdvantEdge software and the material behavior of the workpiece was modeled by JC constitutive model. Based on this comparative investigation, it was concluded that the FEM is a good tool to simulate machining processes with reasonable accuracy.

Arrazola (2010) investigated the effect of FEM with different Arbitrary Lagrangian Eulerian (ALE) techniques and friction models at tool-chip interface on the predicted forces, temperatures and other physical quantities such as normal stress and shear stress on the tool during orthogonal cutting of AISI 4340 steel using tungsten carbide cutting tool. Johnson-Cook material model was adopted in this study. They concluded that the two ALE techniques including the friction models affect the predicted results.

In the work of Maranhão et Davim (2010), a FEM, using AdvantEdge software, was carried out in order to study the effect of the friction coefficient at tool-chip interface on the predicted results (cutting forces, cutting temperature, plastic strain, plastic strain rate, maximum shear stress, and residual stresses) during orthogonal cutting turning of stainless steel (AISI 316) using coated cemented carbide cutting tool. It was concluded that the friction coefficient has a strong effect on the FEM predictions.

Abboud *et al.* (2013) developed a predictive FE model using DEFORM-2D for orthogonal machining induced residual stress in titanium alloy Ti6Al4V. In this study the effect of cutting tool radius and cutting speed on the residual stress is investigated. It is found that compressive residual stresses are obtained when increasing feed rate and less compressive when increasing edge radius or cutting speed.

Miguélez *et al.* (2013) performed an FEM study to analyze the adiabatic shear banding in orthogonal cutting of Ti6Al4V using the commercial FE code ABAQUS/Explicit with Lagrangian formulation. In this work, the influence of yield strength coefficient “ $A$ ” and of strain hardening coefficient “ $n$ ” of the JC law on plastic shear flow stability and chip morphology was investigated. Increasing the value of  $A$  shows increase in the thermal softening and hence plastic shear flow instability which results in smaller band spacing and higher segmentation frequency. On the contrary, increasing the strain hardening coefficient has a stabilizing effect which leads to larger band spacing and lower segmentation frequency.



In the work of Davoudinejad *et al.* (2015), 2D finite element modeling is carried out in order to analyze the influence of dry and cryogenic machining of titanium alloy (Ti6Al4V) on serrated chip formation and cutting forces using AdvantEdge software. The material behavior of the workpiece, which exhibits strain rate hardening, temperature softening, strain hardening in the low strain region as well as strain softening in the high strain region, was introduced in tabular form. The friction phenomenon at the interface tool-chip was modeled using Coulomb law. It was found that by using the cryogenic machining, the cutting forces were increased slightly, but the chip segmentations and the chip thickness were reduced significantly.

Ducobu *et al.* (2016) developed a FE model using the commercial software ABAQUS/Explicit in order to highlight the influence of the material constitutive law and the chip separation criterion on the Ti6Al4V chip formation. In this work, the behavior of the workpiece (Ti6Al4V) is described by the Hyperbolic Tangent (Tanh) model which is an upgraded Johnson-Cook model to introduce strain softening in the material behavior which is one of the mechanisms leading to formation of a Ti6Al4V segmented chip. A chip separation criterion based on the temperature dependent tensile failure (hydrostatic pressure stress) is used. It was shown that the cutting forces and the chip morphology are mainly influenced by the constitutive model and chip separation criterion, respectively.

All previously mentioned FE models were carried out in 2D. By using the robust finite element codes which are able to manage 3D models, it is possible to model complex machining processes such as turning (Guo et Liu, 2002a; Outeiro *et al.*, 2008), milling (Asad *et al.*, 2013; Pittalà et Monno, 2010), and drilling (Guo et Dornfeld, 2000). Although 3D-FEM are needed to analyse some aspects in real metal cutting that cannot be investigated with 2D-FEM, they are still not widely used because of obvious limitations. High computational cost, number and complexity of elements, and remeshing algorithms (Filice *et al.*, 2007a; Miguélez *et al.*, 2013) are to name a few.

## 2.6 Summary and conclusive remarks

After the literature survey, this section outlines the major findings related to the present research work, and underlines the main issues that have not received sufficient attention and require further investigation.

- It was shown that the Johnson-Cook constitutive model could be appropriately used in FEM of machining processes and different methods have been proposed to determine Johnson-Cook constants at high strain, high strain rate, and high temperature. Although machining tests were proved by various research groups to provide accurate and reliable Johnson-Cook constants, a reduced number of experiments have been used which can affect the optimization procedure.
- It has been agreed that the rake angle is regarded as one of the most critical parameter in machining process. This is because the variation in rake angle significantly changes the thermo-mechanical loads in the cutting zone; therefore, the rake angle appears to have a significant impact on the Johnson-Cook constants when the machining tests are used as characterization approach.
- In spite of the fact that extensive studies on FEM of the orthogonal machining have been published until now, the effect of Johnson-Cook constants, obtained by machining tests at different rake angles, on the numerically predicted results was never done before.
- Using FEM in the orthogonal cutting has been widely used to predict physical quantities in the shearing zones such as strains, strain rates, temperatures, and stresses during cutting process. Only a few FEM investigations including the prediction of machined-induced residual stress with reasonable accuracy can be found in the literature. Finally, studies on residual stresses in aluminum alloys due to machining are seldom made available in the literature.

## **CHAPTER 3**

### **EXPERIMENTAL AND FINITE ELEMENT INVESTIGATIONS**

#### **3.1 Experiments**

##### **3.1.1 Orthogonal cutting tests**

Orthogonal cutting tests are conducted on three aluminum alloys using central composite design in order to minimize the experimental work. This design of experiment is presented in section 4.3.

###### **3.1.1.1 Design of cutting tests**

Orthogonal cutting tests are carried out under the following considerations:

- To satisfy the 2D orthogonal cutting, the cutting edge of the tool is positioned perpendicular to the cutting velocity and parallel to the rotational axis of the disk-shaped workpiece;
- To satisfy plane strain conditions, the ratio of the width of cut (disk thickness) to the uncut chip thickness (feed rate) is maintained to be larger than or equal to 10;
- A new insert is used after each cutting experiment in order to eliminate the effect of eventual tool wear and to avoid important changes in the cutting edge radii;
- All orthogonal cutting tests were conducted under dry condition; therefore, no cutting fluid was used during machining.

In these cutting tests, the workpieces are disks in shape having a 75 mm outside diameter, a 16 mm inside diameter, and a 3.14 mm thickness, as shown in Figure 3-1.

Uncoated and sharp carbide cutting inserts were used in all cutting experiments. The geometry and the physical properties of the tool substrate are given in Table 3-1. These cutting inserts are fixed on a left-hand holder (reference CTFPL2525M16, Kennametal Inc.) with a back rake face of  $+5^\circ$ . It is worth noting that the width of the cutting edge of the insert is larger than the workpiece thickness.

### 3.1.1.2 Experimental details

The series of machining tests were carried out using Mazak Nexus 410A, 3-axes, CNC machine with the following characteristics: a power of 25 HP, a maximum spindle speed of 12,000 rpm and, a maximum feed rate of 36 m/min. The experimental setup is schematically shown in Figure 3-3.

The cutting forces were measured using a Kistler Quartz three-component dynamometer (model 9255B). It has a measurement uncertainty of  $\pm 1$  and  $\pm 2\%$  of full scale, arising from linearity and crosstalk, respectively. A specially designed fixture for the tool holder to change the rake angle is fixed on the dynamometer (see Figure 3-4). After each cutting experiment, few chip samples were saved for thickness measurements.

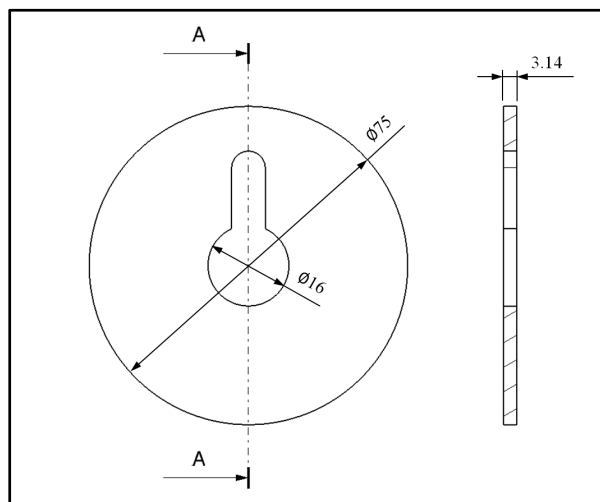



Figure 3-1 Workpiece used in machining tests (dimensions are in mm)

Table 3-1 Geometry and physical properties for the tool substrate (K68)

Reference	TPGN 160308 (K68 grade Kennametal Inc.)	
Geometry	Edge length (mm)	16.50
	Thickness (mm)	3.18
	Rake angle (°)	0
	Clearance angle (°)	11
	Cutting edge radius* (μm)	5
Physical properties	Hardness (HRA)	91
(Yen <i>et al.</i> , 2004a)	Young's modulus E (GPa)	612
	Poisson ratio $\nu$	0.22
	Density $\rho$ (kg/m <sup>3</sup> )	11,900
	Thermal conductivity K (W/m °C)	86
	Specific heat $C_p$ (J/kg °C)	337
	Thermal expansion coefficient $e$ (μm/m °C)	4.9

\* It was determined using a laser confocal microscope, see Figure 3-2.

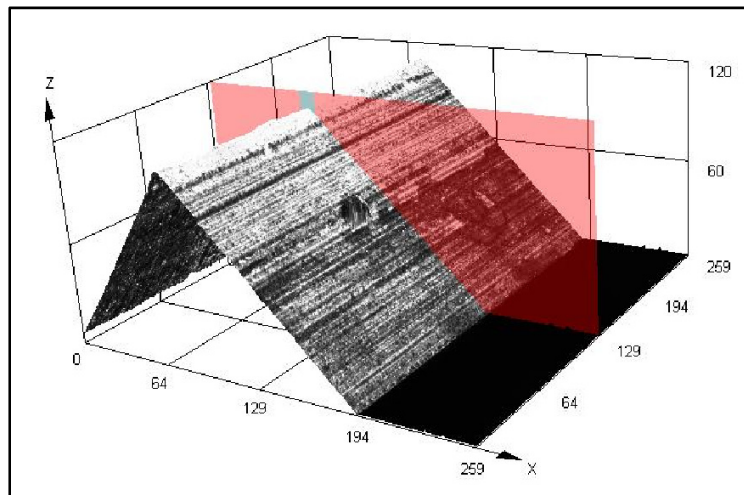


Figure 3-2 Measurement of cutting edge radius

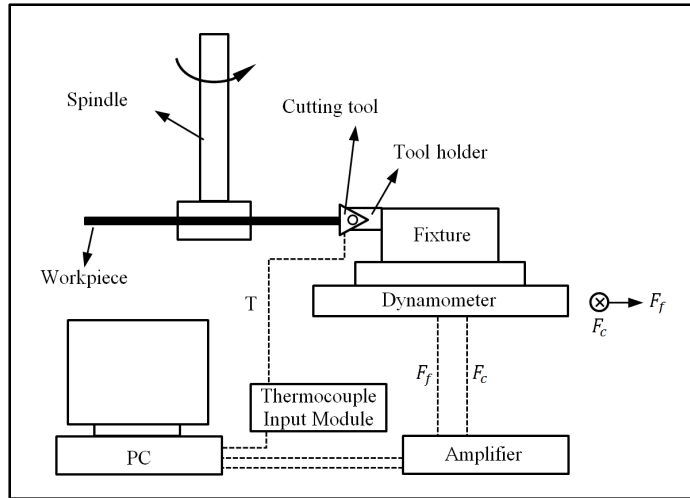


Figure 3-3 Schematic drawing of the orthogonal cutting experiment



Figure 3-4 Fixture configuration

To keep the ratio of  $w/f$  more than 10, the uncut chip thickness was selected to be  $f = 0.01$ - $0.31$  mm/rev. The 16 cutting experiments, listed in Table 4-1, were carried out. Four extra conditions were performed for the validation step. The cutting and thrust forces considered in the analysis are the average values taken in the stable period, as shown in Figure 3-5. The thickness  $t_c$  of the chips was measured by a digital micrometer. An average value at three different locations, far from the ends, was considered to represent the final chip thickness. The tool-chip contact length  $l_c$  was estimated by measuring the track on the insert rack face

after the machining tests using an optical microscope, as shown in Figure 3-6. The measurement uncertainties of  $t_c$  and  $l_c$  are neglected.

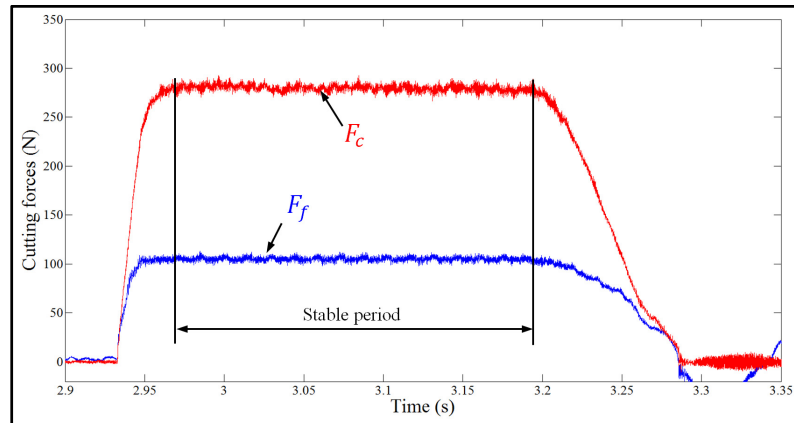


Figure 3-5 Cutting and thrust forces in time domain

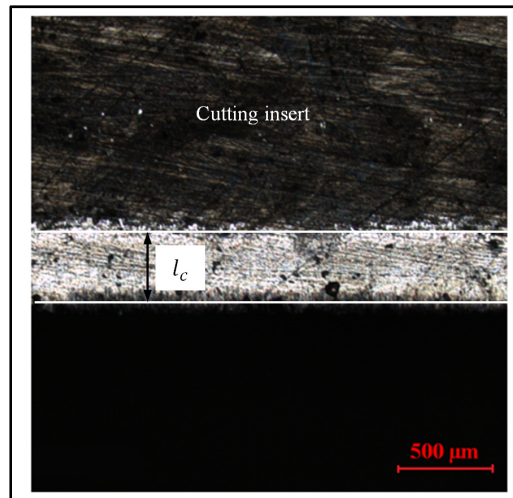


Figure 3-6 Measurement of tool-chip contact length by optical microscope

As far as cutting temperature measurement is concerned in the cutting tool, a chromel/alumel thermocouple (type K) with a diameter of 0.075 mm was utilized. The uncertainty on the temperature measurement arising from this type of thermocouple is  $\pm 1.1^{\circ}\text{C}$  or 0.4% (whichever is greater). Besides, a fine blind hole with a diameter of 0.9 mm was made in the cutting insert by means of an Electrical Discharge Machine (EDM). The diameter of the blind hole and its positions were measured by a laser confocal microscope. The depth of the

hole was measured by a Mitutoyo digital height gauge which has a measurement uncertainty within  $\pm 25\mu\text{m}$ . The thermocouple is then inserted inside the tool and the other end is connected to a data acquisition device (thermocouple module model NI 9213). This acquisition device produces a measurement uncertainty of  $\pm 1.2\text{ }^{\circ}\text{C}$ , when connected with a thermocouple type K, arising from gain errors, offset errors, differential and integral nonlinearity, noise errors, and cold-junction compensation errors.

A LabVIEW software was used to record temperature and cutting forces at sampling frequency of 100 and 24,000 Hz, respectively.

### 3.1.2 Measurements of the residual stress in the workpiece

The X-ray diffraction method ( $\text{Cr-}K_{\alpha}$  radiation) combined with the  $\sin^2\psi$  method were used to measure the residual stress state of the machined surface and sub-surface using a Proro iXRD system with a spot size of 1 mm. The following assumptions were considered:

- The workpiece material is assumed to be isotropic and homogeneous;
- Only elastic strains are considered (Hooke's law);
- Plane stress condition is assumed to exist (XRD penetrates a few microns);
- Strains and stresses are homogeneous in the irradiated volume.

The analysis of induced residual stress state on the workpiece during the cutting test requires the choice of a machined surface zone to be representative of the cutting test. In fact, the part of the workpiece corresponding to the retraction phase of the cutting tool, at the end of cutting test, is not considered for the stress analysis. Therefore, after the cutting tests, circularity profiles of the machined surface were measured using a coordinate measuring machine (CMM), MT Mitutoyo BRIGHT STRATO 7106, as depicted in Figure 3-7.

The electro-polishing technique was used to determine the in-depth residual stresses by removing successive layers of surface material without generating additional residual



stresses. In addition, the electro-polishing technique was combined with a circular mask to represent the region of analysis having a rectangular shape of  $2 \times 5$  mm. The thickness of the removed layer was measured using a Mitutoyo dial indicator with a measurement uncertainty of  $\pm 12 \mu\text{m}$ .

The gradient corrections due to the X-ray penetration were made using a commercial PROTO gradient code. Further corrections to the residual stress measurements due to the removed volume of material were made using the commercial finite element software ANSYS. It is worth noting that the workpiece was mounted on a special designed fixture in order to provide more accurate measurements, as shown in Figure 3-8.

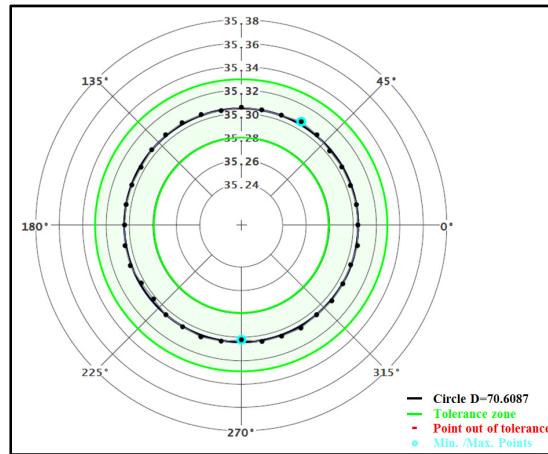


Figure 3-7 Circularity profile on the machined workpiece

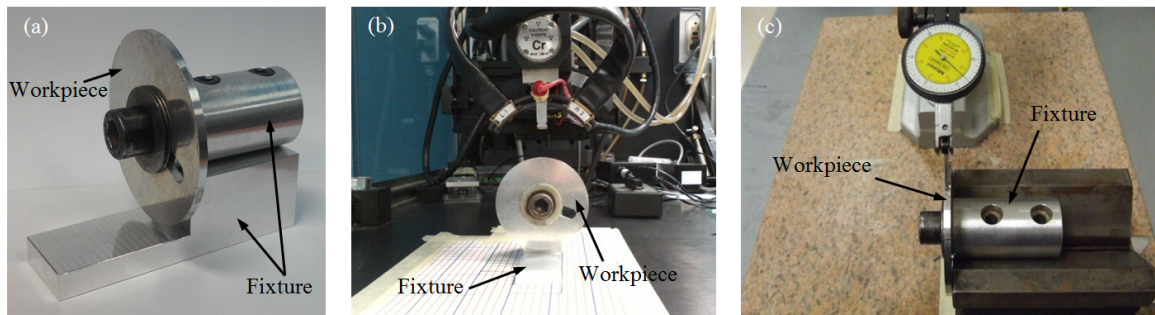


Figure 3-8 (a) Fixture configuration of the workpiece (b) Measurements of the residual stress using Proto iXRD machine (c) Measurements of removed layer thickness using Mitutoyo dial indicator

## **3.2 Finite element modeling**

Two different finite element models have been developed to simulate the machining process for current study. The first one is a thermo-mechanical simulation using Deform-2D finite element software for the cutting simulation. The second one is a pure thermal analysis using Deform-3D software for the heat transfer. The next two sections provide more detailed information on the FEM models.

### **3.2.1 Finite element model for chip formation using Deform-2D**

There are few commercial codes that are able to simulate the cutting process, such as ABAQUS, DEFORM, FORGE, AdvantEdge, ALGOR, FLUENT, and ANSYS. In recent year, DEFORM-2D has proved to be an effective code for machining simulation, because it has the following characteristics that are suited to the analysis large plastic deformation problems:

- Remeshing capability: It helps to generate a new mesh when mesh distortion is detected during large plastic deformation process. Therefore, a dense mesh can be maintained around the cutting tool tip and in the shear zones.
- Chip separation criterion is avoided: In DEFORM, the material is allowed to deform and flow naturally around the tool tip to form the chip and the machined surface. This approach needs a Lagrangian formulation with automatic remeshing technique. The interference depth between a master object (tool) and a slave object (workpiece) is used to trigger a remeshing procedure. If any portion of a master object penetrates a slave object beyond the specified interference depth, remeshing will be started (SFTC, 2012). This realistic approach used in DEFORM is different from other FEM codes in which a chip separation criterion has to be adopted to simulate the chip separation from the workpiece.

The input and output parameters of the orthogonal machining modeling are shown in Figure 3-9. Childs (1997), Sartkulvanich *et al.* (2005a), and (Shi, 2011) all agree that the material constitutive model is the most critical factor that influences the simulation results. Therefore, the impact of this factor on the simulation results should be given a particular attention.

In this study, the workpiece is modeled as an elasto-plastic body. The JC constitutive law was utilized to represent the thermal-visco-plastic behavior of the workpiece material. In addition, the Von Mises yield criteria is used in combination with the isotropic hardening rule to describe the plastic deformation of the workpiece material. In view of the high elastic modulus of the cutting tool relative to the workpiece material, the former was considered as a rigid body.

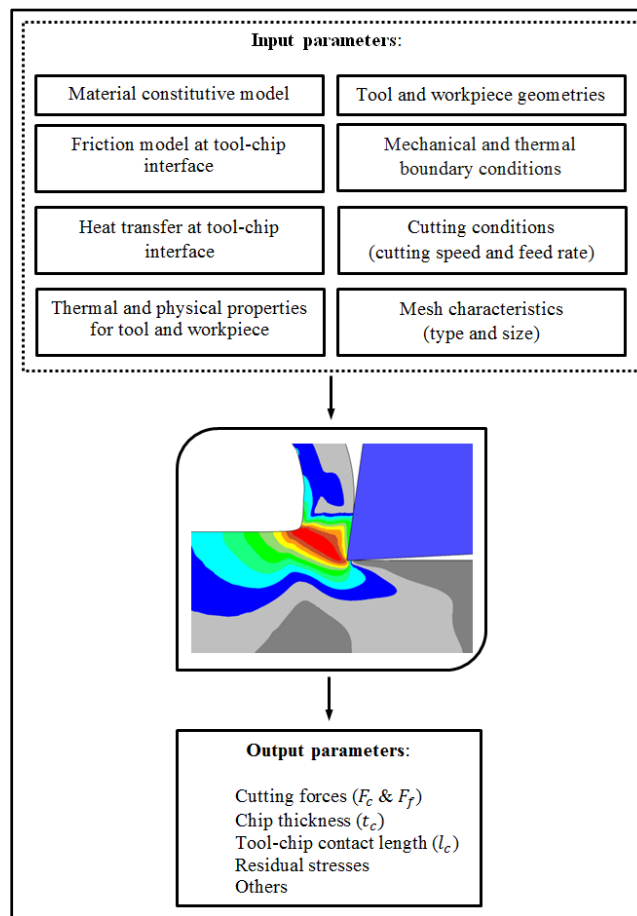


Figure 3-9 Input and output parameters of the orthogonal machining modeling

### 3.2.1.1 Finite element mesh

The initial mesh of the workpiece and the cutting tool are shown in Figure 3-10. Linear quadrilateral elements are used for both structures. In FEM of the cutting process, the meshing size and total element number rely on the particular case to be modeled. In general, using smaller element size and consequently more dense mesh in the zone of interest could provide accurate results but requires more CPU time and storage. The appropriate meshing size and total element number are determined from convergence mesh simulations such that the predicted results are not sensitive to these inputs. For example, approximately 30% change in element size or the number of elements does not cause significant difference in the simulation results (Shi, 2011).

By using the mesh windows option available in DEFORM-2D software, the zones located around the tool tip, the newly machined surface and about 200 $\mu\text{m}$  underneath this surface are modeled with a dense mesh, as shown in Figure 3-10. Therefore, the workpiece has many areas with different element edge length to be followed throughout the simulation. A similar approach is adopted for the cutting tool by applying a dense mesh to the tool tip and part of rake and flank faces. The mesh convergence study showed that when the element size is less than or equal 10  $\mu\text{m}$  in the workpiece within the uncut chip thickness, the predicted results converge rapidly, as shown in Figure 3-11. Previous work (Shi, 2011) showed that when the number of element in the workpiece within the uncut chip thickness is larger than 10, the predicted cutting forces and chip thickness converge consistently with the current results. Convergence mesh within the newly machined surface was also investigated as depicted in Figure 3-12. Based on this investigation the element size of 7  $\mu\text{m}$  is retained.

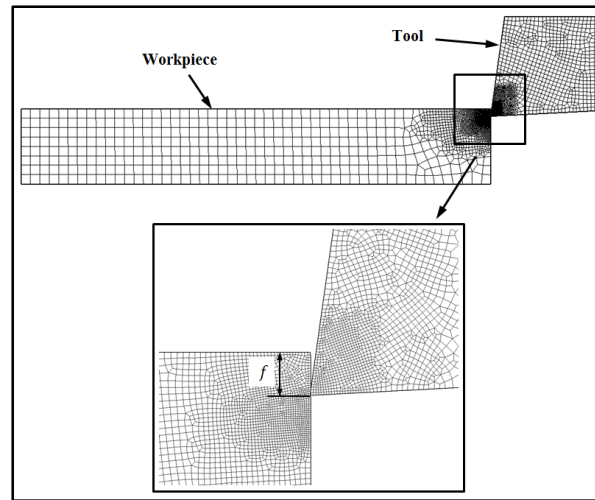


Figure 3-10 Initial workpiece and tool mesh configuration

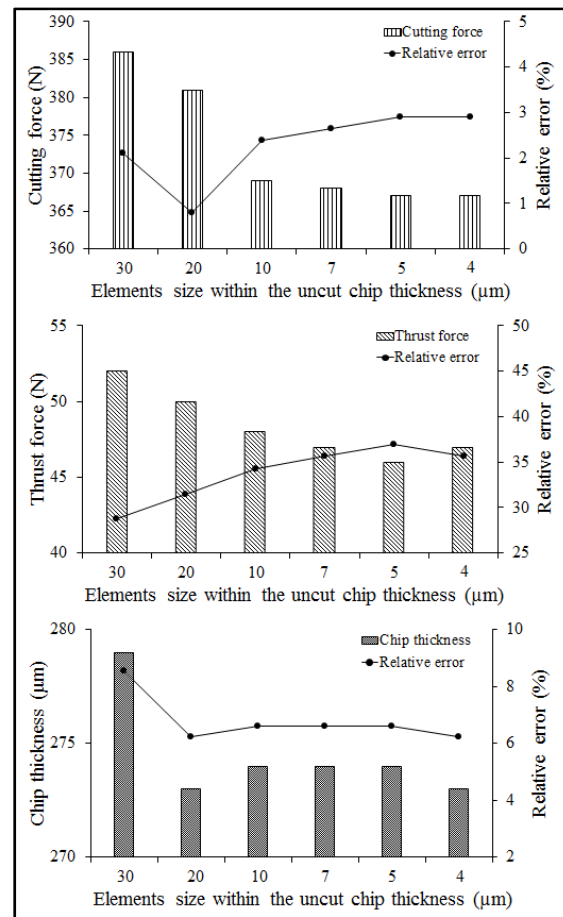


Figure 3-11 Mesh convergence within the uncut chip thickness

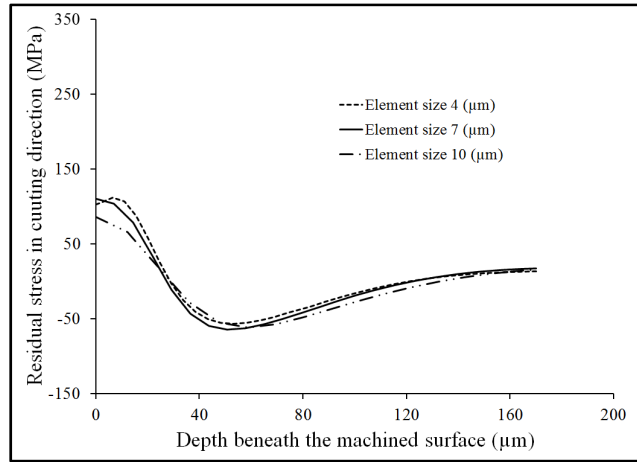


Figure 3-12 Mesh convergence within the newly machined surface

### 3.2.1.2 Boundary conditions

The kinematic boundary conditions assigned to the workpiece and the tool are presented in Figure 3-13. The rigid tool is fully constrained on the right and the top sides in the (X) and (Y) directions. The cutting speed  $V_c$  is applied to the left and the bottom sides in the (X) direction, while the movement in (Y) direction is equal to zero. The appropriate thermal boundary conditions on the workpiece and the tool are detailed in sections 5.5 and 6.5.1

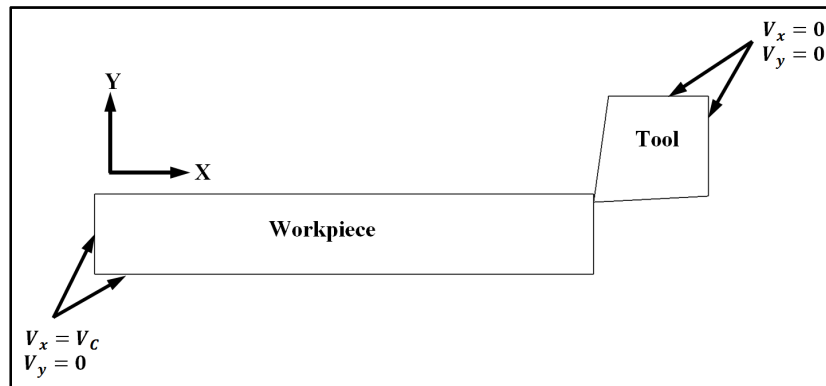


Figure 3-13 Kinematic boundary conditions of the workpiece and the tool

### 3.2.1.3 Chip formation

When the elements in the vicinity of the tool tip are highly distorted during the cutting simulation, as shown in Figure 3-14 (a), the remeshing procedure starts and a new mesh is generated, as shown in Figure 3-14 (b). At this stage, the solution information from the old mesh (stresses, strains, strain rates, and temperatures) is interpolated onto the new mesh and then the simulation continues. As a result, the chip is progressively separated from the workpiece and continues to flow over the rake face of the cutting tool, as shown in Figure 3-15. During this period of simulation, the cutting forces and temperature at the tool-chip interface increase progressively till steady state process is reached, when the workpiece has sufficiently advanced, as shown in Figure 3-16.

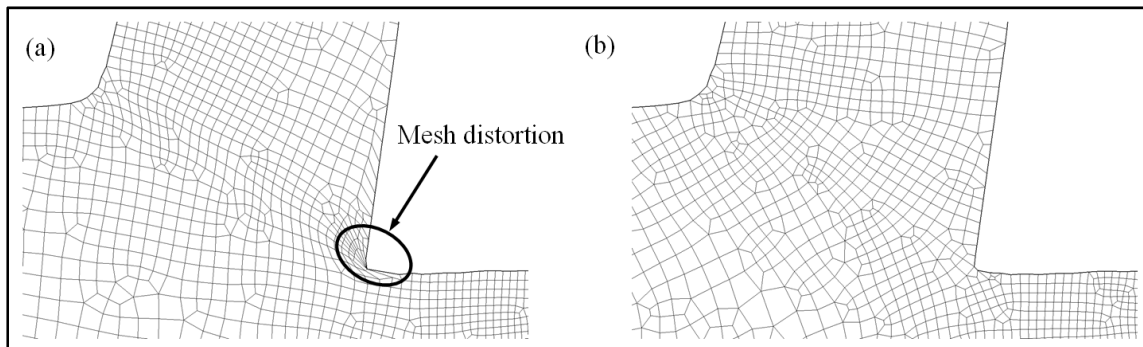


Figure 3-14 Remeshing procedure: (a) Before remeshing (b) After remeshing

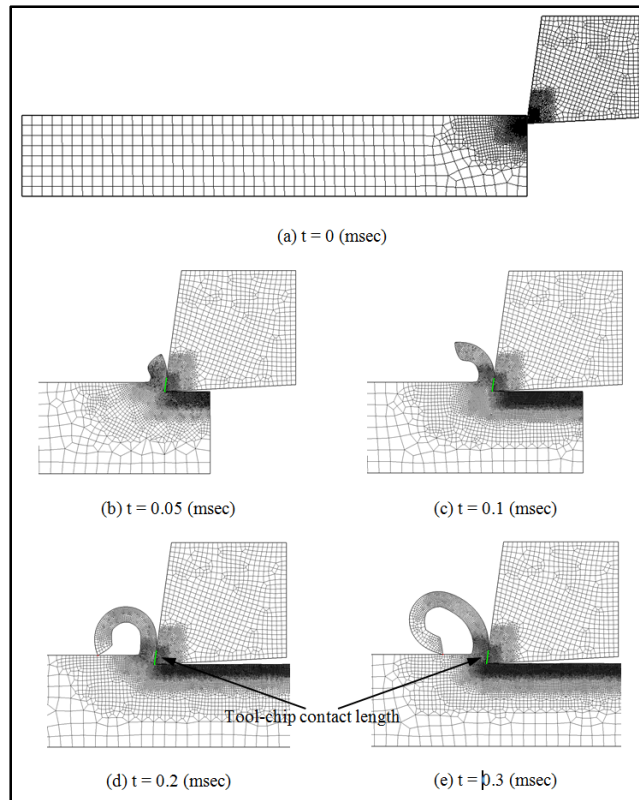


Figure 3-15 Chip formation during orthogonal cutting simulation

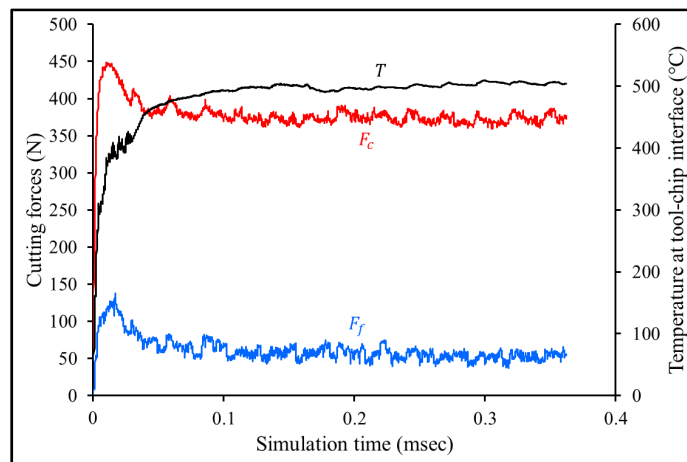


Figure 3-16 Cutting force, thrust force, and temperature versus time during orthogonal cutting simulation



### 3.2.2 Finite element model to predict temperature distribution

A very short time of high speed machining, in terms of few milliseconds, could be simulated using the coupled thermo-mechanical finite element simulation to avoid CPU time consumption. This small time makes the coupled thermo-mechanical ineffective as far as the heat transfer in the cutting tool is concerned, as shown in Figure 3-17. This Figure reports the comparison between the maximum numerically predicted temperature corresponding to the thermocouple position and the measured one when the cutting speed, the feed rate, and the rake angle were 950 m/min, 0.16 mm/rev, and 0°, respectively. It is clear that the predicted temperature, using a fully coupled thermo-mechanical analysis, is very far away the experimental one (25 versus 173 °C, respectively).

Therefore, a pure thermal simulation was proposed to obtain the temperature distribution in the cutting tool using the commercial finite element analysis (FEA) software SFTC-Deform-3D. In this case, the cutting tool was modeled by means of 500,000 solid elements with a dense mesh in the interesting zone (see Figure 3-18 for mesh convergence). Heat convection to the surroundings is considered through the surfaces that are exposed to the environment. In the proposed approach, the temperature distribution in the cutting tool is evaluated by setting the nodal temperatures obtained with the 2D thermo-mechanical numerical simulation at tool-chip contact length, as boundary conditions to the 3D thermal model, through the following steps:

- Start an orthogonal machining process modeling (2D plain strain coupled thermo-mechanical simulation) using DEFORM-2D;
- Stop the simulation when temperatures at the tool-chip interface reach steady state conditions;
- Find the nodal temperatures along the tool-chip contact length;
- Apply the collected nodal temperatures as a boundary condition for a subsequent 3D

tool model;

- Start heat transfer modeling (3D pure thermal simulation) using DEFORM-3D;
- Find the predicted temperature in the cutting tool node corresponding to the thermocouple position.

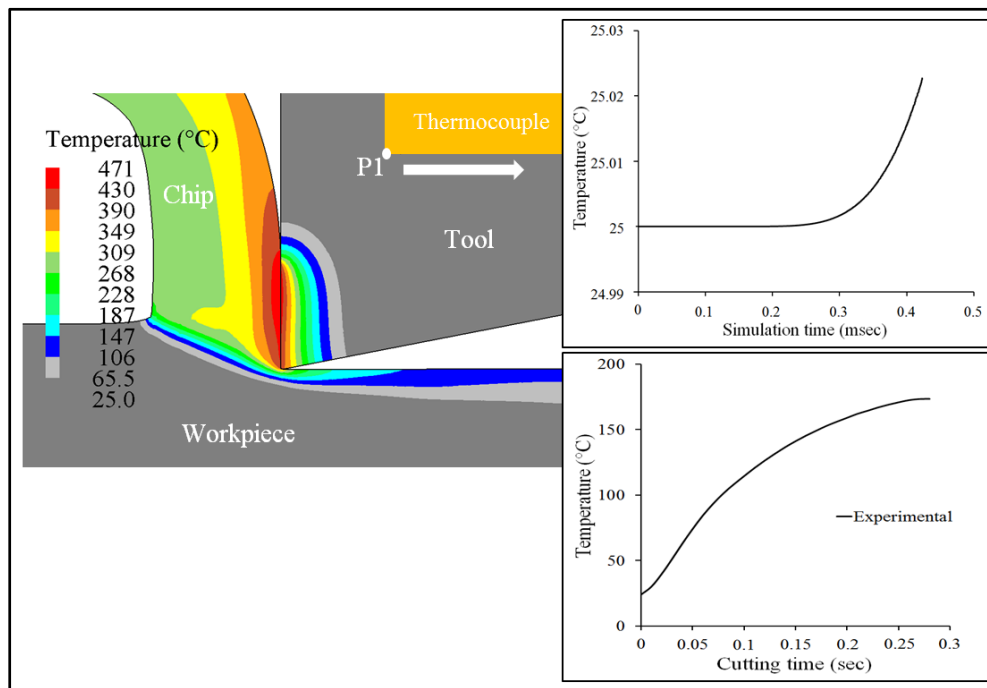


Figure 3-17 Comparison between predicted temperature and experimental one  
( $V_c=950$  m/min,  $f=0.16$  mm/rev,  $\alpha=0^\circ$ )

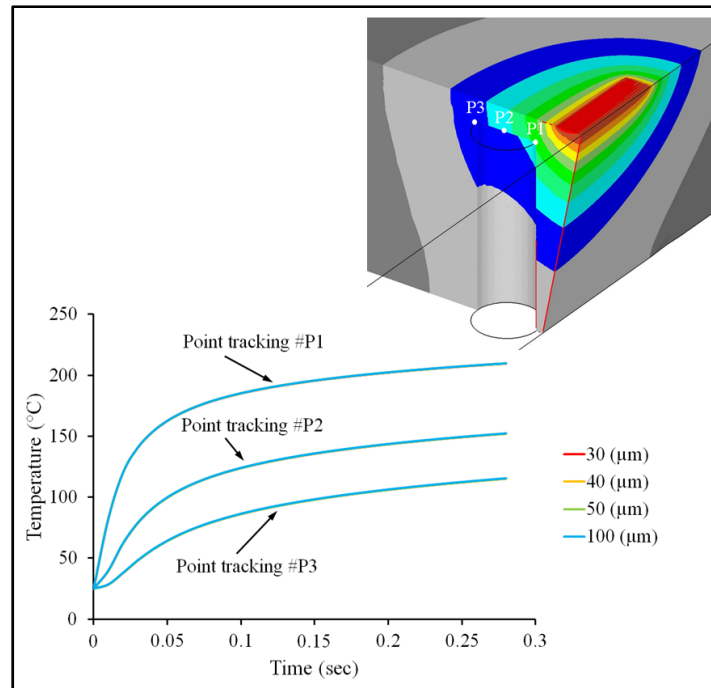


Figure 3-18 Mesh convergence  
(3D cutting tool modeling)

As mentioned before, this research work is presented as thesis by publication. As a result, three articles are presented in the chapters 4, 5, and 6 which could be summarized as follows:

In the first article, an experimental approach for the identification of the material constants used in JC constitutive law was developed for different rake angles. The use of the response surface methodology as technique to improve the existing inverse method is adopted. Machining tests were carried out on three aluminum alloys. The effect of the rake angle on the identification of the JC constitutive law was analyzed. Finally, FEM was performed to validate the proposed approach in terms of cutting forces and chip morphology.

The second article aims at investigating the effect of five sets of JC constants determined at five rake angles on the cutting forces, chip morphology, and tool-chip contact length in the machined components of Al2024-T3 alloy. These sets were implemented in a numerical machining model and the results are compared with experimental ones.

Finally, the effects of three Al2024-T3 JC constants on the numerically predicted residual stresses within the machined workpiece and the temperature of the cutting tool were the objective of the third article. To conduct such a study, two finite element models were used; 2D thermo-mechanical simulation for the chip formation and a 3D pure thermal analysis for the heat transfer.

## CHAPTER 4

### A MACHINING-BASED METHODOLOGY TO IDENTIFY MATERIAL CONSTITUTIVE LAW FOR FINITE ELEMENT SIMULATION

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#### 4.1 Abstract

The success and reliability of any finite element modeling (FEM) depend strongly on the constitutive law which describes the thermo-mechanical behavior of the machined materials. The constitutive model proposed by Johnson and Cook (JC) is widely used in the modeling of machining processes. However, different material constants of the JC constitutive law for the same material could be found in the literature which can affect significantly the predicted results (cutting forces, chip morphology, temperatures, etc.). These differences could be attributed to the different methods used for the determination of the material parameters. In the present work, an inverse approach based on response surface methodology (IABRSM) was developed to determine the parameters of the JC constitutive law. Three aluminum alloys (Al2024-T3, Al6061-T6, and Al7075-T6) were considered in the experiments, and specific attention was given to the effect of the rake angle. It is found that the material constants obtained from the proposed approach predict more accurate values of flow stresses as compared to those reported in the literature. Moreover, the FEM investigation has also shown a good agreement between predicted parameters (cutting forces and chip morphologies) and experimental ones when using material constants obtained by IABRSM.

**Keywords:** Constitutive law; Identification; Machining; RSM; FEM; Aluminum alloys

## 4.2 Introduction

The finite element modeling (FEM) of chip formation has recently become an important tool for better understanding of machining processes (Davim *et al.*, 2008; Gonzalo *et al.*, 2009). By using the high performance of advanced computers and robust finite element codes, it is possible to model complex machining processes such as turning, milling, and drilling. In fact, the success and the reliability of any FEM depend strongly on the constitutive law which describes the thermo-mechanical behavior of the machined materials. The constitutive model proposed by Johnson and Cook (JC) (Johnson et Cook, 1983) is widely used for machining simulation because it can represent material behavior as a function of strains, strain rates, and temperatures. Unfortunately, the different coefficients provided in the literature for the same material are not reliable since they affect significantly the predicted results (cutting forces, temperatures, etc.). These discrepancies could be attributed, principally, to the different methods used for the determination of the material constants. The literature review illustrates that the most common experimental methods used to identify the JC constitutive law are static tests (tensile, compression), dynamic tests (Split-Hopkinson bar technique; Taylor test), and inverse method (machining test). However, previous works (Kececioglu, 1960; Okushima et Hitomi, 1961; Shaw, 2005; Stevenson, 1997) attested that the flow stress data obtained from static tests cannot be used in metal cutting analyses due to the very low strain rates compared to those obtained in the case of machining operations.

Split-Hopkinson bar technique (SHBT) is often used in studying material behavior at high strain rates ( $10^3$  to  $10^4$  s<sup>-1</sup>) (Lesuer, 2001b). Nevertheless, the SHBT experiences some technical difficulties that could affect the accuracy of the final results (Guo, 2003). Rule (1997) has used Taylor test with strain rates up to  $10^5$  s<sup>-1</sup> for the identification of the JC constitutive equation. However, the strains encountered in this test are indeed less than 1 which are often less than those induced by machining test. Moreover, these techniques are costly, complex, and difficult to run (Panov, 2006). Although the higher strain rates and

strain achieved during dynamic tests, as compared to static tests, they are still far from representing the real thermo-mechanical loading encountered in machining. Sartkulvanich *et al.* (2005b) have attested that, in metal cutting simulation, material constants should be obtained at high strain rates (up to  $10^6 \text{ s}^{-1}$ ), temperatures (up to  $1000 \text{ }^\circ\text{C}$ ), and strains (up to 4). Recently, a new approach, based on machining tests, has been developed to identify the material constants. Shatla *et al.* (2001a) have proposed a “hybrid method” to determine the material constants of the JC constitutive equation. This method is focused on the minimization of the error between the measured cutting forces and those predicted by a computer code (OXCUT) developed by the authors. Although the shape of the chip was found serrated for the cutting conditions used to verify the determined flow stress data, the agreement between the predicted and measured cutting forces was still good for the three studied alloys. The FEM in conjunction with orthogonal cutting tests has been also proposed in the literature. The material constants will be identified by matching the cutting force obtained by FEM with the measured one (Yang *et al.*, 2011). However, these methods cannot give a unique solution (Tounsi *et al.*, 2002). In fact, the identified material constants do not represent the behavior of the workpiece material since they are affected by the secondary shear zone and by the friction model at the tool/chip contact. In addition, these estimated material constants were found to reflect only the experimental results in the range where they have been identified. Recently, Shrot et Bäker (2012) has also shown that by using the chip shape and the cutting force, it was possible to find different material constants that lead to indistinguishable chip morphology and cutting force for the same cutting condition. An alternative method known as inverse method has been recently developed based on machining tests (Guo, 2003; Limido, 2008; Ozel *et al.*, 2006; Tounsi *et al.*, 2002). In this method, the measured cutting forces and chip thickness were used to calculate, analytically, the flow stress, strain, strain rates, and temperatures in the primary shear zones. Then, the material constants were obtained using a nonlinear regression solution. Guo (2003) has used this inverse method in combination with conventional compression tests, and the results have shown that the predicted flow stresses agree with the experimental ones with reasonable accuracy. Tounsi *et al.* (2002) have found that the material constants estimated by the inverse method are in good agreement with those obtained by SHBT. Ozel *et al.* (2006) determined

the material constants using data from orthogonal cutting test and SHBT. The applicability of material constants was extended to wide ranges of strain and strain rates of SHBT tests. Limido (2008) has also used cutting tests and hot tensile test to determine the material constants of aluminum alloys. The main advantage of this approach is the identification of the constitutive law in thermo-mechanical conditions (strain, strain rate, and temperature) similar to those induced by machining tests. In summary, in all previous works, a reduced number of experiments for the identification of the constitutive law have been used which can affect the optimization procedure. Moreover, different rake angles were used during machining tests in these works:  $+6^\circ$  in (Guo, 2003);  $0^\circ$  and  $+6^\circ$  in (Tounsi *et al.*, 2002);  $-6^\circ$ ,  $-5^\circ$ , and  $+8^\circ$  in (Ozel *et al.*, 2006); and  $+20^\circ$  in (Limido, 2008). Therefore, the effect of the rake angle on the material constants requires an investigation. In the present study, an experimental procedure for the identification of the material constants was developed for different rake angles. The use of the response surface methodology (RSM) as a technique to improve the existing inverse method is adopted. Machining tests were carried out on three aluminum alloys namely Al2024-T3, Al6061-T6, and Al7075-T6. The effect of the rake angle on the identification of the constitutive law was analyzed. FEM was performed to validate the proposed approach in terms of cutting forces and chip morphology.

### 4.3 Methodology to determine material constants of Johnson-Cook

In this study, the common Johnson-Cook constitutive model was used to represent the material flow stress. This model is widely used to estimate the flow stress for a wide range of strain, strain rates, and temperatures commonly encountered in cutting processing (Johnson et Cook, 1983). The Johnson-Cook model is given as follows:

$$\bar{\sigma} = [A + B (\bar{\epsilon})^n] \left[ 1 + C \ln \left( \frac{\dot{\bar{\epsilon}}}{\dot{\bar{\epsilon}}_0} \right) \right] \left[ 1 - \left( \frac{T - T_{room}}{T_{melt} - T_{room}} \right)^m \right] \quad (4-1)$$

where  $\bar{\sigma}$  is the equivalent flow stress,  $\bar{\epsilon}$  is the plastic equivalent strain,  $\dot{\bar{\epsilon}}$  is the equivalent strain rate,  $\dot{\bar{\epsilon}}_0$  is the reference strain rate ( $1.0 \text{ s}^{-1}$ ),  $T$  is the temperature of the work material,



$T_{melt}$  is the melting point of the work material, and  $T_{room}$  is the room temperature. The material constants are as follows:  $A$  is the yield strength coefficient,  $B$  the hardening modulus,  $C$  the strain rate sensitivity coefficient,  $n$  is the hardening coefficient, and  $m$  is the thermal softening coefficient.

The material constants ( $A, B, n, C, m$ ) of the Equation (4-1) are determined using the inverse method. In this approach, the cutting forces and chip thickness are considered as input data while the output data are the physical quantities on the primary shear zone such as the equivalent flow stress, the plastic equivalent strain, the equivalent strain rate, and the cutting temperature. Based on an analysis conducted on the primary shear zone, these physical quantities were determined using the Oxley's model (Oxley et Young, 1989). A nonlinear regression solution based on the interior point algorithm was used to determine the material constants:

$$(A, B, n, C, m) = \min \left( \sum_{i=1}^N \sqrt{\left[ (A + B (\bar{\epsilon}(i))^n) \left( 1 + C \ln \left( \frac{\dot{\bar{\epsilon}}(i)}{\dot{\bar{\epsilon}}_0} \right) \right) \left( 1 - \left( \frac{T(i) - T_{room}}{T_{melt} - T_{room}} \right)^m \right) - \sigma_{exp}(i) \right]^2} \right) \quad (4-2)$$

In the present work, the use of the RSM to cover a large number of cutting conditions during the optimization procedure is proposed. The orthogonal cutting experiments are conducted using central composite design (CCD). The CCD models provide acceptable accuracy in the resolution of nonlinear responses (Montgomery, 2008). For three factors, the CCD can be represented graphically in space by points on the three-dimensional cube as shown in Figure 4-1. Each axis of the cube corresponds to a factor, and each point in space represents an experiment. According to the CCD, a total of 16 experiments have been generated including  $2^k$  ( $2^3=8$ ) factor points,  $2 \times k$  ( $2 \times 3=6$ ) star points, and two center points (one replication). The independent variables used in this study consist of rake angle ( $\alpha$ ), cutting speed ( $V_c$ ), and feed rate ( $f$ ). The upper limit of a factor was coded as +1.68 ( $\sqrt[4]{8}$ ) and the lower limit as -1.68; these values were used to calculate the machining parameters. Table 4-1 shows the results of 16 experiments forming a central composite design. It is worth mentioning that the same cutting conditions were used for machining the three tested alloys.

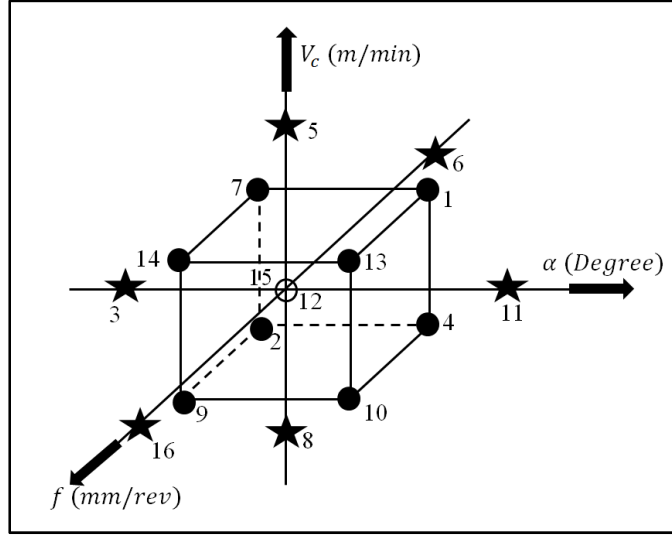


Figure 4-1 Central composite design of experiment for three factors

By using RSM and CCD, a second-order model (Equation (4-3)) has been developed with 95 % confidence level.

$$y = \beta_0 + \sum_{i=1}^k \beta_i x_i + \sum_{i=1}^k \beta_{ii} x_i^2 + \sum_{i < j}^k \beta_{ij} x_i x_j + \epsilon \quad (4-3)$$

where  $y$  is the corresponding response and  $x_i, x_j$  are the  $i$ th and  $j$ th variables related to the machining process parameters. The terms  $\beta_i, \beta_{ii}, \beta_{ij}$  are the regression coefficients and the residual  $\epsilon$  represents the experimental error of the observations. So, by using the machining parameters, such as rake angle ( $\alpha$ ), cutting speed ( $V_c$ ) and feed rate ( $f$ ), the relationship between the response surface  $y$  and these machining parameters can be formulated as follows:

$$y = \beta_0 + \beta_1 V_c + \beta_2 f + \beta_3 \alpha + \beta_{11} V_c^2 + \beta_{22} f^2 + \beta_{33} \alpha^2 + \beta_{12} V_c f + \beta_{13} V_c \alpha + \beta_{23} f \alpha \quad (4-4)$$

Table 4-1 Central composite design matrix for orthogonal cutting experiments

Test no.	Coded variables			Actual variables		
	Cutting speed $V_c$ (m/min)	Feed $f$ (mm/rev)	Rake angle $\alpha$ (°)	Cutting speed $V_c$ (m/min)	Feed $f$ (mm/rev)	Rake angle $\alpha$ (°)
1	1	-1	1	1300	0.07	5
2	-1	-1	-1	600	0.07	-5
3	0	0	-1.68	950	0.16	-8
4	-1	-1	1	600	0.07	5
5	1.68	0	0	1539	0.16	0
6	0	-1.68	0	950	0.01	0
7	1	-1	-1	1300	0.07	-5
8	-1.68	0	0	361	0.16	0
9	-1	1	-1	600	0.25	-5
10	-1	1	1	600	0.25	5
11	0	0	1.68	950	0.16	8
12	0	0	0	950	0.16	0
13	1	1	1	1300	0.25	5
14	1	1	-1	1300	0.25	-5
15	0	0	0	950	0.16	0
16	0	1.68	0	950	0.31	0

#### 4.4 Experimental details

Orthogonal machining tests were performed on Mazak Nexus 410A, 3-axes, CNC machine (spindle speed  $N_{max}$  = 12000 rpm, power  $P$  = 25 hp) under dry cutting conditions. The Mazak machine was equipped with a Kistler force dynamometer (type 9255B) plugged in a computer with LabVIEW software. A series of measurements with applied known forces in different directions were performed to calibrate the dynamometer. Sharp and uncoated carbide cutting inserts (K68 grade Kennametal Inc.) with  $11^\circ$  clearance angle were used in the cutting experiments as shown in Figure 4-2. These inserts were mounted on the left-hand tool holder CTFPL2525M16 with a back rake angle of  $+5^\circ$ . A new insert is used after each cutting experiment in order to eliminate the effect of eventual tool wear and to avoid important changes in the cutting edge radii. Disks of 75 mm in external diameter, 16 mm in internal diameter, and 3.14 mm in thickness were selected as workpieces for these experiments. All orthogonal tests were conducted on three types of aluminum alloys Al2024 -T3, Al6061-T6, and Al7075-T6, which are commonly used in aircraft applications. Four extra conditions were performed for the validation step. Once the cutting test started, the LabVIEW software, which was already prepared, was turned on to record data at 24,000 Hz. After each cutting experiment, some chip samples were saved for thickness

measurements. The thickness of these chips was then measured by a digital micrometer. An average of three measurements in three different locations was considered to represent the final chip thickness.

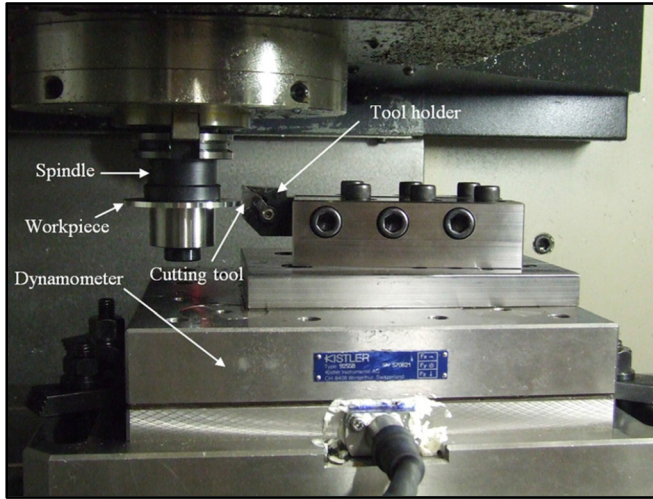


Figure 4-2 Experimental setup of the orthogonal cutting tests

#### 4.5 Finite element model and parameters

The commercial finite element software DEFORM-2D (SFTC, 2012), which is an updated Lagrangian that employs implicit integration method, was used to develop a model of the orthogonal cutting process of Al2024-T3, Al6061-T6, and Al7075-T6. The workpiece was meshed with about 10,000 isoparametric quadrilateral elements, while the tool was divided into 3200 elements. In this simulation, the workpiece was modeled as having an elasto-plastic behavior while the tool was considered as a rigid-stationary body. The JC constitutive model, Equation (4-1), which characterizes the thermo-mechanical behavior, was utilized to represent the behavior of workpiece in this simulation. The material properties for both workpiece and uncoated tool can be seen in Table 4-2. A large value of interface heat transfer coefficient  $h_{int} = 10^4$  (N/s mm °C) was used in order to obtain a fast thermal steady state in the simulations. A simple shear friction model,  $\tau_s = m_f \times K_{chip}$  (SFTC, 2012), was also considered. The choice of this shear friction model is based on recent study (Filice *et al.*, 2007a) where it is shown that the main results (i.e. forces, contact length, etc.) can be well

predicted with this friction law. Under the cutting conditions selected in this work, a slight chip undulation is observed for Al2024-T3 and Al6061-T6. Therefore, the chip formation was modeled as a continuous shearing process with no damage criterion employed for these two alloys. However, because the experimental observations revealed that there is a tendency of Al7075-T6 to form serrated chips, Cockroft and Latham's criterion,  $\int_0^{\epsilon_f} \sigma_1 d\epsilon = D$  (SFTC, 2012), was used to simulate this type of chip morphology. An iterative procedure based on chip thickness and cutting forces was employed to determine both value of  $m_f$  and  $D$  based on test #20. A damage value  $D=60$  MPa was adopted in this case and the shear coefficient  $m_f$  was found to be 0.8.

Table 4-2 Summary of physical properties for the tool substrate (K68) and workpiece material

Property	Material			Cutting tool (Yen <i>et al.</i> , 2004a) WC-CO carbide (K68)
	Workpiece (ASM, 1983)			
	Al2024-T3	Al6061-T6	Al7075-T6	
Young's modulus $E$ (GPa)	73.1	58.5	71.7	612
Poisson ratio $\nu$	0.33	0.33	0.33	0.22
Density $\rho$ (Kg/m³)	2780	2700	2810	11900
Thermal conductivity $K_{th}$ (W/ m °C)	121	167	130	86
Specific heat $Cp$ (J/ Kg °C)	875	896	960	337
Thermal expansion coefficient $e$ (× $10^{-6}$ ) (1/°C)	22.68	23.50	25.20	4.9
Melting temperature $T_{melt}$ (°C)	600	652	635	-

## 4.6 Experimental results

### 4.6.1 Second-order models

In this section, second-order models for cutting forces and chip thickness were developed based on experimental results shown in Table 4-3. The second-order response surface equations have been written, based on the CCD (Table 4-1), to determine the cutting force  $F_c$ , the thrust force  $F_f$ , and the chip thickness  $t_c$ . These analytical equations are given in terms of the machining-independent parameters: rake angle ( $\alpha$ ), cutting speed ( $V_c$ ), and feed rate ( $f$ ). The final regression coefficients determined by STATGRAPHICS software to build the mathematical models for the three materials Al2024-T3, Al6061-T6, and Al7075-T6 are

given in Table 4-4, Table 4-5, and Table 4-6, respectively. The coefficient of determination  $R^2$ , the adjusted coefficient of determination  $R_{adj}^2$ , and the  $P$  value are also presented in these tables. It is clear that in the proposed model, some terms of the response  $y$  are significant (i.e., when the  $P$  value  $< 5\%$ ) while others are insignificant (i.e., when  $P$  value  $> 5\%$ ) depending on the predicted machining parameter. However, because the present work aims to identify the material constitutive law based on a large number of cutting conditions, all terms were considered in order to not lose accuracy. Based on this, the coefficient of determination  $R^2$ , for all observed response values, is found to be between 87.77 and 99.30 5% while the adjusted coefficient of determination  $R_{adj}^2$ , is found to be between 69.43 and 98.25%. These results indicate that the second-order polynomial model gives an overall good prediction of the cutting forces and chip thickness considering the errors associated with the measurements. In order to validate the developed models, four different machining tests (#17, #18, #19, and #20, listed in Table 4-3) were performed. Figure 4-3 shows that the predicted values are in good agreement with the measured ones for the three studied alloys. Based on the above analysis, the developed models can be used to investigate a large number of cutting conditions within fixed ranges of cutting parameters such as speed, feed, and rake angle. So, in the next section, the identification procedure will be performed with 99 cutting conditions.

Table 4-3 Conditions and results of orthogonal cutting experiments performed on three aluminum alloys

Test no.	Al2024-T3			Al6061-T6			Al7075-T6		
	$F_c$ (N)	$F_f$ (N)	$t_c$ (mm)	$F_c$ (N)	$F_f$ (N)	$t_c$ (mm)	$F_c$ (N)	$F_f$ (N)	$t_c$ (mm)
1	198.9	63.4	0.13	190.5	91.8	0.163	201.9	57.1	0.167
2	143.4	132.5	0.145	92.9	177.8	0.261	150	122.4	0.188
3	431.9	168.7	0.271	387.6	229	0.426	404.1	142.3	0.271
4	216.5	95.9	0.146	208.5	123.4	0.202	220.2	79.6	0.177
5	389.9	98.3	0.243	357.6	148.1	0.323	351.3	83.2	0.236
6	54.5	46.4	0.229	30.3	30.3	0.061	54.7	46.2	0.061
7	124.6	96.5	0.158	136.1	128.7	0.218	131.1	95	0.189
8	453.8	192.3	0.304	455.5	282.4	0.553	451.1	158.6	0.324
9	487.0	214.9	0.433	406.5	330.4	0.6	380.8	167.8	0.428
10	590.9	128.7	0.4	558.6	216.8	0.523	539.6	91.9	0.465
11	378.9	73.1	0.257	344	129.7	0.378	377.5	59.4	0.332
12	415.6	119.8	0.26	399.6	188.3	0.388	400.1	101.6	0.307
13	513.8	74.5	0.326	472.6	143	0.296	435.6	63.6	0.37
14	432.7	139.3	0.323	418.4	228.2	0.641	361.6	130.9	0.284
15	414.3	122.9	0.265	401.7	193.8	0.428	401.2	107.1	0.298
16	677.5	137.6	0.393	543.4	246.1	0.645	543.9	115.8	0.342
17	397.5	110	0.253	378.7	163.6	0.321	381.3	88.5	0.297
18	490.8	131.8	0.325	472.1	202.9	0.441	452	107.6	0.33
19	279.5	105.1	0.174	265.5	144	0.292	287.4	93.6	0.242
20	428.7	146.5	0.275	407.9	203.1	0.427	413.4	119.6	0.314

Table 4-4 Model parameters for Al2024-T3

Source	$F_c$ (N)		$F_f$ (N)		$t_c$ (mm)	
	Regression coefficients	$P$ Value	Regression coefficients	$P$ Value	Regression coefficients	$P$ Value
Constant	424.215	-	121.485	-	0.266648	-
$V_c$	-18.9318	0.3081	-26.0895	0.0000	-0.02114	0.1811
$f$	174.289	0.0001	23.6894	0.0000	0.08677	0.0008
$\alpha$	20.1537	0.2895	-28.4886	0.0000	-0.00591	0.6869
$V_c \times V_c$	-11.9783	0.5782	8.25182	0.0059	-0.00254	0.8824
$f \times f$	-32.5453	0.1674	-10.7836	0.0017	0.01079	0.5521
$\alpha \times \alpha$	-20.9951	0.3806	-4.23684	0.8506	-0.00712	0.7087
$V_c \times f$	-9.78	0.6752	-7.67125	0.0120	-0.02286	0.2629
$V_c \times \alpha$	-4.805	0.8360	3.10125	0.2006	0.00084	0.9634
$f \times \alpha$	6.8025	0.7698	-10.1737	0.0033	-0.00031	0.9843
$R^2$	94.81%	-	99.3%	-	87.77%	-
$R^2_{adj}$	87.04%	-	98.25%	-	69.43%	-

Table 4-5 Model parameters for Al6061-T6

Source	$F_c$ (N)		$F_f$ (N)		$t_c$ (mm)	
	Regression coefficients	<i>P</i> Value	Regression coefficients	<i>P</i> Value	Regression coefficients	<i>P</i> Value
Constant	406.487	-	190.385	-	0.413252	-
$V_c$	-15.6447	0.3274	-35.3166	0.0001	-0.0480362	0.0392
$f$	153.693	0.0000	55.7936	0.0000	0.16155	0.0001
$\alpha$	23.3651	0.1697	-34.2181	0.0001	-0.0467022	0.0463
$V_c \times V_c$	-7.00543	0.7043	9.52306	0.0753	0.0024412	0.9150
$f \times f$	-50.3703	0.0305	-18.0211	0.0071	-0.0281144	0.2539
$\alpha \times \alpha$	-24.5134	0.2478	-3.40011	0.5075	-0.0120843	0.6306
$V_c \times f$	-12.4112	0.5415	-11.9112	0.0488	-0.0130704	0.6041
$V_c \times \alpha$	-19.9113	0.3392	5.72625	0.2808	-0.0330729	0.2156
$f \times \alpha$	4.53125	0.8211	-13.4538	0.0318	-0.0383646	0.1595
$R^2$	95.33%	-	98.67%	-	94.20%	-
$R^2_{adj}$	88.34%	-	96.69%	-	85.51%	-

Table 4-6 Model parameters for Al7075-T6

Source	$F_c$ (N)		$F_f$ (N)		$t_c$ (mm)	
	Regression coefficients	<i>P</i> Value	Regression coefficients	<i>P</i> Value	Regression coefficients	<i>P</i> Value
Constant	409.651	-	104.279	-	0.298674	-
$V_c$	-24.0439	0.1928	-17.7045	0.0003	-0.0289167	0.0476
$f$	134.981	0.0002	15.945	0.0006	0.0954789	0.0000
$\alpha$	25.25	0.1819	-27.1712	0.0000	0.0143585	0.0880
$V_c \times V_c$	-13.7931	0.5090	5.97592	0.0838	-0.0017617	0.8387
$f \times f$	-50.9741	0.0434	-8.25286	0.0306	-0.0300465	0.0118
$\alpha \times \alpha$	-20.6079	0.3726	-1.20753	0.7140	0.00705118	0.4641
$V_c \times f$	-10.765	0.6332	-1.91375	0.5653	-0.0285221	0.0196
$V_c \times \alpha$	-10.525	0.6406	1.69875	0.6086	0.00470958	0.6208
$f \times \alpha$	11.485	0.6111	-7.81125	0.0476	0.0196321	0.0727
$R^2$	92.95%	-	97.62%	-	97.60%	-
$R^2_{adj}$	82.38%	-	94.07%	-	94.02%	-



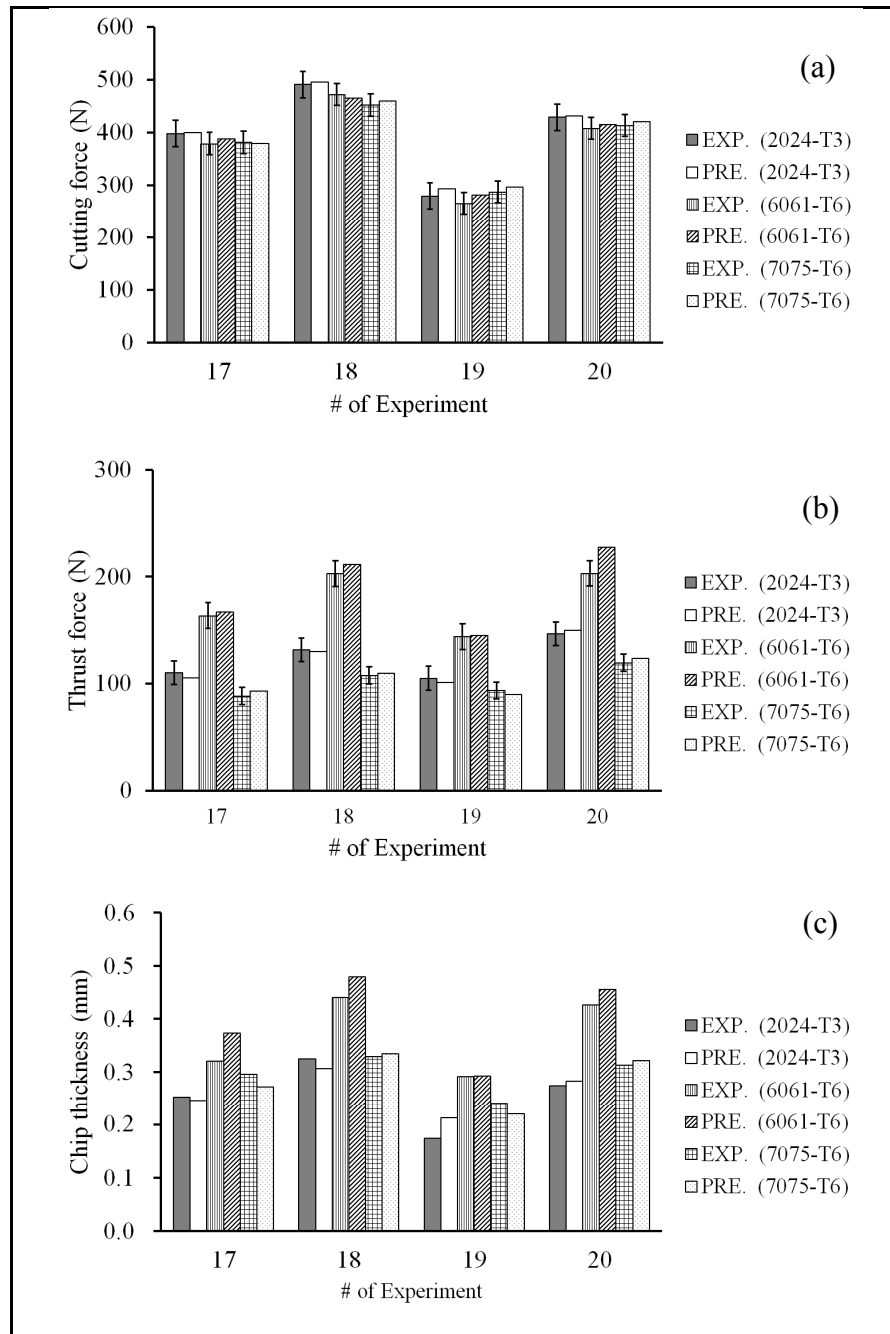


Figure 4-3 Comparison between the predicted and measured parameters: (a) cutting force, (b) thrust force, and (c) chip thickness

#### 4.6.2 Effect of the rake angle on material constants

The rake angle is considered as one of the most critical parameter in metal cutting. In fact, the chip formation mechanism could easily change from continuous to segmented chip when the rake angle changes from positive to negative values. These two mechanisms lead to different thermo-mechanical loads in the cutting zone. Thus, the rake angle appears to have a significant effect on the constitutive models when the inverse method is considered. The developed second-order models (Sect. 4.6.1), allow us to define a set of cutting conditions with fixed rake angles. Then, the obtained cutting forces and chip thickness will be used in the calculation of the physical quantities ( $\bar{\epsilon}_{AB}$ ,  $\dot{\bar{\epsilon}}_{AB}$ ,  $\bar{\sigma}_{AB}$ ,  $T_{AB}$ ) in the primary shear zone for the three tested alloys. Likewise, the effect of the rake angle on the material constants ( $A$ ,  $B$ ,  $n$ ,  $C$ ,  $m$ ) could be analyzed. A nonlinear regression solution based on the interior point algorithm is used to determine the material constants. Table 4-7 shows these constants for Al2024-T3, Al6061-T6, and Al7075-T6.

Table 4-7 Material constants

Material	Rake angle $\alpha$ (°)	$A$ (MPa)	$B$ (MPa)	$n$	$C$	$m$
Al2024-T3	-8	200	300	0.675	0.003	1
	-5	200	300	0.3	0.003	1.364
	0	257	300	0.3	0.0149	1.49
	+5	369	300	0.3	0.00409	1.49
	+8	366	300	0.3	0.003	1.49
Al6061-T6	-8	250	70	0.49	0.001	1
	-5	250	70	0.49	0.001	1.315
	0	250	79	0.49	0.0249	1.49
	+5	250	137	0.49	0.0205	1.49
	+8	250	209	0.49	0.001	1.49
Al7075-T6	-8	350	250	0.49	0.001	0.5
	-5	350	200	0.49	0.001	0.705
	0	350	250	0.49	0.001	1.478
	+5	350	266	0.49	0.001	1.49
	+8	350	250	0.49	0.001	1.49

After the determination of the material constants for the three alloys, the influence of the rake angle was verified by comparing the predicted flow stresses using Equation (4-1) and the material constants from Table 4-7. It is worth mentioning that the cutting conditions used to compare the influence of the rake angle on the flow stress are different from those used initially during the identification procedure.

The cutting data for Al2024-T3, Al6061-T6, and Al7075-T6 are listed in Tables Table 4-8, Table 4-9, and Table 4-10, respectively. The measured and predicted flow stresses for the Al2024-T36, Al6061-T6, and Al7075-T6 are plotted in Figures Figure 4-4, Figure 4-5, and Figure 4-6, respectively. Referring to Al2024-T3, the percentage differences related to these comparisons are of 35.8, 19.9, 14, 19.5, and 17.8% for the five studied rake angles  $-8^\circ$ ,  $-5^\circ$ ,  $0^\circ$ ,  $+5^\circ$ , and  $+8^\circ$ , respectively. Figure 4-5 shows the predicted and the experimental flow stress data for the Al6061-T6 alloy. The percentage differences related to these comparisons are of 28.6, 20.9, 18.3, 25.6, and 22.5% for the five studied rake angles  $-8^\circ$ ,  $-5^\circ$ ,  $0^\circ$ ,  $+5^\circ$ , and  $+8^\circ$ , respectively. Figure 4-6 shows the predicted and the experimental flow stress data for the Al7075-T6 alloy. The percentage differences related to these comparisons are of 37.4, 29.2, 27.7, 30.3, and 28.1% for the five studied rake angles  $-8^\circ$ ,  $-5^\circ$ ,  $0^\circ$ ,  $+5^\circ$ , and  $+8^\circ$ , respectively. It is clear that the difference obtained with  $0^\circ$  rake angle is comparatively smaller than the other ones. However, the percentage difference of the Al7075-T6 at  $0^\circ$  rake angle was found greater than the ones obtained for the two other alloys at the same rake angle. This discrepancy is due to the serrated chip shape obtained for Al7075-T6 which influences the accuracy of the cutting models used to calculate the physical quantities on the primary shear zone. Nevertheless, the rake angle of  $0^\circ$  gives the smallest relative error for all studied alloys. In fact, previous numerical work (Shi *et al.*, 2010a) showed that the rake angle of  $0^\circ$  gives the largest extent of the central portion of the primary shear zone. Therefore, based on this finding, one can argue that the rake angle of  $0^\circ$  is the most suitable cutting condition for the analytical model to better estimate the physical quantities ( $\bar{\epsilon}_{AB}$ ,  $\dot{\bar{\epsilon}}_{AB}$ ,  $\bar{\sigma}_{AB}$ ,  $T_{AB}$ ). Therefore, this rake angle ( $0^\circ$ ) will be considered for further studies.

Table 4-8 Cutting test data for Al2024-T3 ( $\alpha=0^\circ$ )

Test no.	$V_c$ (m/min)	$f$ (mm/rev)	$F_c$ (N)	$F_f$ (N)	$t_c$ (mm)	$\bar{\epsilon}_{AB}$	$\dot{\bar{\epsilon}}_{AB}$ ( $\times 10^5$ 1/s)	$T_{AB}$ ( $^\circ\text{C}$ )	$\bar{\sigma}_{AB}$ (MPa)
1	363	0.07	199.0	141.1	0.181	0.86	2.14	190.8	384.2
2	375	0.16	423.0	186.6	0.295	0.69	1.03	234.6	466.3
3	387	0.25	581.1	210.0	0.428	0.66	0.69	232.5	441.6
4	399	0.31	650.1	212.7	0.528	0.66	0.57	220.3	409.0
5	939	0.07	217.7	87.6	0.191	0.89	5.50	285.4	474.2
6	951	0.16	424.2	121.4	0.267	0.65	2.60	282.4	535.7
7	963	0.25	564.9	133.1	0.363	0.62	1.80	256.5	489.0
8	975	0.31	621.7	128.3	0.437	0.61	1.40	238.2	446.8
9	1503	0.31	538.7	90.2	0.341	0.58	2.52	217.1	405.6
10	1515	0.07	171.4	78.8	0.187	0.88	8.90	244.3	368.3
11	1527	0.16	360.5	100.9	0.225	0.61	4.51	246.4	471.3
12	1539	0.25	483.7	100.9	0.283	0.58	3.16	227.7	465.0

Table 4-9 Cutting test data for Al6061-T6 ( $\alpha=0^\circ$ )

Test no.	$V_c$ (m/min)	$f$ (mm/rev)	$F_c$ (N)	$F_f$ (N)	$t_c$ (mm)	$\bar{\epsilon}_{AB}$	$\dot{\bar{\epsilon}}_{AB}$ ( $\times 10^5$ 1/s)	$T_{AB}$ ( $^\circ\text{C}$ )	$\bar{\sigma}_{AB}$ (MPa)
1	363	0.07	188.1	182.6	0.289	1.26	2.05	172.8	260.1
2	375	0.16	413.3	274.1	0.499	0.99	0.95	215.4	327.1
3	387	0.25	536.8	328.8	0.651	0.86	0.64	199.4	303.9
4	399	0.31	562.6	343.8	0.720	0.80	0.54	171.4	268.6
5	939	0.07	202.5	117.3	0.225	1.02	5.41	249.9	372.4
6	951	0.16	406.4	190.3	0.413	0.86	2.45	254.4	387.3
7	963	0.25	508.8	226.4	0.544	0.76	1.63	211.9	339.6
8	975	0.31	520.1	229.4	0.599	0.71	1.36	178.2	292.2
9	1503	0.31	447.8	169.9	0.500	0.64	2.20	161.2	273.6
10	1515	0.07	178.9	103.6	0.174	0.83	8.98	236.0	375.9
11	1527	0.16	361.7	158.0	0.341	0.75	4.06	232.9	382.3
12	1539	0.25	442.8	175.6	0.451	0.68	2.71	191.9	324.0

Table 4-10 Cutting test data for Al7075-T6 ( $\alpha=0^\circ$ )

Test no.	$V_c$ (m/min)	$f$ (mm/rev)	$F_c$ (N)	$F_f$ (N)	$t_c$ (mm)	$\bar{\epsilon}_{AB}$	$\dot{\bar{\epsilon}}_{AB}$ ( $\times 10^5$ 1/s)	$T_{AB}$ ( $^\circ\text{C}$ )	$\bar{\sigma}_{AB}$ (MPa)
1	363	0.07	207.2	123.4	0.169	0.82	2.16	207.9	434.1
2	375	0.16	411.9	149.5	0.341	0.75	1.00	237.9	452.0
3	387	0.25	514.0	159.0	0.452	0.68	0.68	203.7	397.4
4	399	0.31	524.9	155.6	0.490	0.64	0.59	173.3	342.2
5	939	0.07	224.1	80.6	0.173	0.83	5.57	294.9	523.4
6	951	0.16	409.6	104.2	0.299	0.69	2.60	264.0	506.7
7	963	0.25	492.3	111.3	0.362	0.62	1.80	213.8	427.9
8	975	0.31	490.0	106.5	0.369	0.59	1.58	175.9	350.3
9	1503	0.31	392.3	89.8	0.249	0.59	2.98	136.4	243.2
10	1515	0.07	166.3	70.2	0.168	0.81	9.03	227.5	382.9
11	1527	0.16	332.5	91.3	0.246	0.63	4.38	218.3	429.5
12	1539	0.25	396.0	95.9	0.262	0.58	3.27	170.8	335.1

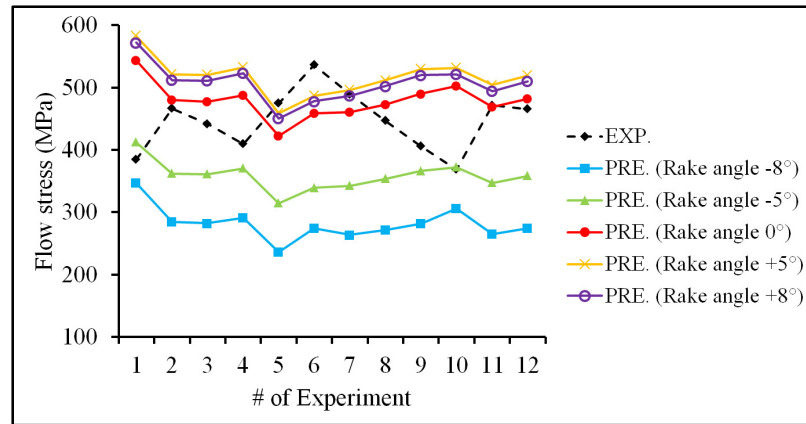


Figure 4-4 Comparison of predicted flow stresses to the experimental data for Al2024-T3

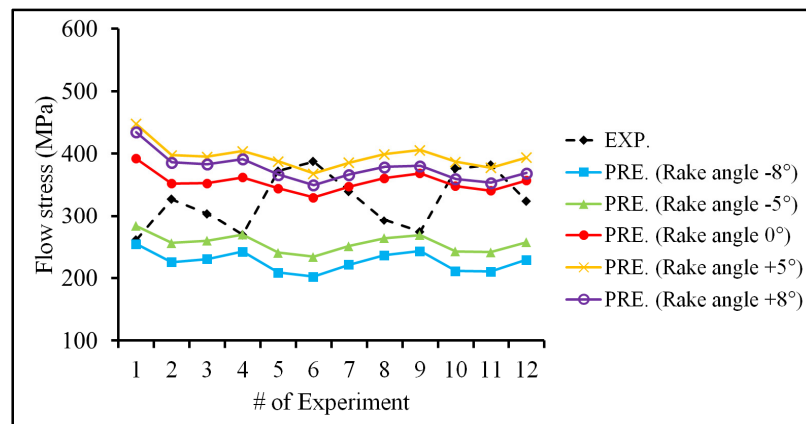


Figure 4-5 Comparison of predicted flow stresses to the experimental data for Al6061-T6

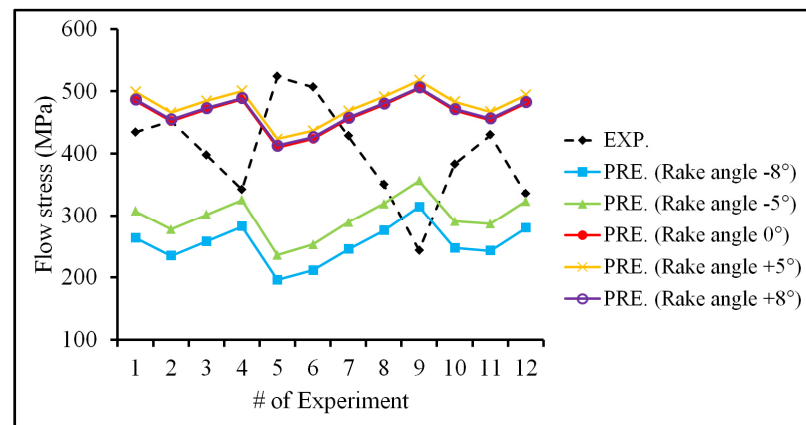


Figure 4-6 Comparison of predicted flow stresses to the experimental data for Al7075-T6

The previous Figures Figure 4-4, Figure 4-5, and Figure 4-6 show that the predicted flow stress takes the low level at  $-8^\circ$ , and it increases when the rake angle changes from  $-8^\circ$  to  $+8^\circ$ . This is because the identification of coefficients depends on the experimental flow stress which has the same behavior when changing the rake angles. This behavior is the result of the interaction between the three physical quantities  $\bar{\epsilon}_{AB}$ ,  $\dot{\bar{\epsilon}}_{AB}$ , and,  $T_{AB}$ . It is also shown that the predicted flow stress curves have the same trend for all rake angles. This is due to the independent effect of strain, strain rate, and temperature on the flow stress predicted by JC model.

#### 4.6.3 Verification of the proposed approach

The material constants, determined by this approach (inverse approach based on response surface methodology (IABRSM)), were compared with the ones obtained by different methods. The material constants of JC model obtained by RSM for each alloy are listed in Table 4-11 along with those obtained by different methods.

Table 4-11 Al2024-T3, Al6061-T6, and Al7075-T6 material constants obtained by different methods

Material	Method	Set of JC	$A$ (MPa)	$B$ (MPa)	$n$	$C$	$m$	Ref.
2024-T3	Cutting and tensile exp.	JC1	328	583	0.5	0.0026	1.98	(Limido, 2008)
	SHBT	JC2	369	684	0.73	0.0083	1.7	(Lesuer, 2001b)
	IABRSM	JC_RSM1	257.65	300	0.3	0.0149	1.49	Daoud et al.
6061-T6	Cutting and compression exp.	JC3	275	86	0.39	(*)	1	(Guo, 2003)
	SHBT	JC4	324	114	0.42	0.002	1.34	(Lesuer, 2001a)
	IABRSM	JC_RSM2	250	79.70	0.49	0.0249	1.49	Daoud et al.
7075-T6	Tensile, torsion and SHBT tests	JC5	496	310	0.3	0	1.2	(Dannemann, 2001)
	IABRSM	JC_RSM3	350	250	0.49	0.001	1.478	Daoud et al.

(\*) C was determined as function of cutting speed and feed (Guo, 2003)

Based on the estimated material constants of Al2024-T3, Al6061-T6, and Al7075-T6, the efficiency of the IABRSM was tested by using cutting experiments available in the literature and the ones listed in Tables Table 4-8, Table 4-9, and Table 4-10. The relative average errors related to the predicted flow stress are summarized in Table 4-12.

Table 4-12 Relative errors of the predicted flow stress

Material	Cutting conditions	Material constants	Error (%)
Al2024-T3	(Limido, 2008)	JC1 (Limido, 2008)	19.57
		JC2 (Lesuer, 2001b)	28.84
		JC_RSM1	27.35
Al6061-T6	RSM Table 4-8	JC1 (Limido, 2008)	62.89
		JC2 (Lesuer, 2001b)	81.09
		JC_RSM1	14.16
	(Guo, 2003)	JC3 (Guo, 2003)	17.50
		JC4 (Lesuer, 2001a)	17.94
		JC_RSM2	16.92
Al7075-T6	RSM Table 4-9	JC3 (Guo, 2003)	22.26
		JC4 (Lesuer, 2001a)	20.16
		JC_RSM2	18.37
	RSM Table 4-10	JC5 (Dannemann, 2001)	54.82
		JC_RSM3	27.74

For Al2024-T3 alloy, the validation of the material constants JC\_RSM1 was conducted with the cutting conditions reported in (Limido, 2008) and the ones listed in Table 4-8. As shown in Table 4-12, the cutting conditions reported in (Limido, 2008) gives a percentage difference of flow stress of 27.35%. This is because these cutting conditions were performed at low cutting speeds which are lower than those used to identify the material constants set JC\_RSM1. Concerning Al6061-T6, the same validation was done using the cutting conditions reported in (Guo, 2003) and the ones listed in Table 4-9 and the difference was found to be 16.92% and 18.37%, respectively. Since the cutting conditions reported in (Guo, 2003) were in the same range of those used to determine the material constants JC\_RSM2, the two differences are close. The material sets JC1, JC2, JC3, and JC4 with the cutting conditions used in this investigation give greater differences. The reason for this is that the flow stress obtained from the tension or compression tests is greater than the one obtained from cutting test (Johnson et Cook, 1983). Since there are no other cutting conditions currently available for Al7075-T6, the material constants JC\_RSM3 were only verified using cutting conditions listed in Table 4-10. Although the chip shape was found serrated for the cutting conditions used to identify the material constants JC\_RSM3, the agreement between the predicted and measured flow stress is still good with a difference of 27.74%. Bearing in mind that there is not yet a robust theory that uses serrated chips to identify the constitutive law, the machining tests can still be utilized for this purpose (Sartkulvanich *et al.*, 2005b).

Though the material constant sets obtained by the proposed method (IABRSM) slightly overestimate the flow stress, it appears that they comparatively give a much better approximation of the flow stress. However, it is obvious that there are still some differences between the predicted and experimental results. This discrepancy could be attributed to some error sources. Firstly, the IABRSM method is based on the cutting tangential force  $F_c$ , and thrust force  $F_f$ , and the chip thickness  $t_c$ . These measurements have their own errors. Secondly, although the material constants of JC are obtained from fitting the physical quantities  $(\bar{\epsilon}_{AB}, \dot{\bar{\epsilon}}_{AB}, \bar{\sigma}_{AB}, T_{AB})$ , it is worth mentioning that all models that consider only strain, strain rate, and temperature are not able to fully predict the complex material cutting behaviour. Additionally, the microstructure has considerable effect on the predicted flow stress accuracy (Guo, 2003). Thirdly, the neglected combined effect of strain rate and temperature on the flow stress predicted by the Johnson Cook model is another source of error. This means that the strain rate is independent of temperature which is not always true for most metals (Panov, 2006). Finally, the analytical cutting models of Oxley (Oxley et Young, 1989), used in this work to calculate the physical quantities, are based on simplifications and assumptions which affect the models' accuracies. Nevertheless, this comparison shows that the material constants obtained by the proposed method (IABRSM) give a small and stable percentage difference in comparison with the ones determined by other methods; the ranges of variation are [14%-27%], [16%-18%] and [ $\sim$ 27%] for Al2024-T3, Al6061-T6, and Al7075-T6 alloys, respectively. Being able to perform the identification procedure under extreme conditions and using a large number of cutting conditions is the major advantage of the proposed IABRSM method. With the use of this high speed cutting machining and under certain cutting conditions, serrated chips (non-continuous) were obtained. Since the analysis used in the proposed approach assumes a continuous chip, the accuracy of the cutting models used to calculate the physical quantities in the primary shear zone is influenced and this affects the results. The machining test results, however, can still be utilised even with non-continuous chip, as mentioned before, and the obtained results would be acceptable and the proposed methodology is still valid.



## 4.7 Finite element validation

The proposed approach (IABRSM) was validated comparing the predicted and experimentally measured cutting forces and chip thickness. In this case of validity it is worth mentioning that the cutting conditions used in FE simulations are the same as in the experimental tests.

### 4.7.1 Cutting forces

For each set of JC listed in Table 4-11, the predicted cutting tangential  $F_c$  and thrust  $F_f$  are compared to the measured ones and reported in Table 4-13. It is interesting to note that JC\_RSM gives an overall good prediction of the averaged cutting forces. In the tangential direction, the predicted tangential forces  $F_c$  are within 8.2%, 20.5% and 19.3% of experimental values for Al2024-T3, Al6061T6, and Al7075-T6 alloys, respectively. Regarding the thrust forces  $F_f$ , the best prediction is obtained with the Al7075-T6 alloy when JC\_RSM is used. However, higher difference is found with the two other alloys. This underestimation of the thrust force has also been observed by many researchers (Filice *et al.*, 2007a; Klocke *et al.*, 2013) and Deform software warns the users against this problem. This phenomenon could be attributed to the reduced number of elements in the secondary shear zone and considered as a numerical issue.

### 4.7.2 Chip morphology

Figures Figure 4-7, Figure 4-8 and Figure 4-9 illustrate the comparison between the experimental and the predicted chip morphology for Al2024-T3, Al6061-T6, and Al7075-T6 alloys, respectively. Table 4-14 lists the predicted chip thickness obtained with all sets of JC constants and for the three studied alloys. With reference to Al2024-T3 alloy, the predicted chip thickness agrees quite well with the measured one when the material constants of JC\_RSM1 are used. In this case the prediction of chip thickness is within 1% of the measured one. For Al6061-T6 alloy, the predicted chip is almost close to measured one, even if the

difference of 23.6% is obtained by the material constant set JC\_RSM2. The predicted and experimentally measured serrated chip geometries for Al7075-T3 alloy were shown in Figure 4-9. In this case, the chip peak thickness and serration frequency could be modeled very well when the material constants JC\_RSM3 are selected. Another observation could be made when comparing the irregular chip shape. As shown, the latter is more pronounced with material constants JC\_RSM3 which is the result of nonlinear effects produced by strain hardening and thermal softening (Davies et Burns, 2001).

Table 4-13 Comparison between experimental (EXP.) and predicted (FE) cutting forces  
( $V_c=650$  m/min,  $f=0.16$  mm/rev,  $\alpha=0^\circ$ )

Material	Material constants	Cutting force $F_c$		Thrust force $F_f$	
		(N)	Error (%)	(N)	Error (%)
Al2024-T3	EXP.	429	-	147	-
	FE_JC1	511	19.1	102	30.4
	FE_JC2	544	26.9	107	26.8
	FE_JC_RSM1	393	8.2	102	30.4
Al6061-T6	EXP.	407	-	203	-
	FE_JC3	331	18.5	89	56.1
	FE_JC4	313	23.2	98	51.3
	FE_JC_RSM2	324	20.5	104	48.5
Al7075-T6	EXP.	413	-	120	-
	FE_JC5	371	10.1	81	31.5
	FE_JC_RSM3	333	19.3	103.8	13.1

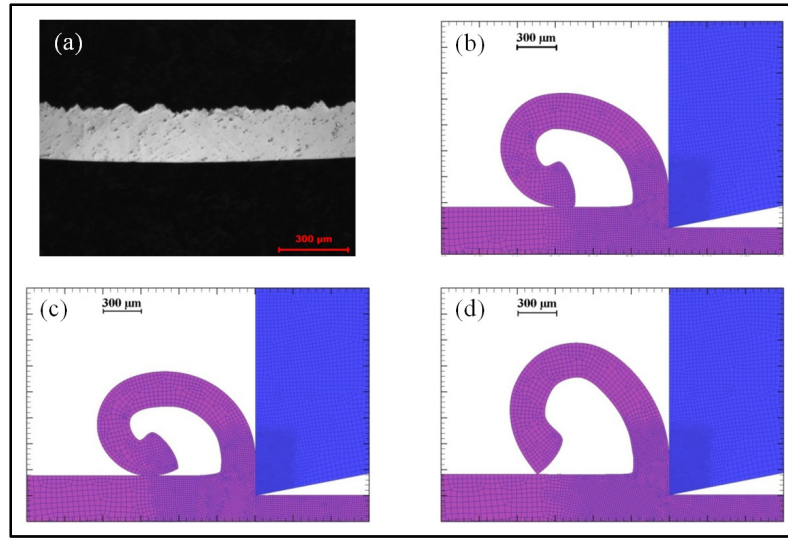


Figure 4-7 Comparison between experimental (EXP.) and predicted (PRE.) chip morphology for Al2024-T3 alloy ( $V_c=650$  m/min,  $f=0.16$  mm/rev,  $\alpha=0^\circ$ )  
(a) EXP., (b) PRE. By FE\_JC1, (c) PRE. By FE\_JC2, and (d) PRE. By FE\_RSM1

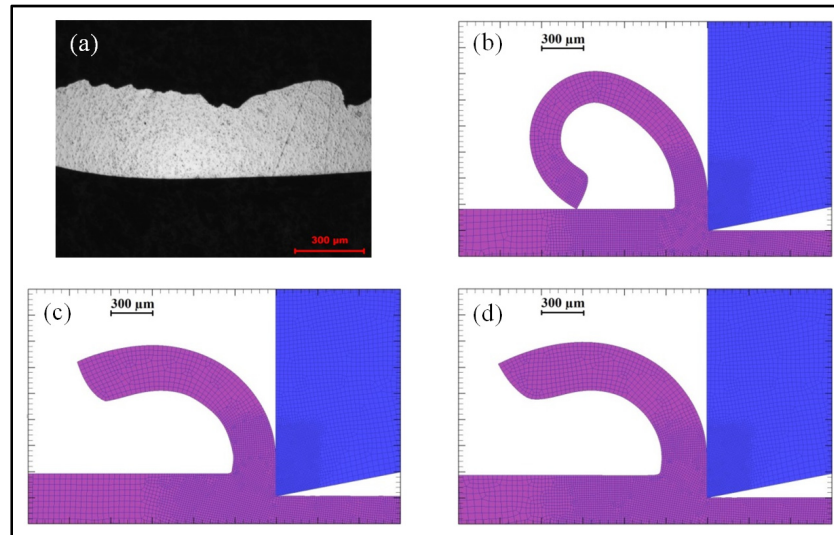


Figure 4-8 Comparison between experimental (EXP.) and predicted (PRE.) chip morphology for Al6061-T6 alloy ( $V_c=650$  m/min,  $f=0.16$  mm/rev,  $\alpha=0^\circ$ )  
(a) EXP., (b) PRE. By FE\_JC3, (c) PRE. By FE\_JC4, and (d) PRE. By FE\_RSM2

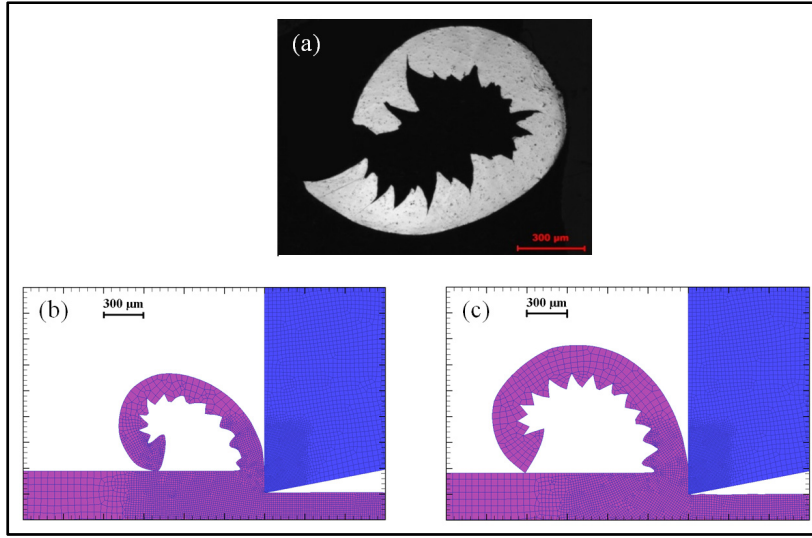


Figure 4-9 Comparison between experimental (EXP.) and predicted (PRE.) chip morphology for Al7075-T6 alloy ( $V_c=650$  m/min,  $f=0.16$  mm/rev,  $\alpha=0^\circ$ )  
(a) EXP., (b) PRE. By FE\_JC5, and (c) PRE. By FE\_RSM3

Table 4-14 Comparison between experimental (EXP.) and predicted (FE) chip thickness ( $V_c=650$  m/min,  $f=0.16$  mm/rev,  $\alpha=0^\circ$ )

Material	Material constants	Chip thickness $t_c$	
		( $\mu\text{m}$ )	Error (%)
Al2024-T3	EXP.	275	-
	FE_JC1	280	1.8
	FE_JC2	276	0.3
	FE_JC_RSM1	278	1
Al6061-T6	EXP.	427	-
	FE_JC3	240	43.8
	FE_JC4	305	28.5
	FE_JC_RSM2	326	23.6
Al7075-T6	EXP.	314	-
	FE_JC5	199	36.6
	FE_JC_RSM3	296	5.7

## 4.8 Conclusions

In the current work, an inverse method, based on orthogonal machining tests, was developed to determine the constants of the Johnson-Cook constitutive law. These material constants are determined by fitting the data from cutting tests. Orthogonal machining tests under dry cutting conditions were performed on Al2024-T3, Al6061-T6, and Al7075-T6 alloys using a

disc shape workpiece. The second-order response surface equations coupled to the central composite design were developed in this study. The high regression coefficients found indicate that the second order models can adequately predict the cutting force  $F_c$ , the thrust force  $F_f$ , and the chip thickness  $t_c$ . Based upon these results, the effect of the rake angle on the material constants was then investigated. It was shown that the rake angle  $\alpha=0^\circ$  gives the smallest relative difference of flow stress for all studied alloys as compared to other rake angles. It is argued that a possible reason that explains the better results with the  $0^\circ$  rake angle is that the cutting conditions used for the estimation of the physical quantities during the identification step are close to those assumed when the analytical model was developed. For the three studied alloys, the results show that a small and stable relative difference of flow stress is obtained using material constants JC\_RSM. The same observation could be made for the numerically predicted cutting forces and chip morphology. A possible reason could be that the material constants JC\_RSM were identified directly from the cutting tests but this is not the case for the other sets of the material constants. Material constants that are not identified over a large range of strain, strain rate, and temperature appear to be limited.



## CHAPTER 5

### **EFFECT OF RAKE ANGLE ON JOHNSON-COOK MATERIAL CONSTANTS AND THEIR IMPACT ON CUTTING PROCESS PARAMETERS OF AL2024-T3 ALLOY MACHINING SIMULATION**

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#### **5.1 Abstract**

Finite element modeling (FEM) of machining has recently become the most attractive computational tool to predict and optimize metal cutting processes. High-speed computers and advanced finite element code have offered the possibility of simulating complex machining processes such as turning, milling, and drilling. The use of an accurate constitutive law is very important in any metal cutting simulation. It is desirable that a constitutive law could completely characterize the thermo-visco-plastic behavior of the machined materials at high strain rate. The most commonly used law is that of Johnson and Cook (JC) which combines the effect of strains, strain rates, and temperatures. Unfortunately, the different coefficients provided in the literature for a given material are not reliable since they affect significantly the predicted results (cutting forces, temperatures, residual stresses, etc.). In the present work, five different sets of JC are determined based on orthogonal machining tests. These five sets are then used in finite element modeling to simulate the machining behavior of Al2024-T3 alloy. The effects of these five different sets of JC constants on the numerically predicted cutting forces, chip morphology, and tool-chip contact

length are the subject of a comparative investigation. It is concluded that these predicted cutting parameters are sensitive to the material constants.

**Keywords:** Johnson-Cook constitutive law; FEM; Identification; Inverse method; Machining; Al2024-T3

## 5.2 Introduction

Finite element modeling has become a vital and irreplaceable tool to simulate machining processes (Davim *et al.*, 2008; Gonzalo *et al.*, 2009). To achieve a successful machining simulation, an accurate constitutive law model is required (Seshadri *et al.*, 2013). However, one can find in the literature several constitutive law models that are adopted for machining simulation. The constitutive model proposed by Johnson and Cook (JC) (Johnson et Cook, 1983) is the most popular model used for machining simulation because it can represent material behavior as a function of strains, strain rates, and temperatures. This constitutive model, like other models, depends heavily on its material model parameters and the method used to determine them. The literature review illustrates that the most common experimental methods used to identify the JC constitutive law constants are dynamic tests (Split-Hopkinson bar technique (SHBT) and Taylor test), inverse method (machining test), and combinations between them. Lesuer (2001b) has used data from Split-Hopkinson bar technique (SHBT) to determine the first three parameters of the JC model that are required for the elasto-plastic term at high strain rate ( $10^3$  to  $10^4$  s<sup>-1</sup>). Rule (1997) developed a numerical approach to extract material strength coefficients from Taylor test data with strain rates up to  $10^5$  s<sup>-1</sup>. Even though these strains and strain rates achieved by these dynamic tests are high, they are still far from those encountered in machining (Sartkulvanich *et al.*, 2005b). Moreover, these techniques are costly, complex, and difficult to run (Panov, 2006). Recently, a new approach, based on finite element method in conjunction with orthogonal cutting tests has been proposed (Yang *et al.*, 2011). In this approach, the material constants are determined by matching the cutting force obtained by finite element modeling (FEM) with the measured one. In addition of the approach being time consuming, there are two



limitations associated with this type of method. First, the material constants should be identified from the same range of experimental machining conditions to which the simulation model will be applied. Second, these determined constants are only applicable for that specific commercial finite element software with which this method is applied. Moreover, this method cannot give a unique solution since it is affected by the secondary shear zone and by the friction model at tool-chip interface (Tounsi *et al.*, 2002). Shroff et al. (2012) have also shown that based on the chip shape and the cutting force, it was possible to find different sets of material constants that lead to indistinguishable results such as chip morphology and cutting force for the same cutting condition. An alternative method known as the inverse method has also been developed based on machining tests used for characterization (Guo, 2003; Limido, 2008; Ozel *et al.*, 2006; Tounsi *et al.*, 2002). The main advantage of such an approach is that extreme conditions such as strain rates up to  $10^6 \text{ s}^{-1}$ , temperature up to  $1000^\circ\text{C}$  and strains up to 4 are achieved directly with machining tests. Unfortunately, the above mentioned methods give different material strength coefficients which can affect the predicted cutting parameters such as cutting forces, chip morphology, temperature distributions, and residual stresses. Recently, Filice *et al.* (2007a) concluded that contact length, deformed chip thickness, and shear angle strictly depend on the material rheology. Umbrello *et al.* (2007b) studied the effects of JC constants of the same material obtained from different methods on the previously described parameters. They observed that in general, most of these parameters are reasonably well predicted when using material constants set obtained from machining tests. Different rake angles were used by different researchers during the machining tests to obtain the JC material strength coefficients:  $0^\circ$  and  $+6^\circ$  in (Tounsi *et al.*, 2002);  $+6^\circ$  in (Guo, 2003);  $-6^\circ$ ,  $-5^\circ$ , and  $+8^\circ$  in (Ozel *et al.*, 2006); and  $+20^\circ$  in (Limido, 2008). Despite the different rake angles used in the previous works, none of them has studied the effect of the rake angle on the constants used in the constitutive flow stress model. This is a key issue since the chip formation mechanism could easily change from continuous to segmented chip when the rake angle changes from positive to negative values. These two mechanisms, therefore, lead to different thermo-mechanical loads in the cutting zone. As a result, the rake angle appears to have a significant effect on the constitutive models when the inverse method is considered. In consequence, the effect of the

rake angle on the constants used in the constitutive law requires an investigation. This work aims to find the optimum rake angle at which the JC constants should be identified by machining tests. To achieve this goal, five sets of JC material constants were therefore determined by experimental machining tests at five different rake angles using inverse approach based on response surface methodology (IABRSM). These constitutive law constants were then employed to simulate the machining behavior of Al2024-T3 alloy using FEM.

### 5.3 Identification procedure of material constants of Johnson-Cook

The Johnson-Cook constitutive flow stress model (Johnson et Cook, 1983) adopted in this study to represent the material flow stress as a function of strains, strain rates, and temperatures is defined by:

$$\bar{\sigma} = [A + B (\bar{\epsilon})^n] \left[ 1 + C \ln \left( \frac{\dot{\bar{\epsilon}}}{\dot{\bar{\epsilon}}_0} \right) \right] \left[ 1 - \left( \frac{T - T_{room}}{T_{melt} - T_{room}} \right)^m \right] \quad (5-1)$$

where  $\bar{\sigma}$  is the equivalent flow stress,  $\bar{\epsilon}$  is the plastic equivalent strain,  $\dot{\bar{\epsilon}}$  is the equivalent strain rate,  $\dot{\bar{\epsilon}}_0$  is the reference strain rate,  $T$  is the temperature of the work material,  $T_{melt}$  is the melting point of the work material, and  $T_{room}$  is the room temperature. The material constants are as follows:  $A$  is the yield strength coefficient,  $B$  the hardening modulus,  $C$  the strain rate sensitivity coefficient,  $n$  is the hardening coefficient, and  $m$  is the thermal softening coefficient.

The determination of the material constants ( $A, B, n, C, m$ ) of the Equation (5-1) is done on the basis of the proposed approach developed by (Daoud *et al.*, 2015b) which is summarized in Figure 5-1. Based on this approach, the effect of the rake angle on the material constants ( $A, B, n, C, m$ ) could be analyzed under a large number of cutting conditions.

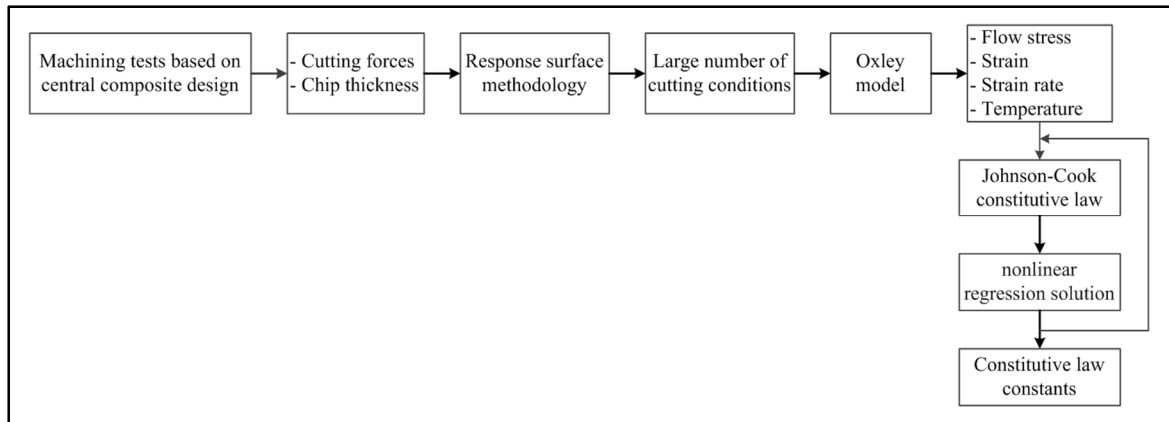


Figure 5-1 Inverse approach based on response surface methodology (IABRSM)

Table 5-1 shows the five different sets of workpiece material constants that were identified experimentally using five different rake angles. As mentioned before, the JC(-8°), JC(-5°), JC(0°), JC(+5°), and JC(+8°) sets were determined based on IABRSM applied to the orthogonal machining tests results and with rake angles -8°, -5°, 0°, +5°, and +8°, respectively. More detailed information on the IABRSM used to determine these constants can be found in (Daoud *et al.*, 2015b). To give a clear comparison between the five sets of JC, it is more convenient to represent the JC constants versus the rake angle, instead of the more commonly used table to list these constants as shown in Figure 5-2.

Table 5-1 Material constants for Al2024-T3  
(Daoud *et al.*, 2015b)

Set of JC	$A$ (MPa)	$B$ (MPa)	$n$	$C$	$m$	$\dot{\epsilon}_0$ (s <sup>-1</sup> )
JC(-8°)	200	300	0.675	0.003	1	1
JC(-5°)	200	300	0.3	0.003	1.36	1
JC(0°)	257.6	300	0.3	0.0149	1.49	1
JC(+5°)	369.9	300	0.3	0.004	1.49	1
JC(+8°)	366.4	300	0.3	0.003	1.49	1

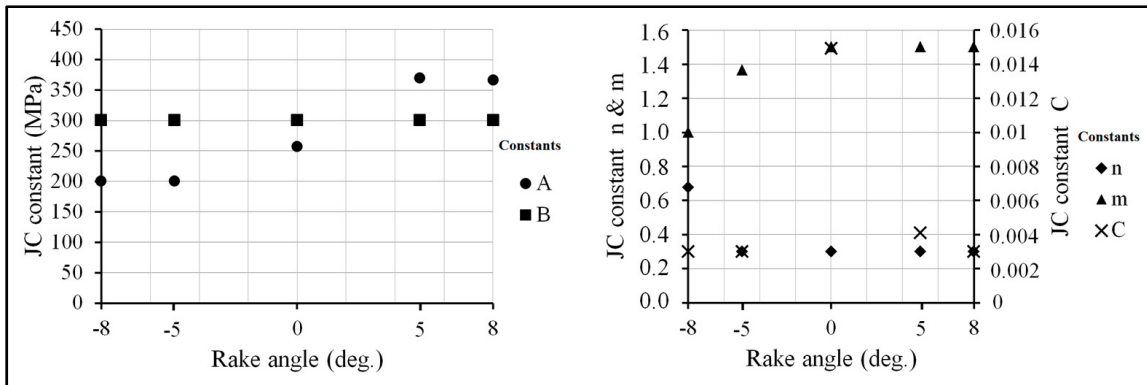


Figure 5-2 Comparison between the five sets of JC: effect of rake angle

#### 5.4 Experimental setup

Orthogonal cutting tests using sharp and uncoated carbide cutting inserts K68 grade of Kennametal Inc. with  $11^\circ$  clearance angle were carried out on Mazak Nexus 410A 3-axis CNC machine with a spindle speed of 12,000 rpm and a power of 25 HP under dry cutting conditions, as shown in Figure 5-3. A Quartz three-component dynamometer (model 9255B), connected to a computer-based data acquisition system that runs under LabView software, was mounted on the Mazak machine. The calibration of the dynamometer was performed by a series of measurements with known applied forces in the three special directions. The inserts were mounted on the left-hand tool holder, CTFPL2525M16 with a back rake angle of  $+5^\circ$ . A new edge is used for each cutting experiment in order to avoid tool wear and to keep the cutting edge as sharp as possible. The workpieces are disks in shape having a 75 mm outside diameter, a 16 mm inside diameter, and a 3.14 mm thickness. The machining tests were conducted on Al2024-T3 alloy which is commonly used in aircraft applications. Once the cutting test started, the amplified signals of the components of the cutting force was monitored and recorded in the computer. The sampling frequency of data was held at 24,000 Hz and average values of the force components were considered in the analysis. The thickness of the chips was measured by a digital micrometer. An average value at three different locations, far from the ends, was considered to represent the final chip thickness. Finally, an optical microscope was used to estimate the tool-chip contact length by measuring the track on the insert rack face after the machining tests.

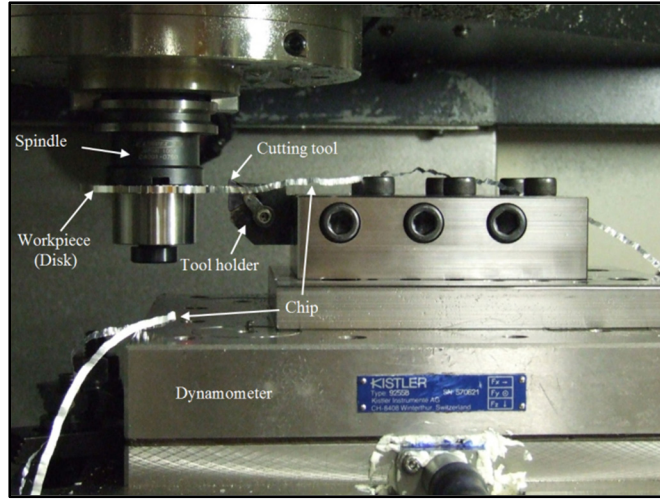


Figure 5-3 Experimental setup utilized during the orthogonal cutting tests

## 5.5 Finite element machining simulation

In this study, finite element (FE) modeling was used to simulate the orthogonal dry cutting process of Al2024-T3 using FEA software DEFORM-2D (SFTC, 2012). This commercial software is an updated Lagrangian that employs implicit integration method. This code has optimized remeshing capability to overcome element distortions due to large deformation in the shear zones. Therefore, the chip separation criterion and associated drawbacks could be avoided (Ee *et al.*, 2005). In this simulation, the workpiece, modeled with elasto-plastic behavior (8 mm long×1.6 mm high), was meshed with over 10,000 isoparametric quadrilateral elements with 4 integration points. The tool, considered as a rigid-stationary body, was divided into 3200 elements. The material properties of the workpiece and the uncoated tool are given in Table 5-2. A large value of the interface heat transfer coefficient  $h_{int} = 10^4$  (N/sec mm °C) was used to reach the thermal steady state faster in the simulations (Filice *et al.*, 2007a). Since aluminum alloys are susceptible to adhesion with tool at tool-chip interface (Roy *et al.*, 2009), a constant shear friction model,  $\tau_s = m_f \times K_{chip}$  (SFTC, 2012), was also considered with the shear friction coefficient equal to 0.6. Under the cutting conditions selected in this work and listed in Table 5-3, a slight chip undulation is observed for machining tests (no. 1, no. 3, no. 4, and no. 5). Therefore, the chip formation was modeled as a continuous material shearing process with no damage criterion applied for these

tests. However, since the experimental test no. 2 showed tendency of Al2024T3 to form serrated chips, the Cockroft and Latham's (Cockroft *et al.*, 1966) damage criterion was used to simulate this type of chip morphology. Cockroft and Latham's criterion is expressed as follows:

$$\int_0^{\varepsilon_f} \sigma_1 d\varepsilon = D \quad (5-2)$$

where the  $\varepsilon_f$  is the effective strain,  $\sigma_1$  is the maximum principal stress,  $D$  is the critical damage value. The basic idea behind this criterion is that chip segmentation occurs when a critical damage value  $D$  is reached. A damage value  $D=100$  MPa was used in this study.

The model setup and boundary conditions are shown in Figure 5-4. The workpiece and tool edges that are sufficiently far from the deformation zone are maintained at the room temperature  $T_{room}=25$  °C. In addition to conduction, convection heat transfer is also considered and applied on free sides of heated chip and tool. The convection coefficient imposed is based on the assumption of the chip motion through stagnant air. By using the mesh windows option available in DEFORM-2D software, the deformation zone and the newly machined surface are modeled with a dense mesh. Therefore, the workpiece has two areas with different element edge size: 10  $\mu\text{m}$  at the machined surface and 48  $\mu\text{m}$  for the rest of workpiece. Similarly, a dense mesh was considered at the tool tip and a refined mesh was applied to parts of the rake and flank faces with 9  $\mu\text{m}$  element edge size. This meshing size approach was adopted to reduce CPU time without losing accuracy. An interference depth of 6  $\mu\text{m}$  is used to start a remeshing procedure for the workpiece. During the FE simulation, the workpiece moves towards the stationary cutting tool at a cutting speed  $V_c$  and a feed rate  $f$ .

Table 5-2 Physical properties of the workpiece material and the tool substrate (K68)

Property	Material	
	Workpiece (ASM, 1983) Al2024-T3	Cutting tool (Yen <i>et al.</i> , 2004a) WC -CO carbide (K68)
Young's modulus $E$ (GPa)	73.1	612
Poisson ratio $\nu$	0.33	0.22
Density $\rho$ (kg/m <sup>3</sup> )	2780	11900
Thermal conductivity $K_{th}$ (W/m °C)	121	86
Specific heat $C_p$ (J/ Kg °C)	875	337
Thermal expansion coefficient $\alpha$ ( $\times 10^{-6}$ ) (1/°C)	22.68	4.9
Melting temperature $T_{melt}$ (°C)	600	-

Table 5-3 Cutting conditions

# of test	Rake angle (deg.)	$V_c$ (m/min)	$f$ (mm/rev)
1	-8	950	0.16
2	-5	600	0.25
3	0	950	0.16
4	+5	600	0.25
5	+8	950	0.16

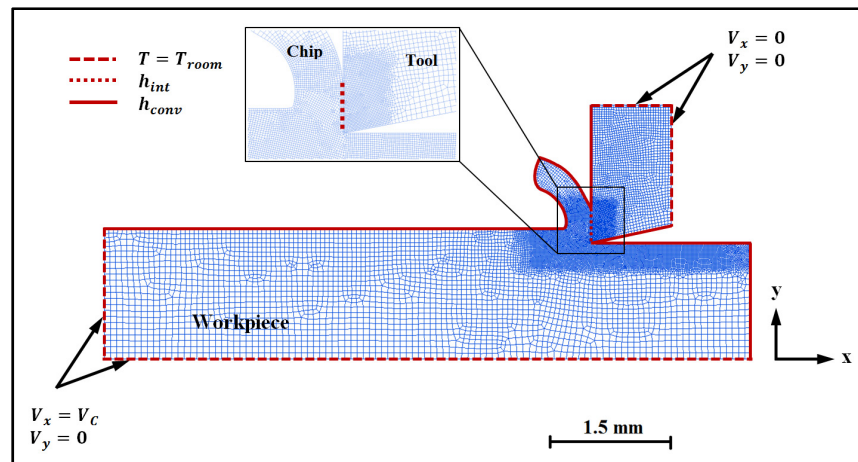


Figure 5-4 Displacement and thermal boundary conditions of the 2D FE model

As mentioned before, the JC constitutive law, Equation (5-1), was utilized to model the thermal-visco-plastic behavior of the workpiece material.

Figures Figure 5-5 and Figure 5-6 show the effect of the temperature and the strain as well as the temperature and the strain rate on the material flow stress for Al2024-T3 using the five sets of JC. On the one hand, it can be seen from Figure 5-5 that the temperature and the strain have a significant effect on the flow stress. On the other hand, at a strain of 1.5, a temperature of 400°C, and a variation of strain rate between  $10^2$  to  $10^5$  s<sup>-1</sup>, the predicted flow stress increases from 209 to 215 MPa for JC(-8°) (Figure 5-6 (a)), from 241 to 249 MPa for JC(-5°) (Figure 5-6 (b)), from 301 to 340 MPa for JC(0°) (Figure 5-6 (c)), from 341 to 354 MPa for JC(+5°) (Figure 5-6 (d)), and from 338 to 347 MPa for JC(+8°) (Figure 5-6 (e)). This shows that while the strain rate has a limited effect on the flow stress for Al2024-T3 alloy as compared to other metal alloys (Fang, 2005), there is a wide variation between different JC constants. It is clear also that the flow stress is low level at -8° and it increases when the rake angle changes from -8° to +8°. This is because the identification of coefficients of JC depends on the experimental flow stress which has the same behavior with changing the rake angles. This behavior is the result of the interaction between the three physical quantities such as  $\bar{\epsilon}_{AB}$ ,  $\dot{\bar{\epsilon}}_{AB}$ , and  $T_{AB}$ .

In the current study, both numerical model and experimental test are based on dry orthogonal cutting using a sharp and uncoated carbide cutting tool. To evaluate the effect of five sets of JC constitutive law, the FE simulations and the experimental tests were carried out for five rake angles -8°, -5°, 0°, +5°, and +8°. As to the cutting conditions used in this study, the rake angle, the cutting speed, and the uncut chip thickness were given in Table 5-3.



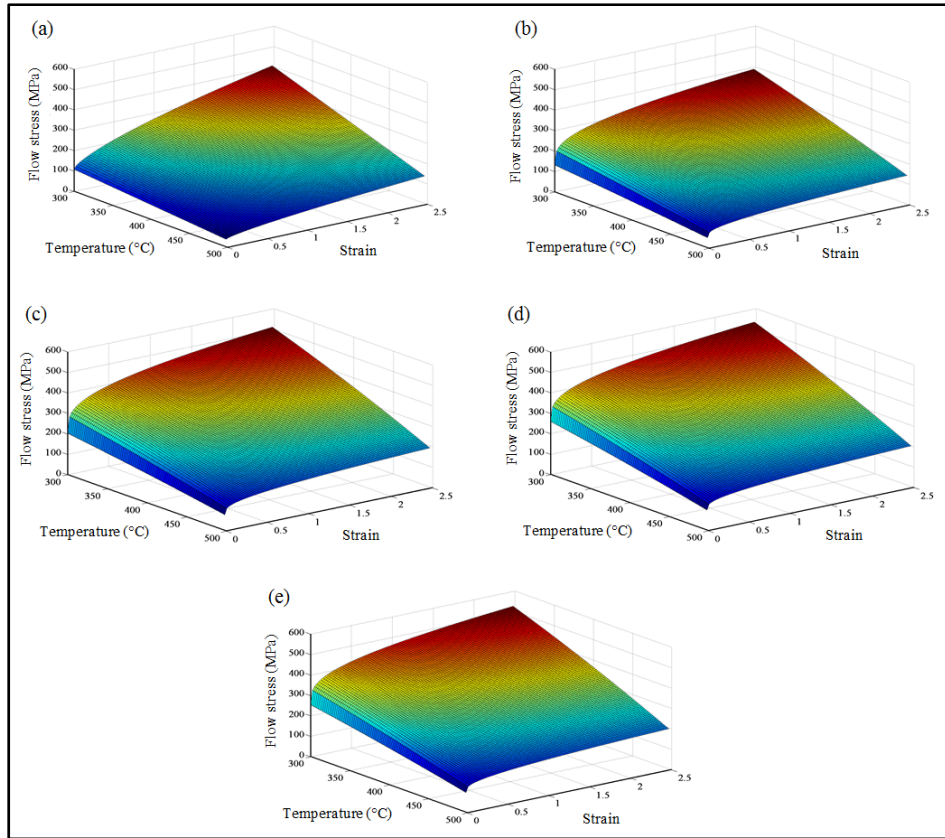


Figure 5-5 Influence of the temperature and strain on the material flow stress ( $\dot{\epsilon} = 10^5 \text{ s}^{-1}$ )  
 (a) JC(-8°), (b) JC(-5°), (c) JC(0°), (d) JC(+5°), and (e) JC(+8°)

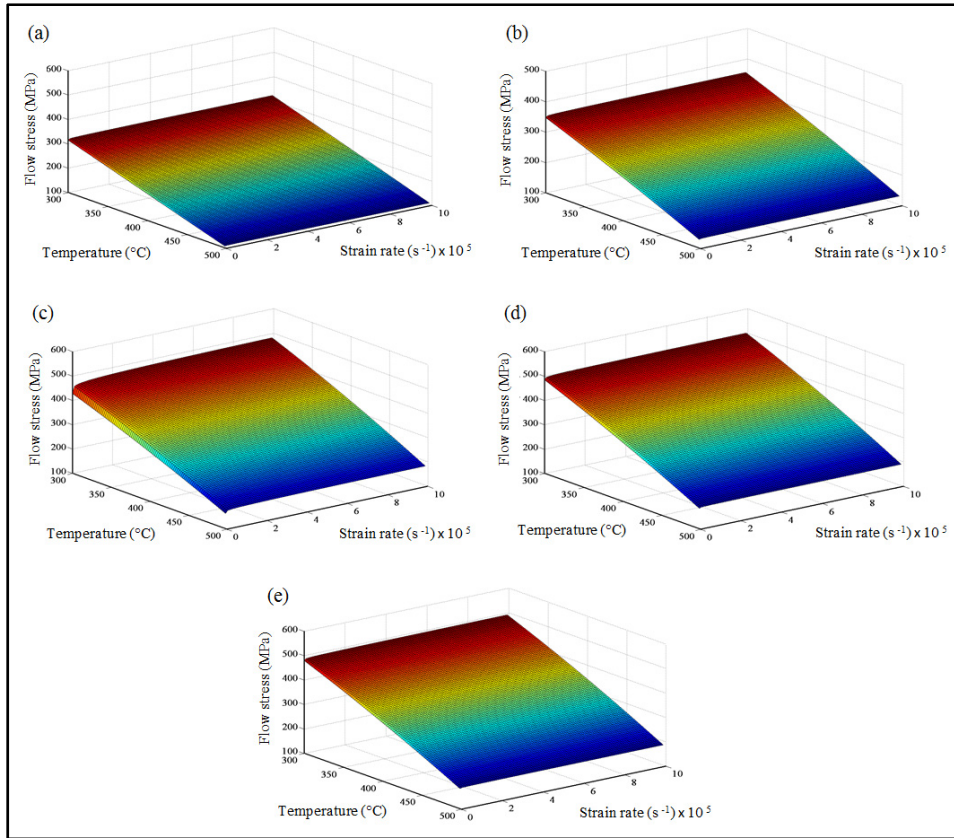


Figure 5-6 Influence of the temperature and strain rate on the material flow stress ( $\varepsilon=1.5$ )  
 (a) JC(-8°), (b) JC(-5°), (c) JC(0°), (d) JC(+5°), and (e) JC(+8°)

## 5.6 Results and discussion

To investigate the effect of five different sets of JC constants obtained from machining tests for five rake angles, the predicted and experimentally measured cutting forces, chip thickness, and tool-chip contact length are compared and discussed hereafter. It is worth pointing that the cutting conditions used in the FE simulations are identical to those used in the experimental tests as mentioned above.

### 5.6.1 Cutting forces

Cutting forces results were obtained experimentally with two feed rates (0.16 and 0.25 mm/rev), two cutting speeds (600 and 950 m/min), and five rake angles (-8°, -5°, 0°, +5°, and

+8°) as listed in Table 5-3. Based on these cutting conditions, the measured cutting forces (tangential ( $F_c$ ) and thrust ( $F_f$ )) are presented in Figure 5-7. As shown in this figure, the cutting forces are heavily influenced by the rake angle. An analysis of the results shows that the cutting forces decrease with alteration of rake angle from -8° to +8°. This is due to the drop of the contact pressure and the friction force at the tool-chip interface when the rake angle changes from a negative to a positive value. However, the diminution of the rake angle is favorable to a change in chip formation mechanism from continuous to serrated, thus decreasing the tangential force. This explains the decrease of the tangential force when the rake angle decreases from +5° to -5°.

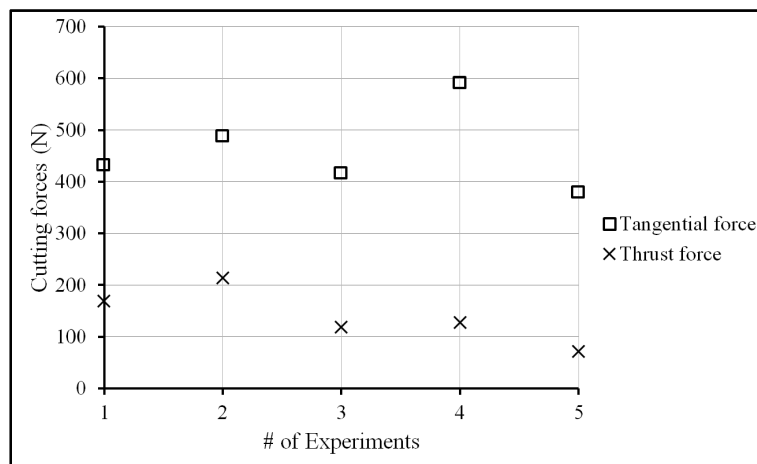


Figure 5-7 Variation of cutting forces with the cutting conditions during the experiments

For the five sets of JC constants, the predicted tangential ( $F_c$ ) and thrust ( $F_f$ ) forces are compared to the measured ones. Figure 5-8 and Figure 5-9 give a clear comparison of the results obtained during the five different tests listed in Table 5-3. All JC constants underestimate the cutting forces. It is interesting to note that JC(0°), JC(+5°), and JC(+8°) predict better the tangential force, as shown in Figure 5-8. The predicted tangential forces are within 7.9, 9, and 10.2% of the experimental values with JC(0°), JC(+5°), and JC(+8°), respectively. By taking the average relative error for each set of JC, the JC(0°), JC(+5°), and JC(+8°) have practically similar values of 4, 4.36, and 4.93%, respectively. This can be explained by the fact that the flow stresses predicted by these sets of JC are close to each

other as shown in Figure 5-5 and Figure 5-6. Oppositely, higher relative errors are encountered when evaluating the tangential forces with material constant set JC(-8°) and JC(-5°) because of the lower flow stress predictions. However, the predictions of thrust force are less accurate for all five JC sets, as shown in Figure 5-9. The underestimation of this predicted force has already been the subject of discussion by many researchers (Filice *et al.*, 2007a; Klocke *et al.*, 2013), and Deform software itself has already mentioned this problem (SFTC, 2012). This phenomenon could be attributed to the reduced number of element in the secondary shear zone (numerical issues). However, the average error related to the thrust force was found to be 27.8 and 30.6% for JC(-8°) and JC(0°), respectively.

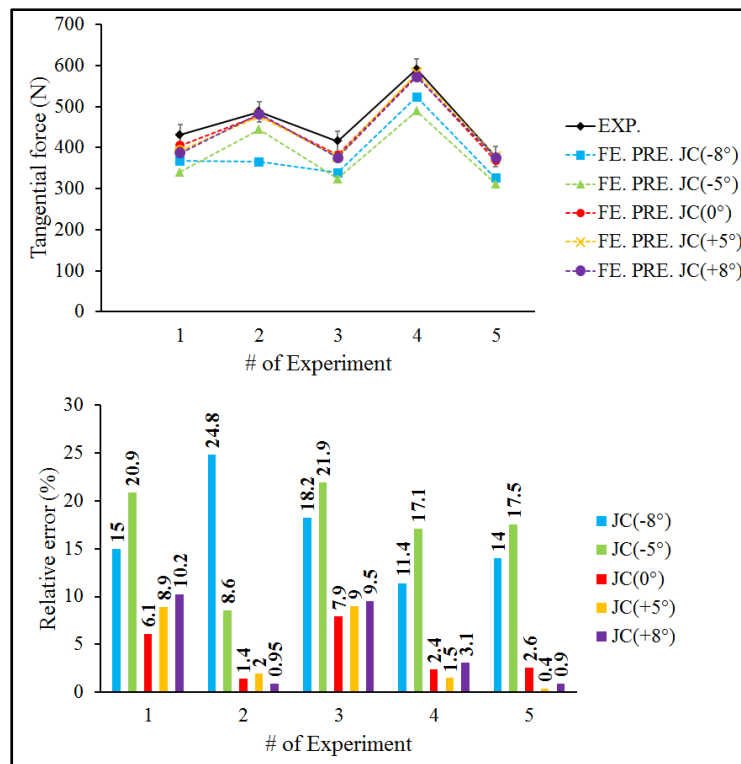


Figure 5-8 Comparison between experimental (EXP.) and predicted (FE. PRE.) tangential forces

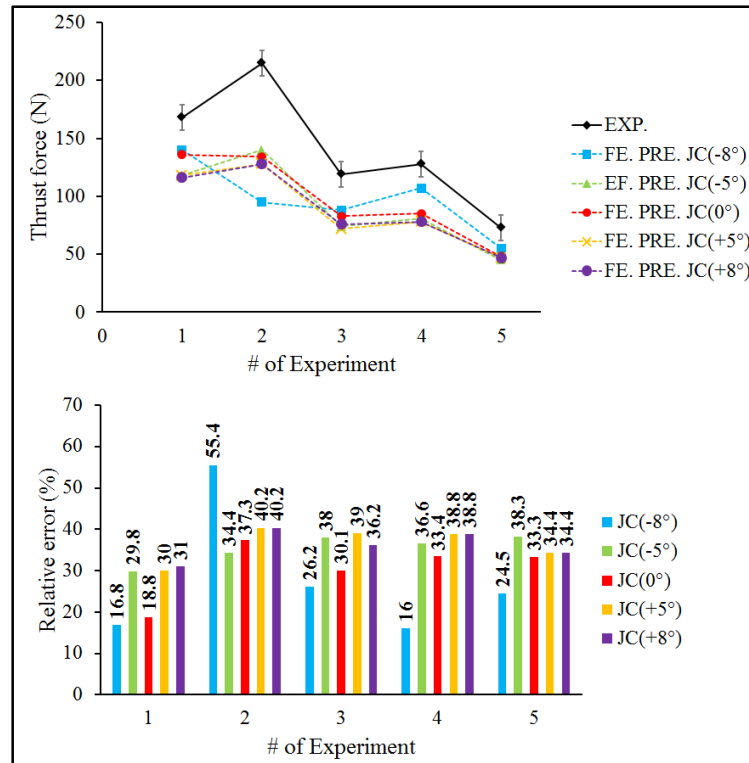


Figure 5-9 Comparison between experimental (EXP.) and predicted (FE. PRE.) thrust forces

## 5.6.2 Chip thickness

Figure 5-10 compares the experimental chip geometry for the five tests. As shown, under the cutting conditions selected in this work, a slight chip undulation is observed in tests (no. 1, no. 3, no. 4, and no. 5) and a serrated chip is found in test no. 2. It is clear that the chip formation mechanism changes from continuous to serrate as the rake angle changes from  $+5^\circ$  to  $-5^\circ$ . As mentioned above, these two mechanisms lead to different thermo-mechanical loads in the cutting zone. In fact, under certain cutting conditions, the plastic strain rates become high enough to generate considerable heat in the primary shear zone which cannot rapidly be dissipated to the rest of workpiece material. This results in a quasi-adiabatic condition which causes material thermal softening (Xie *et al.*, 1996). As the cutting process continues, the cooler material enters the primary shear zone to be cut leading to the serrated chip geometry. Figure 5-11 compares the predicted chip thickness obtained with the five material constant

sets. It is clear that the JC(0°) gives a best prediction with smaller average relative error of 4.26%. However, large discrepancies in chip thickness were obtained using material constant set JC(-8°) with a maximum relative error of about 66%. The worst prediction of chip thickness with JC(-8°) could be explained by the fact that lower yield strength coefficient  $A$  and higher value of hardening coefficient  $n$  predict lower flow stresses and higher plastic deformation. Consequently, early plastic deformation occurs ahead of the tool tip in the material and hence thicker chip is produced (Nasr *et al.*, 2007b; Sartkulvanich *et al.*, 2005a). An analysis of the predicted results, Figure 5-11, shows that the chip thickness is more sensitive to the JC material constants than the cutting forces. The predicted and experimentally measured serrated chip geometries of test no. 2 are shown in Figure 5-12. As shown in this figure, the chip thickness and the serration frequency are relatively well predicted when the material constants JC(0°) is used.

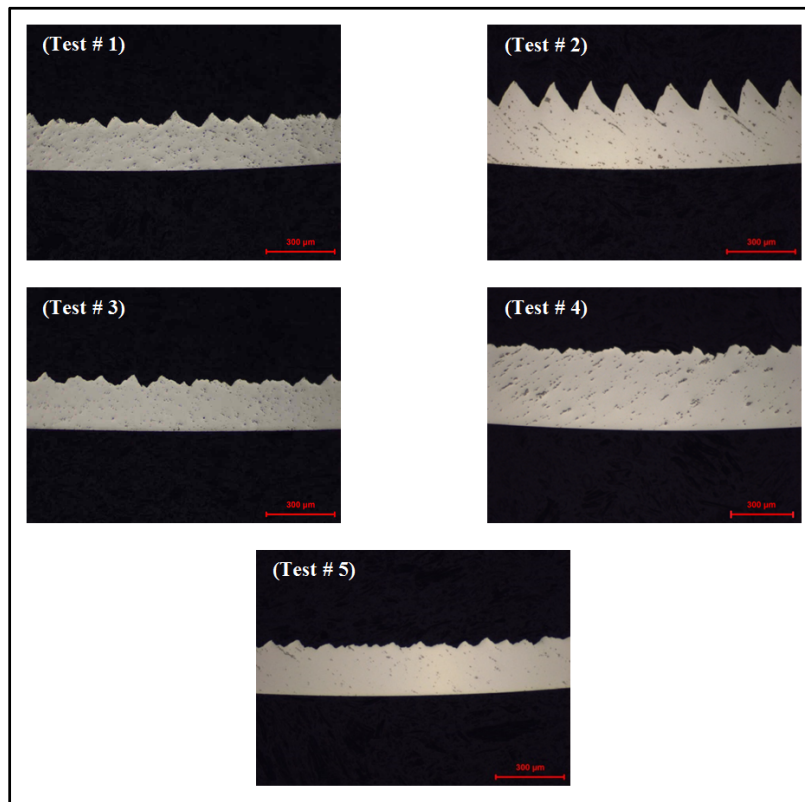


Figure 5-10 Comparison between experimental chip geometry

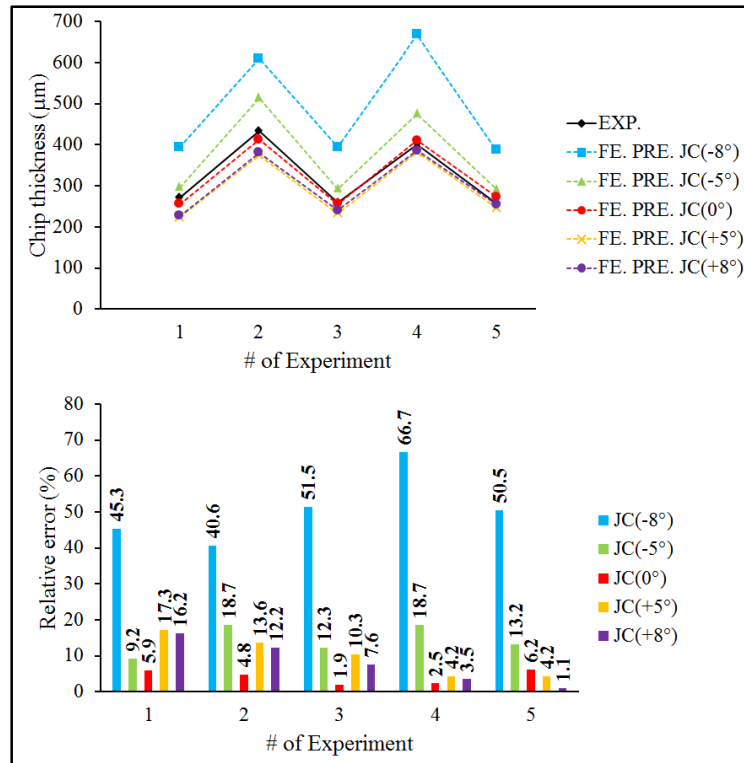


Figure 5-11 Comparison between experimental (EXP.) and predicted (FE. PRE.) chip thickness

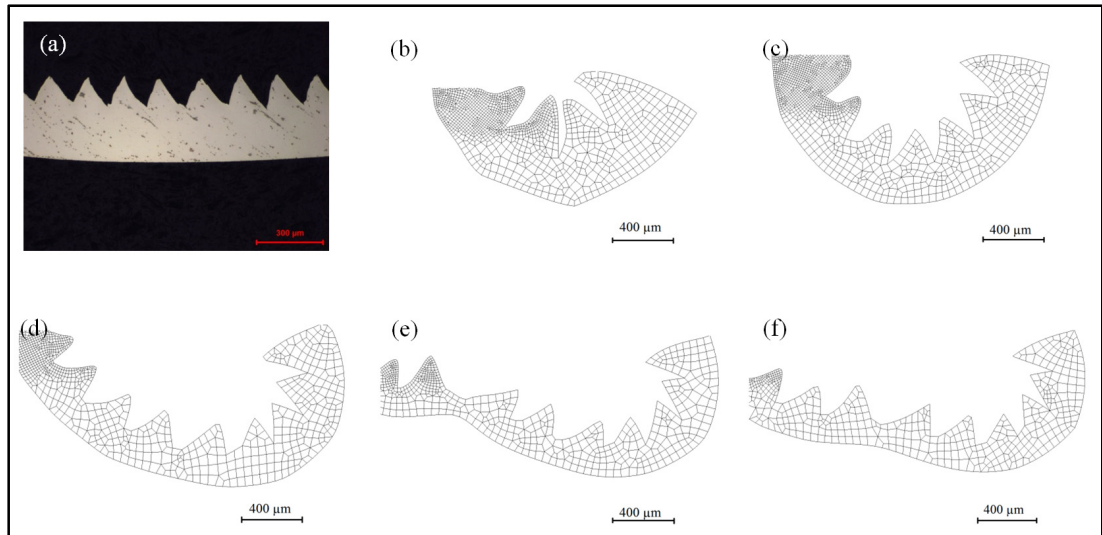


Figure 5-12 Comparison between experimental (EXP.) and predicted (FE. PRE.) chip morphology for test no. 2. (a) EXP. (b) FE. PRE. JC(-8°), (c) FE. PRE. JC(-5°), (d) FE. PRE. JC(0°), (e) FE. PRE. JC(+5°), and (f) FE. PRE. JC(+8°)

### 5.6.3 Tool-chip contact length

The tool-chip contact length is a significant parameter because it defines the amount of heat generation in the secondary shear zone in the thermo-mechanical analysis. This parameter is the main factor that contributes to tool temperature increase in high machining and leads to properties degradation and premature wear. In fact, the longer the tool-chip contact length is, the greater the secondary contact zone is obtained. This results again in more heat generation. The effect of the five sets of material constants on the prediction of the tool-chip contact length was, therefore, investigated. The results are presented in Figure 5-13. In this case, the material constant set JC(-5°) gives the best prediction, followed by the material constants set JC(0°), JC(+5°), and JC(+8°), respectively, with an average difference of 3.64, 9.38, 13.4, and 12.52%, respectively. Even though the material constant set JC(0°) underestimates the contact length by up to 17.2%, it still predicts tool-chip contact length with reasonable accuracy as compared to other sets. Finally, it is not surprising that the prediction of the contact length with material constant set JC(-8°) is less accurate because the tool-chip contact length is directly related to the chip thickness as reported in (Sartkulvanich *et al.*, 2005a). In fact, larger chip thickness is difficult to curl which increases the tool-chip contact length.

Finally, it is worth outlining that previous numerical work (Shi *et al.*, 2010b) has shown also that 0° rake angle gives the largest extent of the central portion of the primary shear zone. Based on our findings, the analytical model used for the estimation of the physical quantities ( $\bar{\epsilon}_{AB}$ ,  $\dot{\bar{\epsilon}}_{AB}$ ,  $\bar{\sigma}_{AB}$ ,  $T_{AB}$ ) during the identification step gives better predictions of these quantities with a 0° rake angle.



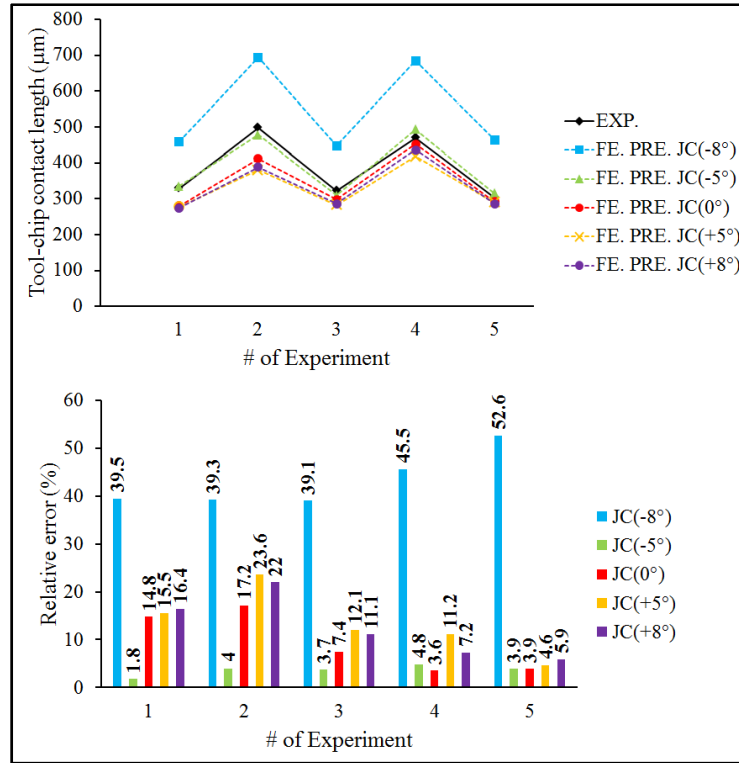


Figure 5-13 Comparison between experimental (EXP.) and predicted (FE. PRE.) Tool-chip contact length

## 5.7 Conclusions

In this work, the effect of five different sets of JC constants on the numerical FEM predicted cutting parameters was investigated. Orthogonal machining tests under dry cutting conditions were performed on Al2024-T3 using a disk-shaped workpiece and an uncoated carbide cutting tool. An inverse approach based on response surface methodology (IABRSM) was used to determine five sets of JC constitutive law at different five rake angles. These sets of JC were then implemented in the FEA software DEFORM-2D. Finally, the predicted results were compared with the experimental ones. An analysis of the predicted cutting parameters shows that the material constants set obtained with 0° rake angle gives an overall more accurate prediction of cutting forces, chip thickness, and tool-chip contact length. It is argued that the cutting with the rake angle of 0° is the closest cutting condition to the assumption of the analytical model used for the estimation of the physical quantities ( $\bar{\epsilon}_{AB}$ ,  $\dot{\bar{\epsilon}}_{AB}$ ,  $\bar{\sigma}_{AB}$ ,  $T_{AB}$ ) in

the primary shear zone. This can explain the more accurate results obtained with the  $0^\circ$  rake angle.

## CHAPTER 6

### **PREDICTION OF RESIDUAL STRESSES AND TEMPERATURES GENERATED DURING AL2024-T3 CUTTING PROCESS SIMULATION WITH DIFFERENT RAKE ANGLE-BASED JOHNSON-COOK MATERIAL CONSTANTS**

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#### **6.1 Abstract**

Nowadays the use of finite element modeling (FEM) to simulate machining processes has become a vital and irreplaceable tool; thanks to advanced finite element codes which offer the possibility to analyze local physical quantities such as strain, stress, and temperature, which in some cases are difficult to measure experimentally. The accuracy and reliability of the results obtained from FEM simulation of machining processes depend strongly on the constitutive law which describes the thermo-visco-mechanical behavior of the machined material. Johnson and Cook's (JC) constitutive model is widely used in the modeling of machining processes. However, one can find in the literature, different material constants of the JC constitutive law for the same material which can significantly affect the predicted results (cutting forces, temperatures, residual stresses, etc.). Therefore, understanding how these material constants affect the FEM predictions can make the simulation of machining processes more reliable. In the present work, three different sets of JC constants, determined through orthogonal machining tests, are used in finite element modeling to simulate the machining behavior of Al2024-T3. The effects of these sets on the numerically predicted residual stresses within the machined workpiece and the temperatures of the cutting tool are the subject of a comparative investigation. It is found that the residual stress distributions are

much more sensitive to these sets of JC than the temperature distributions. Additionally, it is concluded that the set of JC obtained at 0° rake angle, JC(0°), gives an overall more accurate prediction of residual stresses and temperatures.

**Keywords:** Machining; Johnson-Cook constitutive law; FEM; Residual stresses; Cutting temperature; Al2024-T3

## 6.2 Introduction

Today, the high-speed machining (HSM) is a major material removal process of metal structural parts in aerospace industry. Numerical modeling techniques such as finite element modeling (FEM) are widely employed in HSM to better understand the chip formation mechanism. In addition, it gives access to many difficult-to-measure machining process variables and it has proven to be highly efficient (Vaz Jr *et al.*, 2007). Moreover, FEM appears to be the most suitable method to design and develop the machining processes and cutting tool as compared to the experimental procedures which are costly and time consuming. Consequently, the experimental trial and error approach could be avoided. In fact, extreme conditions such as high levels of strain, strain rate, and heat are usually encountered in metal cutting processes. One of the most important governing factors during the cutting simulation is the use of an accurate material model which represents the material behaviour especially at the extreme conditions that exist in the shear zone (Childs, 1997; Sartkulvanich *et al.*, 2005a). Various material models that are used for machining simulation have been proposed to reproduce the thermo-mechanical effects involved in metal cutting. The Johnson-Cook constitutive model has been widely used in metal cutting simulation (Arrazola *et al.*, 2008; Mabrouki *et al.*, 2008; Miguélez *et al.*, 2009; Nasr *et al.*, 2007a; Nasr *et al.*, 2007b; 2007c; Outeiro *et al.*, 2008). Furthermore, it was proved to be appropriate in modeling the cutting process (Adibi-Sedeh *et al.*, 2003; Huang et Liang, 2003; Karpas et Özel, 2006; Lalwani *et al.*, 2009; Lee, 2011; Long et Huang, 2005; Özel et Zeren, 2004). However, the task of determining accurate and reliable JC constants for numerical simulation of machining processes is often regarded as a key issue. This is because the predicted results

of the machining process such as cutting forces, chip morphology, temperatures, tool wear, and residual stresses are strongly influenced by these material constants. In the published literature, many researchers evaluated JC constants for aluminum alloy which is widely used in aircraft applications. Various experimental tests have been proposed to determine these constants at high strain, high strain rate, and high temperature. These experimental tests include dynamic tests such as torsion test, split-Hopkinson bar technique (SHBT), and Taylor impact test and inverse method such as machining test. The first set of JC constants for Al2024-T351 alloy was determined using torsion tests with a variation of strain rate between  $0.088$  to  $123 \text{ s}^{-1}$  (Johnson et Cook, 1983). Lesuer (2001b) used data from SHBT carried out on aluminum alloys (Al2024-T3) to determine a new set of JC material constants at high strain rates ranging from  $10^3$  to  $10^4 \text{ s}^{-1}$ . To analyze the machining of aluminum alloys (Al2024-T3, Al6061-T6, and Al6081-T6), Adibi-Sedeh *et al.* (2003) used data from SHBT at high strains to determine the first three parameters of JC model while the other parameters were obtained from the literature. Dannemann (2001) used SHBT at high strain rates in combination with other quasi-static tests, to obtain the constants for two aluminum alloys (Al6061-T6 and Al7075-T6). Taylor impact test was also used by (Rule, 1997) to extract the constants for Al6061-T6 alloy at high strain rates up to  $10^5 \text{ s}^{-1}$ . In addition to being complex, taking considerable effort, and causing technical difficulties (Panov, 2006), dynamic tests produce lower levels of strains and strain rates than those induced by cutting processes (Li *et al.*, 2011). As a result, the generated data do not represent the real thermo-mechanical loading encountered in machining.

Another approach known as the inverse method based on machining tests has been proposed by various research groups to provide the JC constants (Daoud *et al.*, 2015b; Guo, 2003; Limido, 2008; Ozel *et al.*, 2006). In such an approach, the experimental data (i.e. cutting forces, thrust forces and chip geometry) were converted to physical quantities (i.e. average stresses, strains, strain rates and temperatures) in the primary shear zone by using analytical and empirical models. This approach has the advantage of providing material constants at extreme conditions such as strain rates up to  $10^6 \text{ s}^{-1}$ , temperature up to  $1000^\circ\text{C}$  and strains up to 4. Guo (2003) used the inverse method in combination with conventional compression

tests to identify the JC constants of Al6061-T6 alloy. In this work, the predicted flow stresses are found to be in good agreement with the experimental ones. Ozel *et al.* (2006) developed a methodology based on machining tests combined with SHBT to determine the material constants of Al6082-T6 at machining regimes. Limido (2008) has also used the inverse method with hot tensile test to determine the material constants of aluminum alloys (Al2024-T3 and Al7070-T7). Daoud *et al.* (2015b) has developed an inverse approach based on response surface methodology (IABRSM) to determine the JC constants for three aluminum alloys (Al2024-T3, Al6061-T6, and Al7075-T6) over a large range of strain, strain rate, and temperature. Unfortunately, the above mentioned methods result in different sets of material constants for the same material and consequently the accuracy of the predicted results will be affected. Adibi-Sedeh et Madhavan (2003) attested that material constants fitted to the flow stress data obtained from machining tests can give more accurate numerical predicted results. Recently, Umbrello *et al.* (2007b) conducted a sensitivity study of JC constants in predicting cutting parameters in the machined steel alloy. It was concluded that the predicted results such as cutting forces, chip morphology, temperature distributions, and residual stresses are well predicted when using a set of JC constants determined from machining experiments. Based on the previous literature review, one can argue that machining tests appear to be the most suitable method to determine the material constants. Although they have been used as characterization test by many research works, no particular attention was paid to the effect of rake angle on the JC constants and consequently, on the predicted results. The literature shows that different rake angles were used during the machining tests: +6° in (Guo, 2003), -6°, -5° and +8° in (Ozel *et al.*, 2006), 20° in (Limido, 2008), and -8°, -5°, 0°, +5°, +8° in (Daoud *et al.*, 2015b). It is worth outlining that the rake angle is regarded as one of the most critical parameter in machining process. This is because the variation in rake angle significantly changes the thermo-mechanical loads in the cutting zone (Sartkulvanich *et al.*, 2005a; Shih, 1995). Thus, it can be concluded that the rake angle appears to have a significant impact on the material models when the inverse method is adopted and requires an investigation. Although extensive studies on FEM of the orthogonal metal cutting are reported in the literature for steel, titanium, and aluminum alloys (Abboud *et al.*, 2013; Davim *et al.*, 2008; Ee *et al.*, 2005; Filice *et al.*, 2007a; Filice *et al.*, 2006; Guo et Liu, 2002b;

Liu et Guo, 2000; Mabrouki *et al.*, 2008; Outeiro *et al.*, 2006; Yen *et al.*, 2004b), all these works were primarily focused on the prediction of machining forces, chip morphology, cutting temperatures, chip-tool contact length, strains and residual stresses. This research work aims to give a special attention to the effect of the Al2024-T3 JC constants, obtained by machining tests (inverse method) at different rake angles, on the machined workpiece residual stresses and the cutting tool temperature distributions. To conduct such a study, two numerical approaches have been used, namely a 2D thermo-mechanical simulation and a 3D pure thermal analysis.

### 6.3 Johnson-Cook constitutive law and identification approach

The Johnson-Cook model developed to represent the material flow stress under extreme conditions similar to those found in metal cutting was adopted in this study (Johnson et Cook, 1983). This constitutive law, available in many finite element codes, has been successfully used with aluminum alloy to predict the flow stress in conditions similar to metal cutting (Jaspers et Dautzenberg, 2002). The Johnson-Cook model is defined as follows:

$$\bar{\sigma} = \underbrace{[A + B (\bar{\epsilon})^n]}_{\text{Elasto-plastic term}} \underbrace{\left[1 + C \ln\left(\frac{\dot{\bar{\epsilon}}}{\dot{\bar{\epsilon}}_0}\right)\right]}_{\text{Viscosity term}} \underbrace{\left[1 - \left(\frac{T - T_{room}}{T_{melt} - T_{room}}\right)^m\right]}_{\text{Thermal softening term}} \quad (6-1)$$

where  $\bar{\sigma}$  is the equivalent flow stress,  $\bar{\epsilon}$  is the equivalent plastic strain,  $\dot{\bar{\epsilon}}$  is the equivalent strain rate,  $\dot{\bar{\epsilon}}_0$  is the reference strain rate,  $T$  is the temperature of the work material,  $T_{melt}$  is the melting point of the work material and  $T_{room}$  is the room temperature. The material constants are as follows:  $A$  is the yield strength coefficient;  $B$  the hardening modulus;  $C$  the strain rate sensitivity coefficient;  $n$  the hardening coefficient and  $m$  the thermal softening coefficient. The material constants are determined using the inverse approach based on response surface methodology (IABRSM) developed by (Daoud *et al.*, 2015b). In this approach, the orthogonal machining tests are conducted using central composite design. Then, the response surface methodology (RSM) was used to analyse the effect of the rake angle on the material constants of JC under a large number of cutting conditions. The three

different sets of workpiece material constants obtained at three rake angles using (IABRSM) are given in Table 6-1.

Table 6-1 Material constants for Al2024-T3 identified (IDE.)  
at three rake angles (Daoud *et al.*, 2015b)

Set of JC	$A$ (MPa)	$B$ (MPa)	$n$	$C$	$m$	$\dot{\bar{\epsilon}}_0$ (1/s)
JC(-8°) (IDE. at -8°)	200	300	0.675	0.003	1	1
JC(0°) (IDE. at 0°)	257.6	300	0.3	0.0149	1.49	1
JC(+8°) (IDE. at +8°)	366.4	300	0.3	0.003	1.49	1

The efficiency of this approach is verified by comparing the predicted flow stresses using Equation (6-1) with the cutting test data given in Table 6-2. For such cutting conditions, the measured and predicted flow stresses are plotted in Figure 6-1 using the material constants listed in Table 6-1 and four different sets of JC reported in the literature (Adibi-Sedeh *et al.*, 2003; Dannemann, 2001; Johnson et Cook, 1983; Lesuer, 2001b). Even though the material constant sets obtained by IABRSM slightly overestimate the flow stress, it is clear that JC(0°) and JC(+8°) give a much better approximation than JC(-8°) with a percentage difference of flow stress of 13.3 and 13.7%, respectively.

Table 6-2 Cutting test data for Al2024-T3 ( $\alpha=+5^\circ$  &  $W=3.14$  mm)

Test no.	$V_c$ (m/min)	$f$ (mm/rev)	$F_c$ (N)	$F_f$ (N)	$t_c$ (mm)	$\bar{\epsilon}_{AB}$	$\dot{\bar{\epsilon}}_{AB}$ ( $\times 10^5$ 1/s)	$T_{AB}$ (°C)	$\bar{\sigma}_{AB}$ (MPa)
1	363	0.07	199	117	0.166	0.76	2.10	194.0	429.0
2	375	0.16	430	152	0.280	0.62	1.00	238.6	515.9
3	387	0.25	595	166	0.413	0.60	0.67	236.2	490.0
4	399	0.31	668	162	0.513	0.60	0.56	225.3	456.4
5	939	0.07	210	69	0.177	0.80	5.40	277.2	501.8
6	951	0.16	423	92	0.253	0.59	2.61	277.4	576.6
7	963	0.25	570	94	0.349	0.56	1.75	255.6	531.9
8	975	0.31	631	82	0.423	0.56	1.44	239.8	490.8
9	1503	0.31	541	49	0.329	0.53	2.46	216.1	438.1
10	1515	0.07	155	65	0.175	0.79	8.73	221.2	359.4
11	1527	0.16	351	77	0.213	0.55	4.41	234.7	489.0
12	1539	0.25	481	67	0.271	0.53	3.09	222.6	458.5



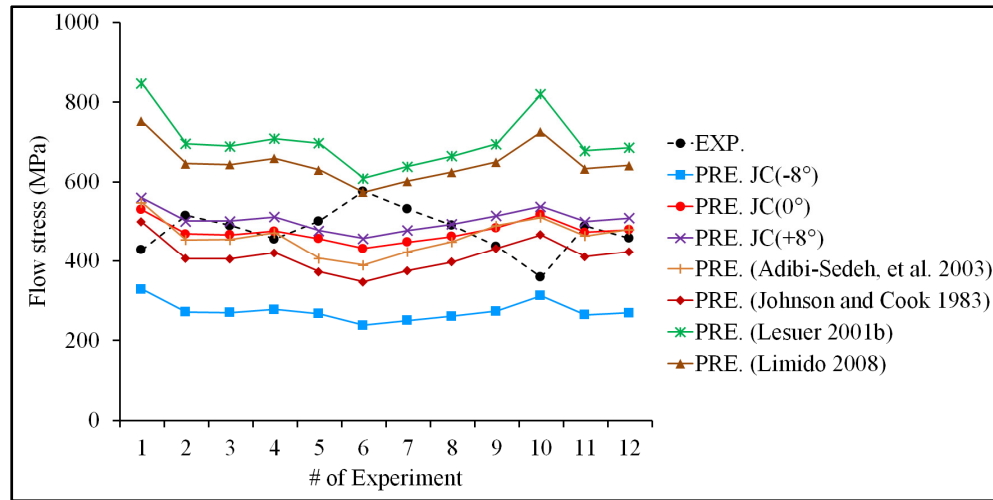


Figure 6-1 Comparison between experimental (EXP.) and predicted (PRE.) flow stresses (cutting conditions listed in Table 6-2)

## 6.4 Experiments

### 6.4.1 Workpiece material

The machining tests were conducted on aluminum alloy Al2024-T3 which has been commonly used in aircraft components because of its combination of high mechanical properties and light weight.

### 6.4.2 Machining set-up

A series of dry machining tests were carried out on Al2024-T3 alloy using Mazak Nexus 410A, 3-axes, CNC machine with a spindle speed of 12,000 rpm and a power of 25 HP, as shown in Figure 6-2. In order to obtain orthogonal cutting conditions, the workpieces are disks in shape having an outer diameter of 75 mm and an inner diameter of 16 mm with a thickness of 3.14 mm. Uncoated and sharp carbide cutting inserts referenced as TPGN 160308 (K68 grade Kennametal Inc.), which are of a triangle shape with  $11^\circ$  clearance angle, were used in all cutting experiments. These cutting inserts are fixed on a left-hand holder (reference CTFPL2525M16, Kennametal Inc.) with a back rake face of  $+5^\circ$ .

As far as cutting temperature measurement is concerned, a chromel/alumel thermocouple (type K) with a diameter of 0.075 mm was utilized. The uncertainty on the temperature measurement arising from this type of thermocouple is  $\pm 1.1^{\circ}\text{C}$  or 0.4% (whichever is greater). The time constant for this thermocouple was measured by using hot water and it is found to be 0.03 sec, as shown in Figure 6-3, taking into account the high temperature chemical set cement utilized to fix the thermocouple inside the cutting tool. Therefore, the system output could fall within 99% of the final value according to the current cutting conditions. Besides, a fine blind hole with a diameter of 0.9 mm was made in the cutting insert by means of an Electrical Discharge Machine (EDM). The diameter of the blind hole and its positions were measured by a laser confocal microscope (see Figure 6-4) while the depth of the hole was measured by Mitutoyo digital height gauge. The geometry of the insert and the position of the hole made inside it are shown in Figure 6-5 and Table 6-3. The thermocouple is then inserted inside the tool and the other end is connected to data acquisition device (thermocouple module model NI 9213).

The cutting forces were also measured by Kistler Quartz three-component dynamometer (model 9255B). It has a measurement uncertainty of  $\pm 1$  and  $\pm 2\%$ , arising from linearity and crosstalk, respectively. LabVIEW software was used to record temperature and cutting forces (tangential ( $F_c$ ) and thrust ( $F_f$ )) at sampling frequency of 100 and 24,000 Hz, respectively.

As to the cutting conditions used in this work, the rake angle, the cutting speed, the uncut chip thickness, and the depth of cut are given in Table 6-4. It is worth noting that each of the three cutting conditions given in Table 6-4 were repeated three times under the same environment conditions using the three cutting tools listed in Table 6-3.

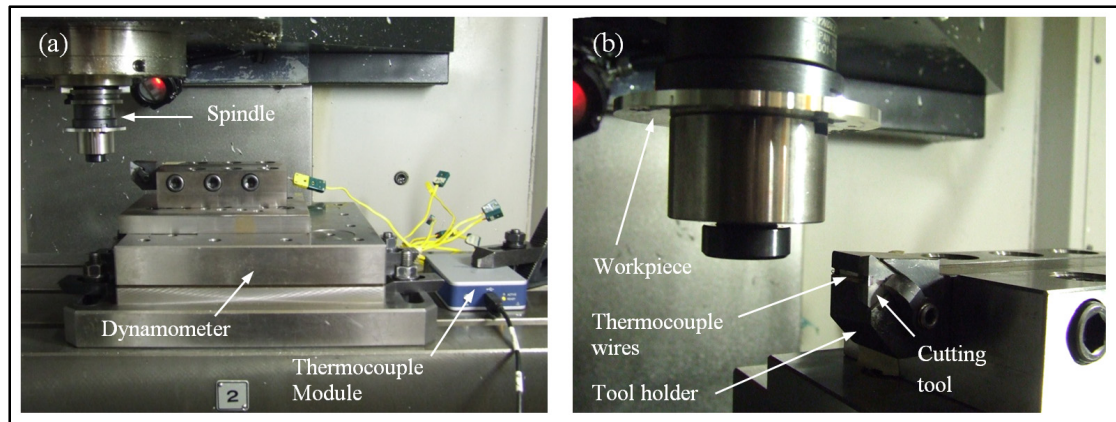


Figure 6-2 Orthogonal machining test  
(a) experimental setup (b) side view of the cutting components

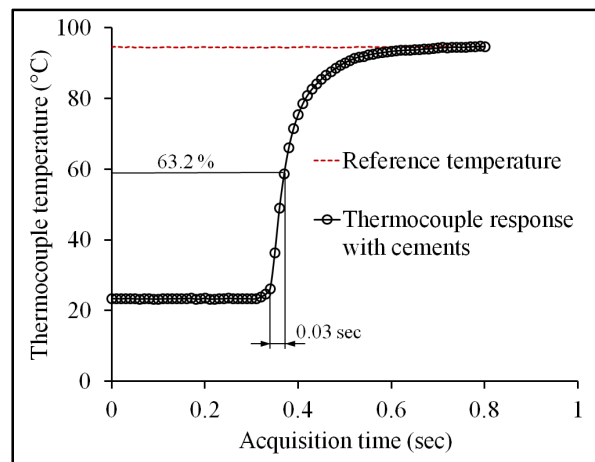


Figure 6-3 Time constant required to reach 63.2 % of the final temperature measurement

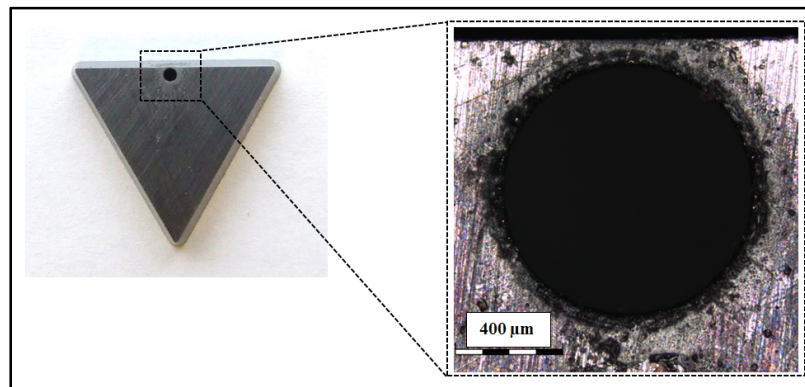


Figure 6-4 Appearance of the blind hole made in the cutting insert by EDM

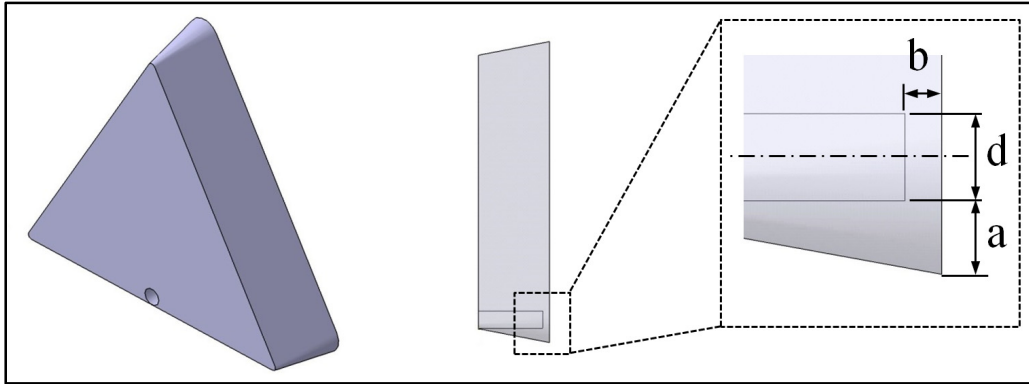


Figure 6-5 Hole position inside the cutting insert for embedded thermocouple

Table 6-3 Geometrical position of the hole for embedded thermocouples

Cutting tool no.	a (mm)	b (mm)	d (mm)
1	0.72	0.35	0.94
2	0.74	0.33	0.92
3	0.73	0.13	0.93

Table 6-4 Cutting conditions

Test no.	$\alpha$ (deg.)	$V_c$ (m/min)	$f$ (mm/rev)	Depth of cut (DOC) (mm)
1	-8	950	0.16	3
2	0	950	0.16	
3	+8	950	0.16	

#### 6.4.3 Residual stress measurement

The residual stress state of the machined surface and sub-surface was measured by means of X-ray diffraction technique using the  $\sin^2\psi$  (SAE et International, 2003). These measurements were carried out on a Proto iXRD system, as shown in Figure 6-6. The main parameters utilized in the X-ray analysis of the Al2024-T3 alloy are listed in Table 6-5. The electro-polishing technique was used to determine the in-depth residual stresses by removing successive layers of surface material without generating additional residual stresses. In addition, the electro-polishing technique, Figure 6-7 (a), was combined with a circular mask

to represent the region of analysis. The thickness of the removed layer was measured using a Mitutoyo dial indicator as shown in Figure 6-7 (b). In the present work, the gradient corrections due to the X-ray penetration were made using a commercial PROTO gradient code. Further corrections to the residual stress measurements due to the removed volume of material were made using the commercial finite element software ANSYS. The analysis of induced residual stresses during the cutting test requires the choice of a machined surface zone which is representative of the cutting test. In fact, at the end of the cutting test, a part of the workpiece is machined with a feed rate which is different from that coded for a given cutting condition. This discrepancy in feed rate is due to the deceleration needed for the disk holder to change its direction of movement during the retraction phase. In order to determine the representative zone, circularity profiles of the machined surface were measured using a coordinate measuring machine (CMM), MT Mitutoyo BRIGHT STRATO 7106, as depicted in Figure 6-8. Only the residual stresses in the cutting direction (circumferential component) were considered in this work.

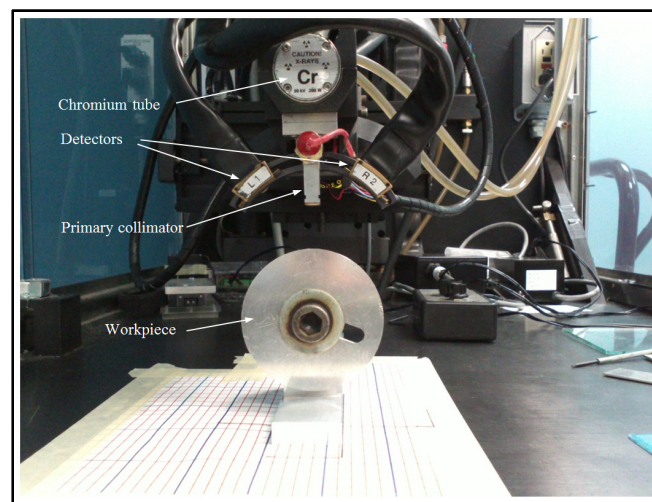


Figure 6-6 Experimental setup of the residual stress measurements

Table 6-5 Parameters utilized in the X-ray measurements

Work material	Al2024-T3
Young's modulus $E$ (GPa), (Prevey, 1986)	53.8
Poisson ratio $\nu$	0.33
tube	Cr- $K_{\alpha}$ (25Kv, 5mA)
Measurement area ( $mm^2$ )	1
Bragg angle $2\theta$ ( $^{\circ}$ ), (Prevey, 1986)	139.3
Crystallographic plane (Prevey, 1986)	(hkl)=(311)
Number of $\beta$ angles	9

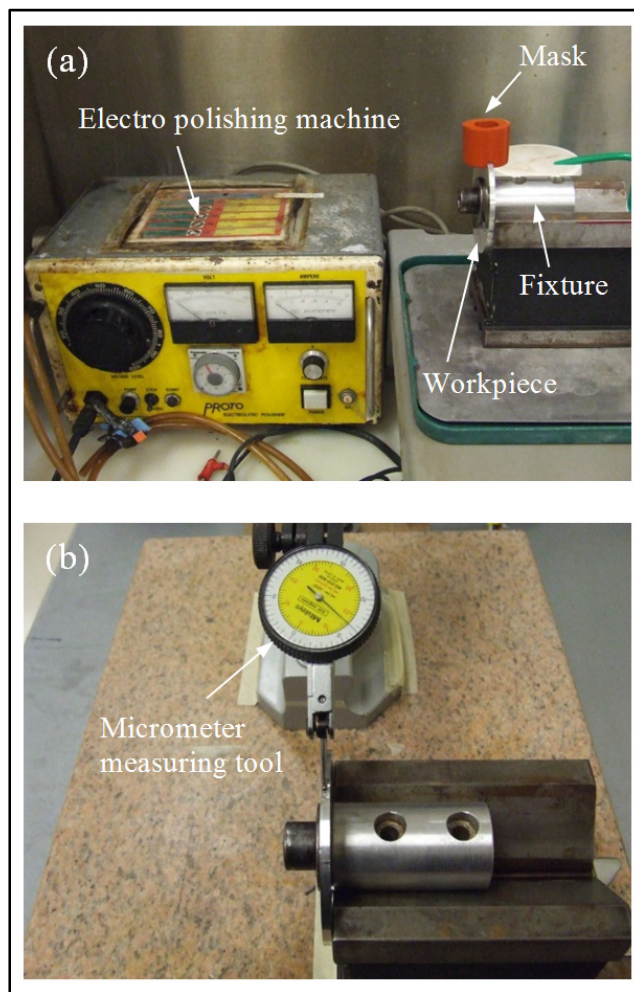


Figure 6-7 Removing successive layers of material (a) Electro-polishing set-up  
(b) Measurements of removed layer thickness

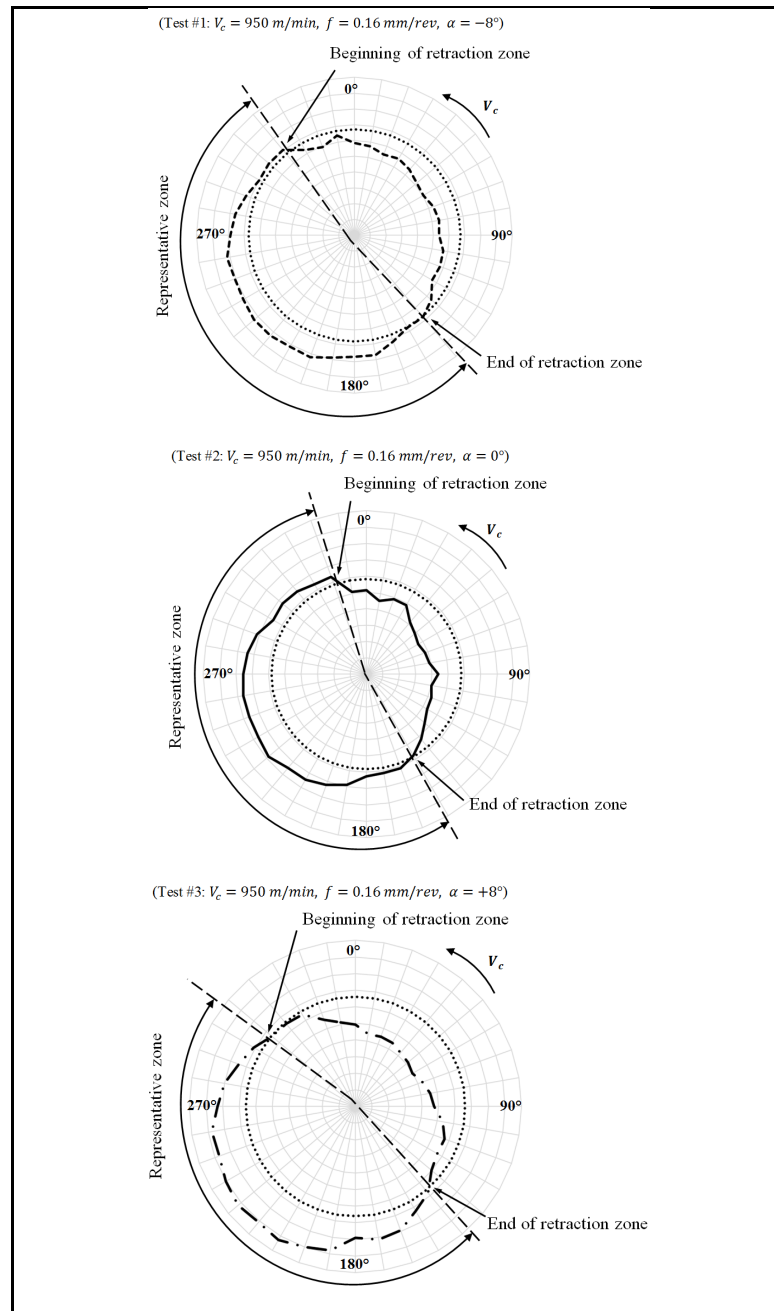


Figure 6-8 Circularity profile of the machined workpiece

## 6.5 Finite element modeling and parameters

It is known that for an accurate prediction of some local parameters, e.g. residual stresses, temperatures, strains, and strain rates, a high mesh density is required. Consequently, a very



short time of high speed machining, only few milliseconds, could be simulated using the coupled thermo-mechanical finite element simulation to avoid CPU time consumption. On the one hand, this small time is sufficient to provide an accurate prediction of cutting forces, chip morphology, tool-chip contact length, and residual stresses (Daoud *et al.*, 2015a; Nasr *et al.*, 2007b; Outeiro *et al.*, 2006). On the other hand, it makes the coupled thermo-mechanical ineffective as far as the heat transfer in the cutting tool is concerned (Umbrello *et al.*, 2007b). Few approaches have been proposed in the literature to overcome such problem (Umbrello *et al.*, 2007a). According to the above considerations, two different approaches are retained for the current study. The first one is a thermo-mechanical simulation using Deform-2D finite element software which is able to predict the residual stresses induced in the workpiece. The second one is a pure thermal analysis using Deform-3D software to obtain the temperature distribution in the cutting tool. The next two sections provide more detailed information on the FEM models.

### **6.5.1 Finite element model for residual stress prediction using Deform-2D**

A plain-strain coupled thermo-mechanical finite element (FE) model was developed to simulate the orthogonal dry machining process of Al2024-T3 using SFTC-Deform-2D finite element software (SFTC, 2012). This commercial software is an updated Lagrangian formulation that has optimized remeshing capability to alleviate element distortions due to large deformation in the shear zones and consequently the material flow around the tool tip could be simulated without the use of a separation criterion (Ee *et al.*, 2005). In this simulation the model is composed of the workpiece and the cutting tool. The workpiece, modeled as an elasto-plastic body (10 mm long  $\times$  1.6 mm high), was initially meshed with 2500 isoparametric quadrilateral elements with 4 integration points and the remeshing algorithm increased the number of element up to 18,000 elements at the end of the simulation. As mentioned before, the JC constitutive law, given by Equation 6-1, was utilized to represent the thermal-visco-plastic behavior of the workpiece material. In addition, the Von Mises yield as a yield function was used in combination with isotropic hardening rule to describe the plastic deformation of the workpiece material. Temperature dependent physical



properties of the workpiece were considered due to the wide variation in temperature in the shear zones. Oppositely, the Young's modulus was kept constant because the workpiece has the tendency to yield and plastically deform instantaneously during the machining process. In view of the high elastic modulus of the cutting tool relative to the workpiece material, the cutting tool was considered as a rigid-stationary body composed of 2100 elements. The material properties of the workpiece and the uncoated cutting tool are presented in Table 6-6. In order to model the heat transfer at the tool-chip interface, a large value of the interface heat transfer coefficient  $h_{int} = 10^4$  (N/ sec mm °C) was utilized to reach the thermal steady state faster in the simulations (Filice *et al.*, 2007b). Due to the adhesion of aluminum alloys to the cutting tool encountered during the machining process (Roy *et al.*, 2009), a constant shear friction model,  $\tau_s = m_f \times K_{chip}$  (SFTC, 2012), was considered with the shear friction coefficient  $m_f = 0.8$ . It is worth pointing out that a slight chip undulation is observed for the cutting conditions selected in this work and listed in Table 6-4. Therefore, the chip morphology was modeled as a continuous chip formation with no damage criterion applied for these conditions. For thermal boundary conditions, the workpiece and tool are initially set at 25 °C (room temperature) and their edges that are sufficiently far from the cutting zone are also maintained at 25 °C. Heat due to radiation and convection to the surroundings is neglected. The mechanical and thermal boundary conditions are presented in Figure 6-9. The zones located around the tool tip, the newly machined surface and about 200µm underneath this surface are modeled with a dense mesh (average element edge length is 7 µm), as shown in Figure 6-9. Therefore, the workpiece has many areas with different element edge length to be followed throughout the simulation. A similar approach is adopted for the cutting tool by applying a dense mesh to the tool tip and part of rake and flank faces. An interface depth of 3.5 µm was adopted as remeshing criterion for the workpiece. During the FE simulation, the cutting tool is considered stationary and the workpiece moves towards it at a cutting speed  $V_c$  and a feed rate  $f$ .

The residual stresses are stresses that remain in the machined workpiece after machining is completed and a return to the initial state of temperature and loading is achieved. Taking into consideration that the residual stresses could not be obtained directly by Deform-2D

software, the relevant steps described in the flowchart of Figure 6-10 are proposed to predict them. These steps are based on two main phases; a cutting phase and a stress relaxation phase. In the first phase, the cutting is simulated with time duration sufficient to reach the steady-state cutting conditions. When cutting forces, chip thickness, tool-chip contact length, and tool-chip interface temperature reach steady-state, the stress relaxation phase starts by disengaging the cutting process and letting the workpiece return to room temperature.

Table 6-6 Physical properties of the workpiece material and the tool substrate (K68)

Property	Material	
	Workpiece (Mabrouki <i>et al.</i> , 2008)	Cutting tool (Yen <i>et al.</i> , 2004a)
	Al2024-T3	WC-CO carbide K68)
Young's modulus $E$ (GPa)	73	612
Poisson ratio $\nu$	0.33	0.22
Density $\rho$ (Kg/m <sup>3</sup> )	2700	11,900
Thermal conductivity $K_{th}$ (W/m °C)	$25 \leq T \leq 300 : K_{th} = 0.247 \times T + 114.4$ $300 < T < T_{melt} : K_{th} = -0.125 \times T + 226$	86
Specific heat $C_p$ (J/Kg °C)	$C_p = 0.557 \times T + 877.6$	337
Thermal expansion coefficient $e$ ( $\mu\text{m/m } ^\circ\text{C}$ )	$e = 8.9 \times 10^{-3} \times T + 22.2$	4.9
Melting temperature $T_{melt}$ (°C)	600	-

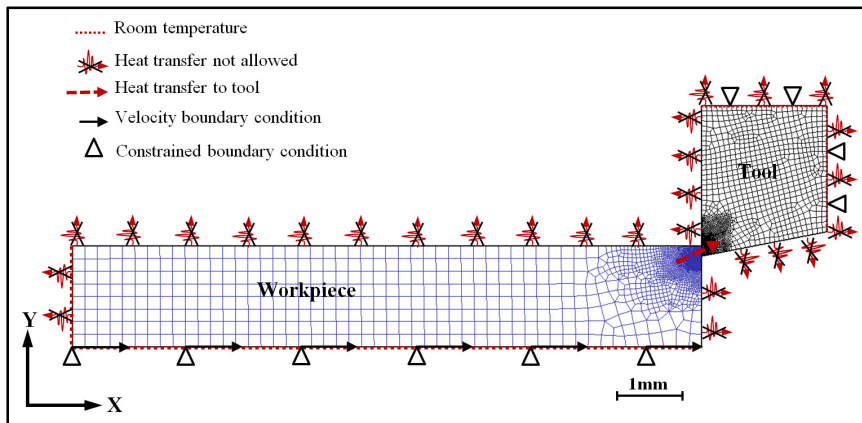


Figure 6-9 Initial boundary conditions of the 2D finite element model

### 6.5.2 Finite element model for temperature prediction using Deform-3D

A pure thermal simulation was conducted to obtain the temperature distribution in the cutting tool using the commercial finite element analysis (FEA) software SFTC-Deform-3D. In this

case, the cutting tool was modeled by means of 500,000 solid elements (4 integration points with bubble node inside the element) with a dense mesh in the interesting zone, as shown in Figure 6-11. Heat convection to the surroundings is considered through the surfaces that are exposed to the environment. In the proposed approach, the temperature distribution in the cutting tool was obtained by applying the nodal temperatures obtained with the 2D thermo-mechanical numerical simulation at tool-chip contact length, as boundary conditions to the 3D thermal model, thermal phase, as detailed in the flowchart of Figure 6-10.

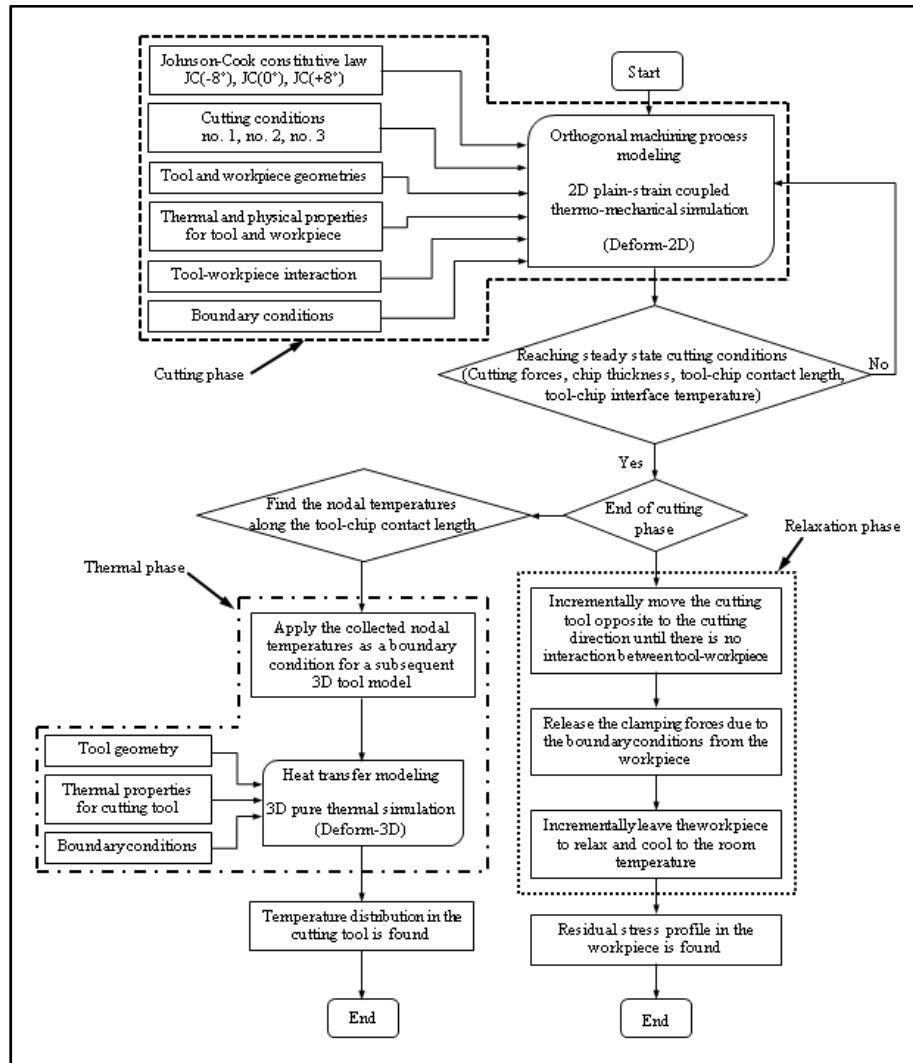


Figure 6-10 Flow chart of FEM for residual stress and temperature predictions

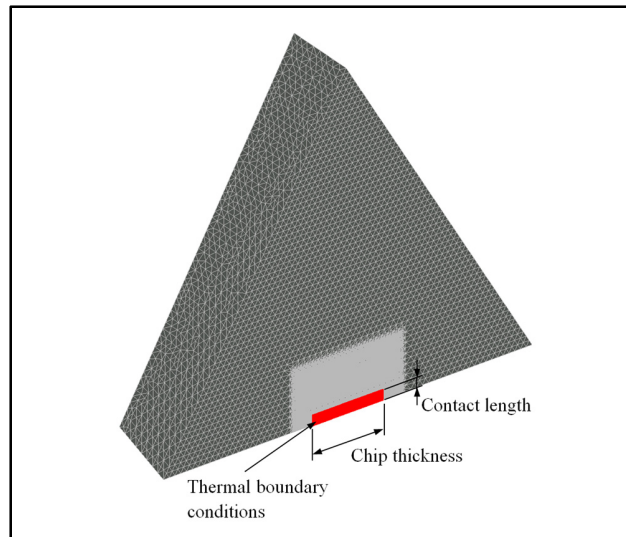


Figure 6-11 3D finite element model and the thermal boundary conditions

## 6.6 Results and discussion

To investigate the effect of three different sets of JC constants determined by (IABRSM) at three rake angles, the numerical predicted residual stresses on the machined surface and subsurface as well as the temperature in the cutting tool are compared to the experimentally measured ones and discussed hereafter.

### 6.6.1 Residual stresses

The functional behavior of a structural component is heavily influenced by the residual stress distribution caused by the machining process (Brinksmeier *et al.*, 1982). It is argued that the residual stresses in the axial direction are the result of those produced in the cutting direction because of the plane strain assumption assumed in orthogonal cutting (Nasr *et al.*, 2007c). A clear explanation of the existence of these stresses is not available in the literature. In fact, a lot of attention is focused on the residual stresses in cutting direction and their relationship to the cutting parameters. Consequently, the present work will consider only the residual stresses in the cutting direction. Figure 6-12 shows the residual stresses obtained experimentally with feed rate of 0.16 mm/rev, cutting speed of 950 m/min, and three rake

angles  $-8^\circ$ ,  $0^\circ$ , and  $+8^\circ$  as listed in Table 6-4. As shown in this Figure 6-12, the residual stresses are tensile in nature at the machined surface with the three curves having the same trend and showing a fluctuation with depth. These fluctuations could be attributed to the coarse grain microstructure of the aluminum alloy. An analysis of the results shows that the values at surface are slightly lowered when the rake angle changes from positive to negative values. This decrease in the level of tensile stress could be explained as follows. The decrease in the rake angle results in more material plastic deformation and an increase in temperature. Consequently, the tensile stress is increased due to the thermal effect. More importantly, the decrease of the rake angle causes chip sticking at the tool rake face and hence promotes built-up edge formation, the effect of which reduces the rake angle and results in an overall decrease in the tensile stresses.

For the three sets of JC constants and the corresponding experimental conditions (Table 6-4: Test #1, #2, and #3), the predicted residual stresses are compared to the measured ones as depicted in Figure 6-13. It is obvious that the predicted and measured residual stress profiles are well correlated using JC( $0^\circ$ ) while JC( $-8^\circ$ ) and JC( $+8^\circ$ ) tend to overestimate and underestimate the residual stresses, respectively. Though the predicted profiles using JC( $0^\circ$ ) do not match exactly with the measured ones, it appears that JC( $0^\circ$ ) comparatively gives a much better estimation of the surface and the in-depth residual stress profiles. It is important to keep in mind that a good correlation between the predicted and experimental results is difficult to obtain. The differences can be attributed to different sources. The measurement of the residual stresses, the thickness of the etched layer, and the material homogeneity are to name a few. Additionally, the constitutive flow stress modeling, friction conditions, numerical integration, discretization errors, and repetitive remeshing technique with its related interpolation error are error sources attributed to FEM.

As shown in Figure 6-13, the three sets of JC predict different residual stress profiles. In order to explain these tendencies, the physical quantities around the tool-tip were analyzed during cutting. Figure 6-14 shows the equivalent plastic strain contours around the tool-tip with the three sets of JC. It can be seen that early plastic deformation ahead of the tool-tip

(thicker chip) with higher magnitudes are produced when the material constants  $JC(-8^\circ)$  is used, followed by the material constants set  $JC(0^\circ)$  and  $JC(+8^\circ)$ , respectively. Since  $JC(-8^\circ)$  has lower yield strength coefficient  $A$  and higher hardening coefficient  $n$ , the material begins to deform plastically under lower stresses (increasing  $n$  produces lower flow stresses when the equivalent plastic deformation is smaller than one). Consequently, early plastic deformation occurs ahead of the tool tip in the material and hence thicker chip is produced (lower shear angle). The lower shear angle obtained with this constant set gives rise to higher plastic shear strain as reported in (Oxley et Young, 1989), which results in higher values of equivalent plastic strain. An opposite trend, however, was observed with  $JC(0^\circ)$  and  $JC(+8^\circ)$  due to their higher  $A$  values of 257 MPa and 366 MPa, respectively, and lower  $n$  value of 0.3. A similar trend for the equivalent plastic strain is observed for other cutting conditions (test #1 and #2) but with an increase in the magnitude when the rake angle changes from a positive to a negative value.

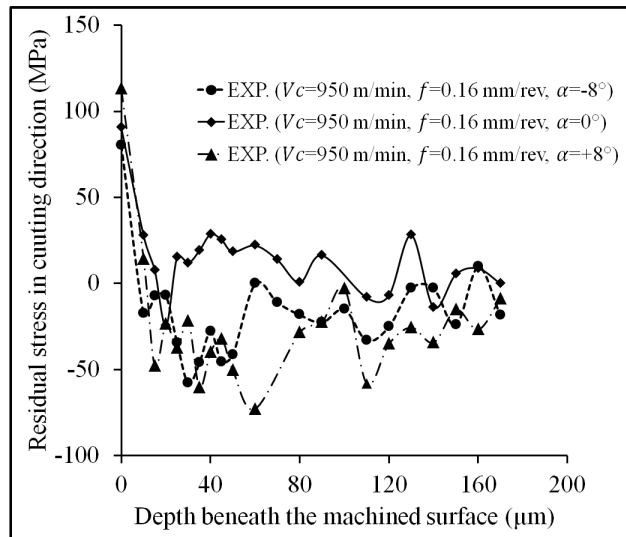


Figure 6-12 Experimental (EXP.)  
residual stresses distribution in cutting  
direction for Al2024-T3

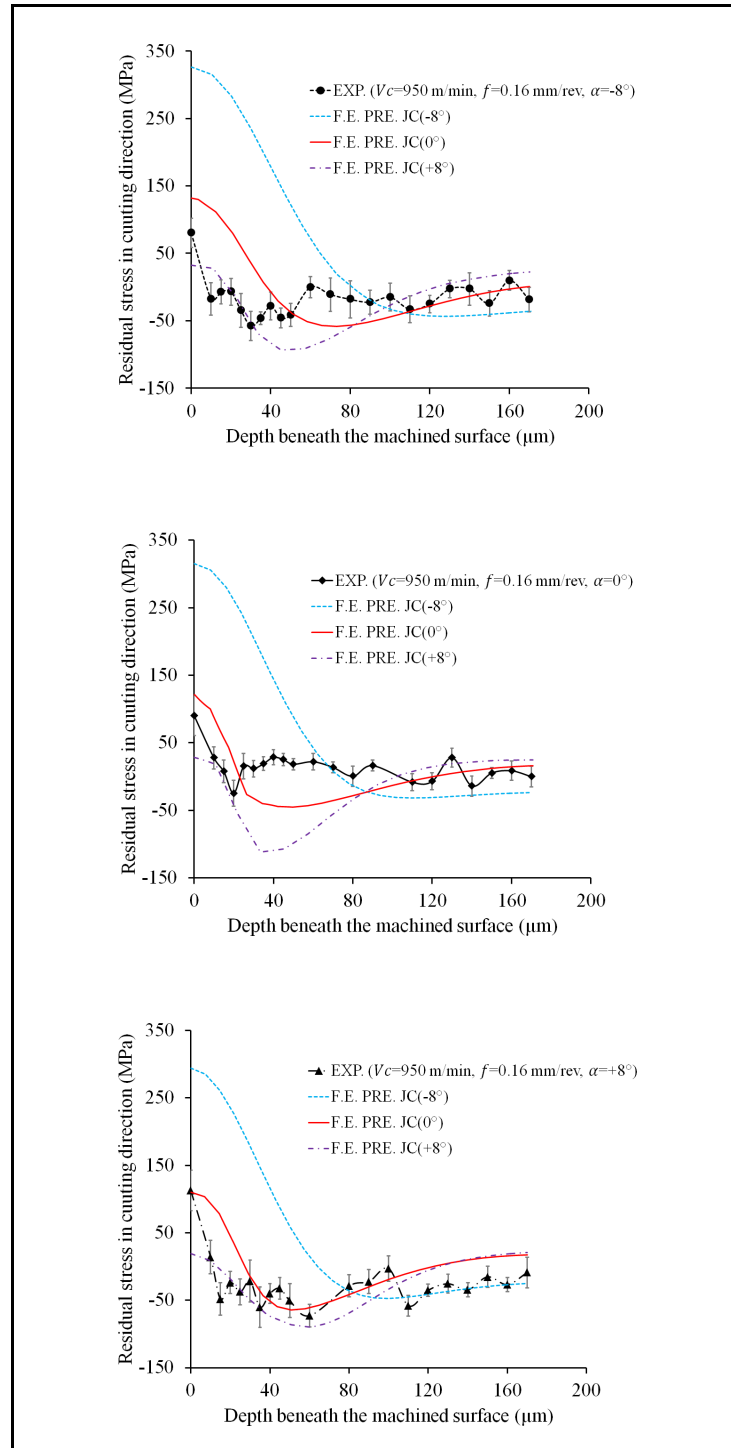


Figure 6-13 Comparison between experimental (EXP.) and predicted (F.E. PRE.) residual stress profiles

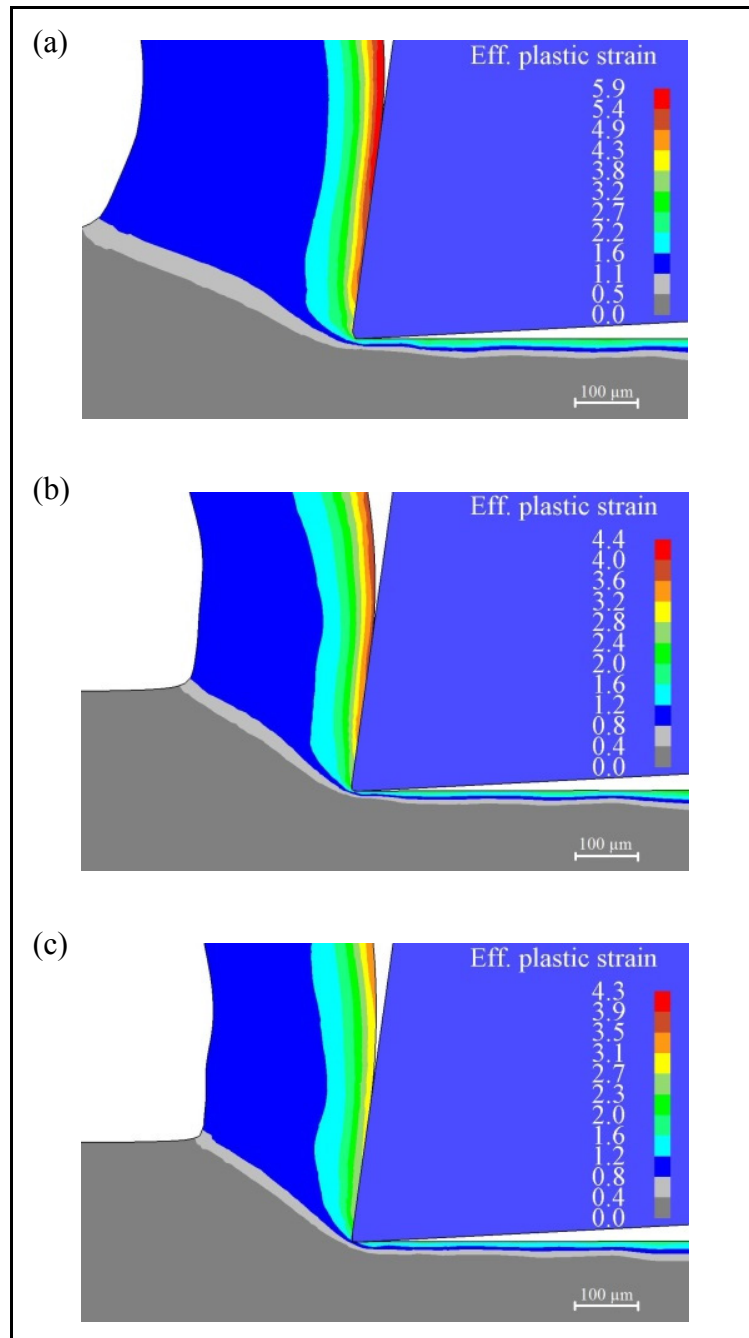


Figure 6-14 Effect of JC sets on equivalent plastic strain during cutting, ( $V_c=950$  m/min,  $f=0.16$  mm/rev,  $\alpha=+8^\circ$ )  
 (a) JC(-8°) (b) JC(0°) (c) JC(+8°)



Figure 6-15 shows the effect of the three sets of JC on the material flow stresses during cutting when the material moves toward the tool-tip. Unlike the material constants JC(0°) and JC(+8°), JC(-8°) produces lower flow stresses and higher values of equivalent plastic strain for reasons mentioned earlier.

The strain rate sensitivity coefficient  $C$  and the thermal softening coefficient  $m$  affect also the material flow stress under extreme cutting conditions. This effect is due to higher strain rates and temperature encountered when using a higher cutting speed and lower feed rate. It was shown that materials with higher  $C$  and  $m$  values experience higher flow stresses rather than higher equivalent plastic strain (Nasr *et al.*, 2007a; Sartkulvanich *et al.*, 2005a). However, the variation observed in terms of the equivalent plastic strain is attributed to the influence of the other material constants.

With reference to the stress-strain curve, it was found that higher plastic strain energy was obtained with JC(-8°) due to the increase in the equivalent plastic strain as shown in Figure 6-15. Assuming that all the plastic strain energy produced by the machining process is converted into heat, 90% in the present study, higher temperatures were observed with JC(-8°) as compared to the two other sets, as shown in Figure 6-16. As a result, the tensile residual stresses of the machined surface are higher with JC(-8°).

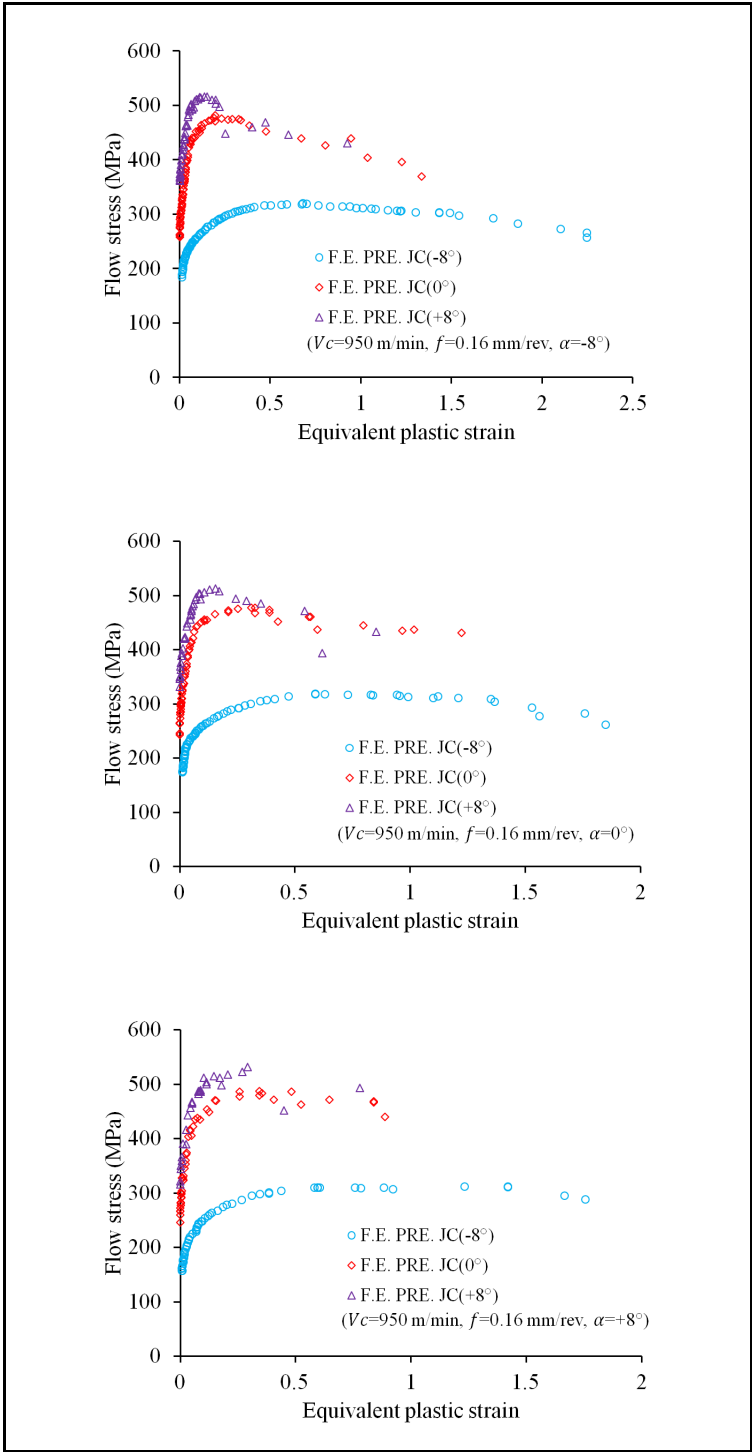


Figure 6-15 Effect of JC sets on the material flow stress during cutting

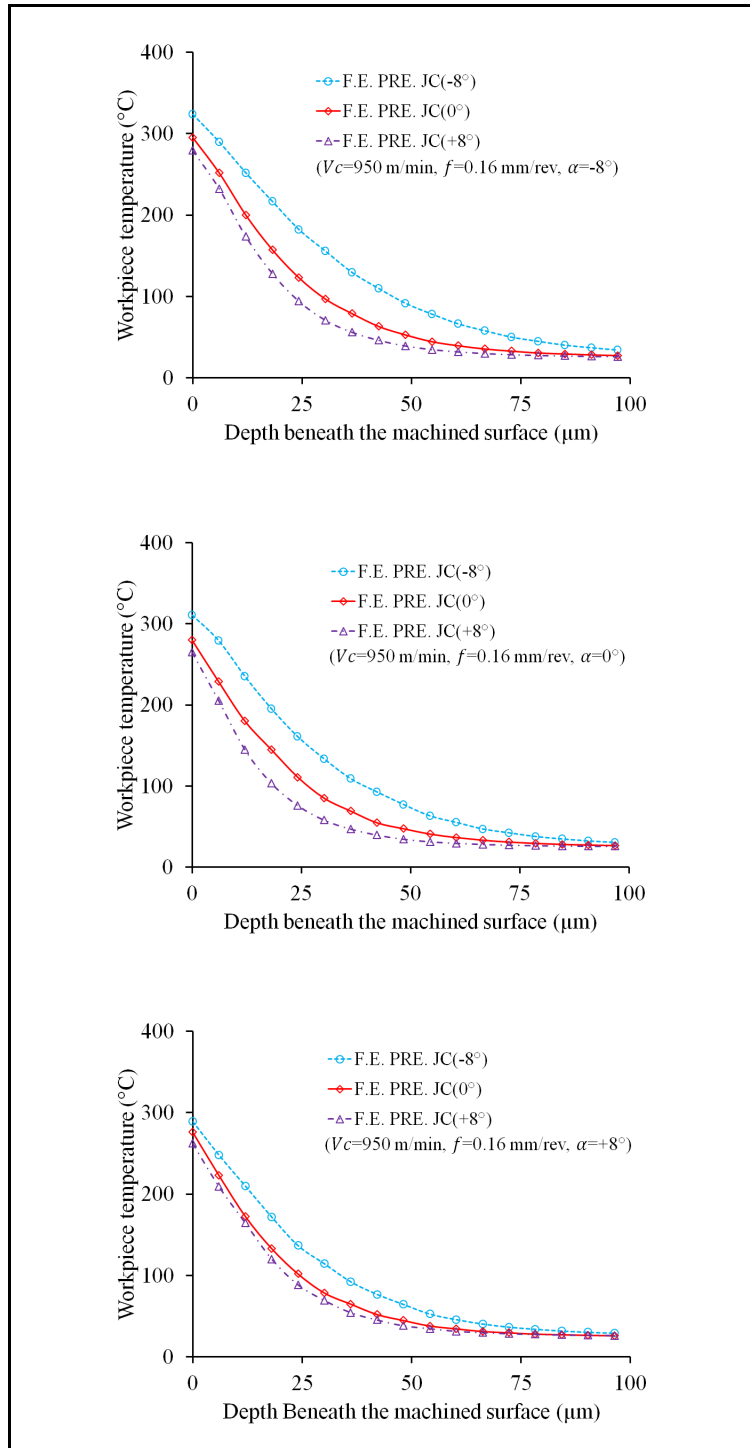


Figure 6-16 Effect of JC sets on temperature beneath the tool-tip during cutting

### 6.6.2 Temperature in the cutting tool

In dry HSM, the heat generation at tool-chip interface plays a crucial role in determining the tool life, as well as, the quality of the machined part. For example, temperature generation at the tool-chip interface affects the contact mechanism by altering the friction conditions, which in turn alters the maximum temperature location, heat partition, and the diffusion of the tool material into the chip (Abukhshim *et al.*, 2006). As a result, an accurate prediction of the temperature distribution in the cutting tool is highly important. The effect of the three sets of JC material constants on the prediction of temperature in the cutting tool was, therefore, investigated.

As mentioned above, the FE flowchart shown in Figure 6-10 was followed to predict the temperature in the cutting tool. The 3D thermal analysis was conducted based on nodal temperatures obtained from a prior 2D thermo-mechanical simulation and collected at tool-chip interface when steady-state conditions are reached (see Figure 6-17). The variation of the maximum temperature at the tool-chip interface is influenced by JC sets and tends to be higher with JC(0°) and JC(+8°). This could be explained by the fact that the flow stresses predicted by JC(0°) and JC(+8°) are higher which results in higher frictional shear stress at tool-chip interface. Therefore, these materials need higher frictional energy for sliding to occur leading to higher heat generation and an increase in temperature. Changing the rake angle from positive to negative values results in a slightly increase in the level of the temperature. Figure 6-18 shows the nodal temperatures along the contact length starting from the tool tip which are obtained under three cutting conditions and for the three sets of JC.

In the light of above considerations, the 3D temperature distribution of the cutting tool is shown in Figure 6-19, after a 0.28 sec of the beginning of the cutting simulation for JC(-8°) with a cutting speed of 950 m/min, a feed of 0.16 mm/rev, and a rake angle of -8°. This time duration corresponds to the experimental cutting time duration under the same cutting conditions. Similar temperature distributions were found with other cutting conditions for JC(0°) and JC(+8°).

As mentioned earlier, the experimental tests were repeated three times to ensure repeatability. The results of the experimental tests are shown in Table 6-7. It is clear that the cutting forces and temperature decrease when the rake angle is changed from  $-8^\circ$  to  $+8^\circ$ . This is due to the drop of the contact pressure and the friction force at the tool-chip interface.

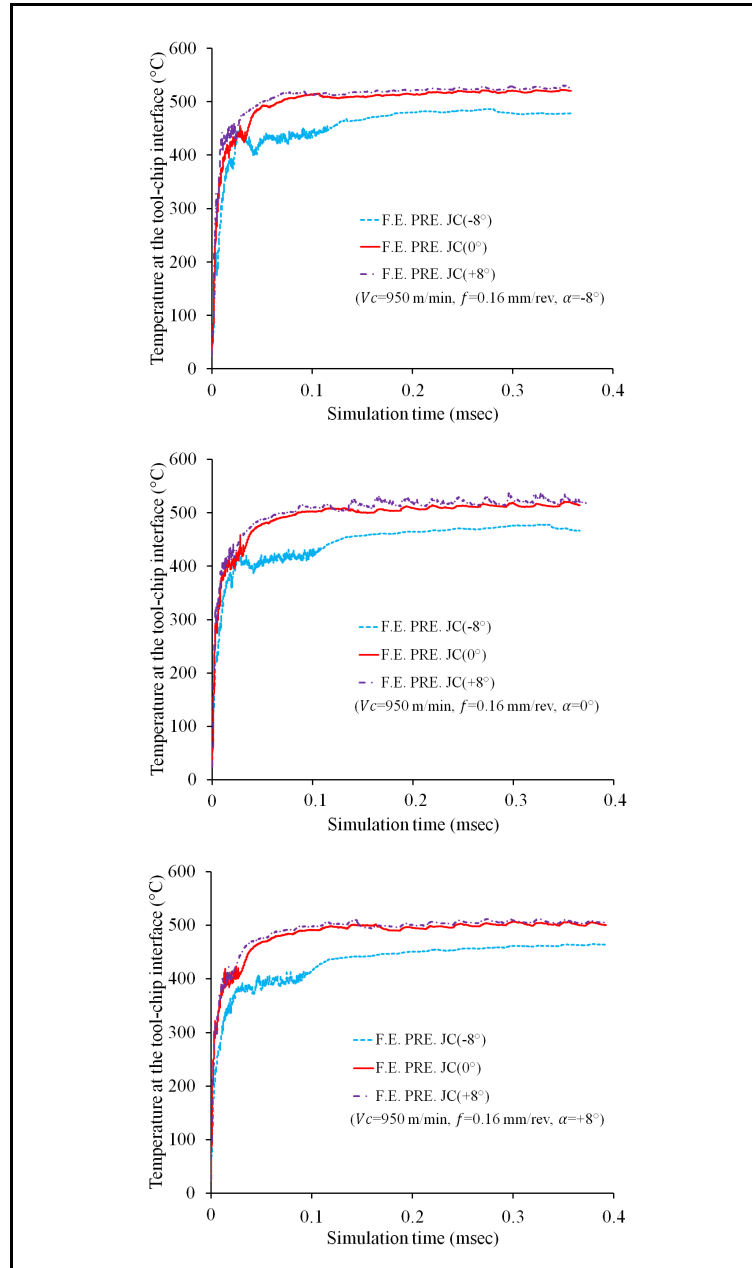


Figure 6-17 Effect of JC sets on temperature at tool-chip interface during cutting

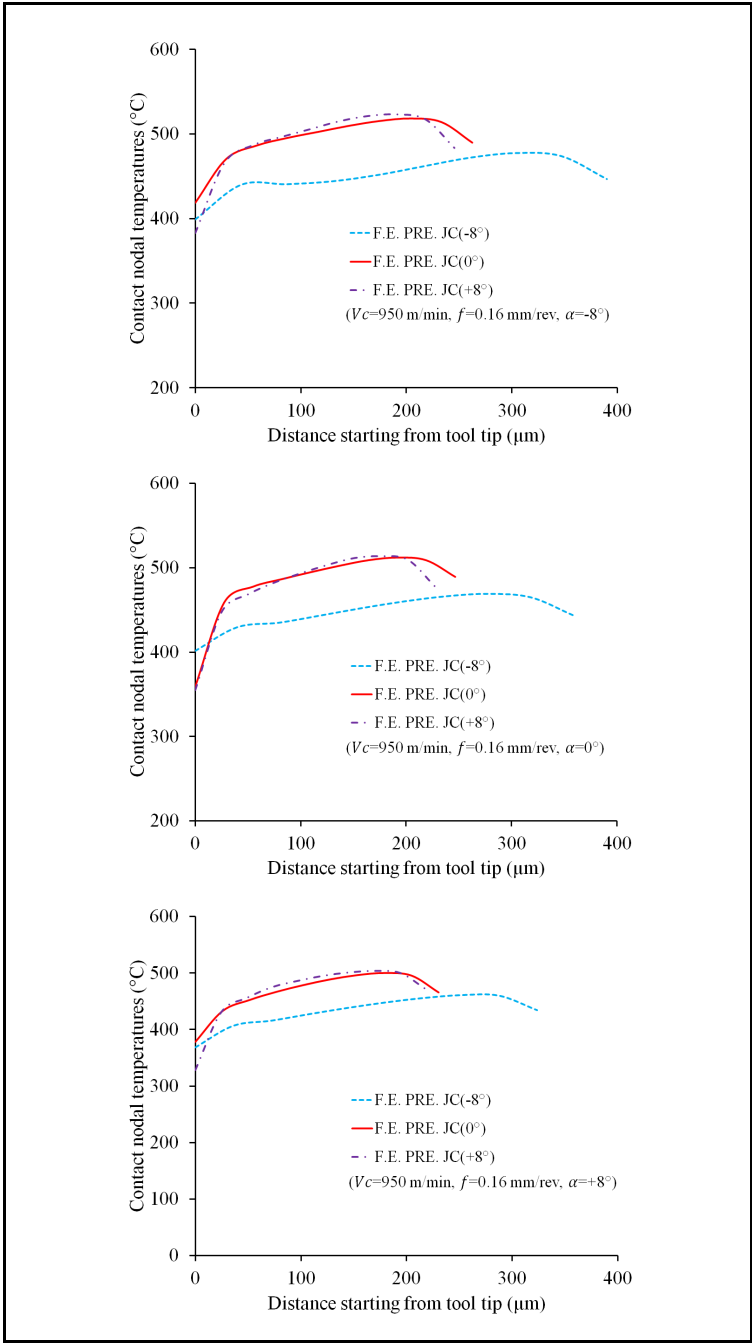


Figure 6-18 Contact nodal temperature coming from 2D thermo-mechanical simulations

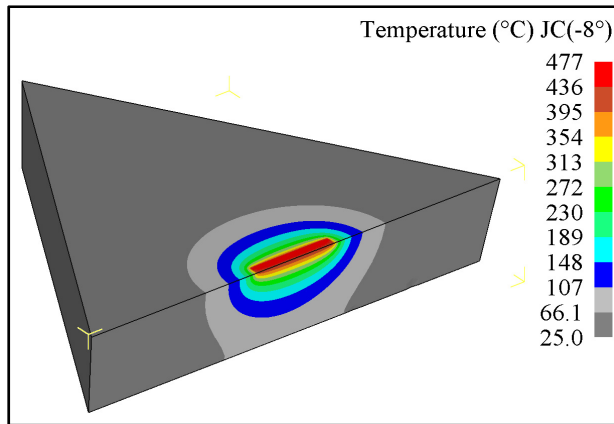


Figure 6-19 Predicted temperature distribution  
( $V_c=950$  m/min,  $f=0.16$  mm/rev,  $\alpha=-8^\circ$ )

Table 6-7 Comparison between experimental results  
( $V_c=950$  m/min,  $f=0.16$  mm/rev)

No. of run	Cutting tool no.	$\alpha$ (deg.)	$F_c$ (N)	$F_f$ (N)	Temperature ( $^\circ\text{C}$ )
1	1	-8	443	174	181
2	1	0	425	132	173
3	1	+8	394	92	167
4	2	+8	390	77	140
5	2	0	427	127	146
6	2	-8	456	181	150
7	3	0	417	128	172
8	3	-8	455	180	176
9	3	+8	388	84	166

In view of the difference in diameter between the hot junction of the used thermocouple and the hole made in the cutting tool (435 versus 940  $\mu\text{m}$ , respectively), it is more convenient to compare the experimentally measured temperatures with the predicted values for three assumed thermocouple positions closer to the rake face, as shown in Figure 6-20. Figure 6-21 gives a clear comparison of the results obtained with the three sets of JC and for three thermocouple positions P1, P2, and P3.

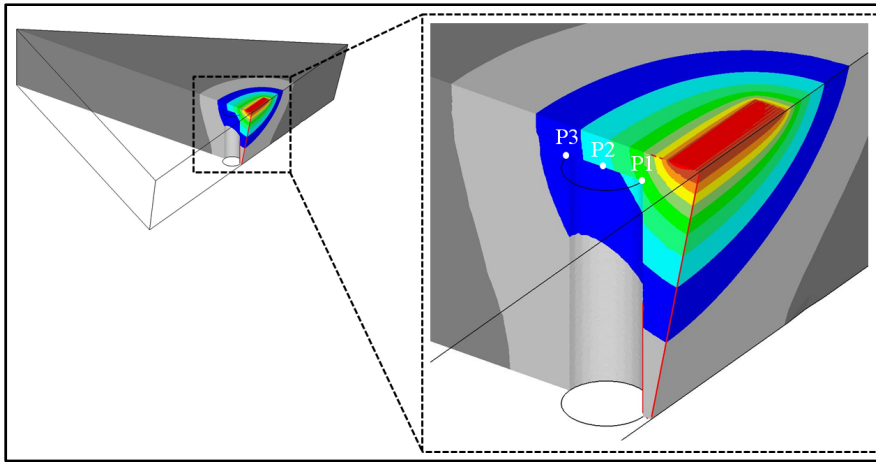


Figure 6-20 Thermocouple positions selected inside the cutting tool

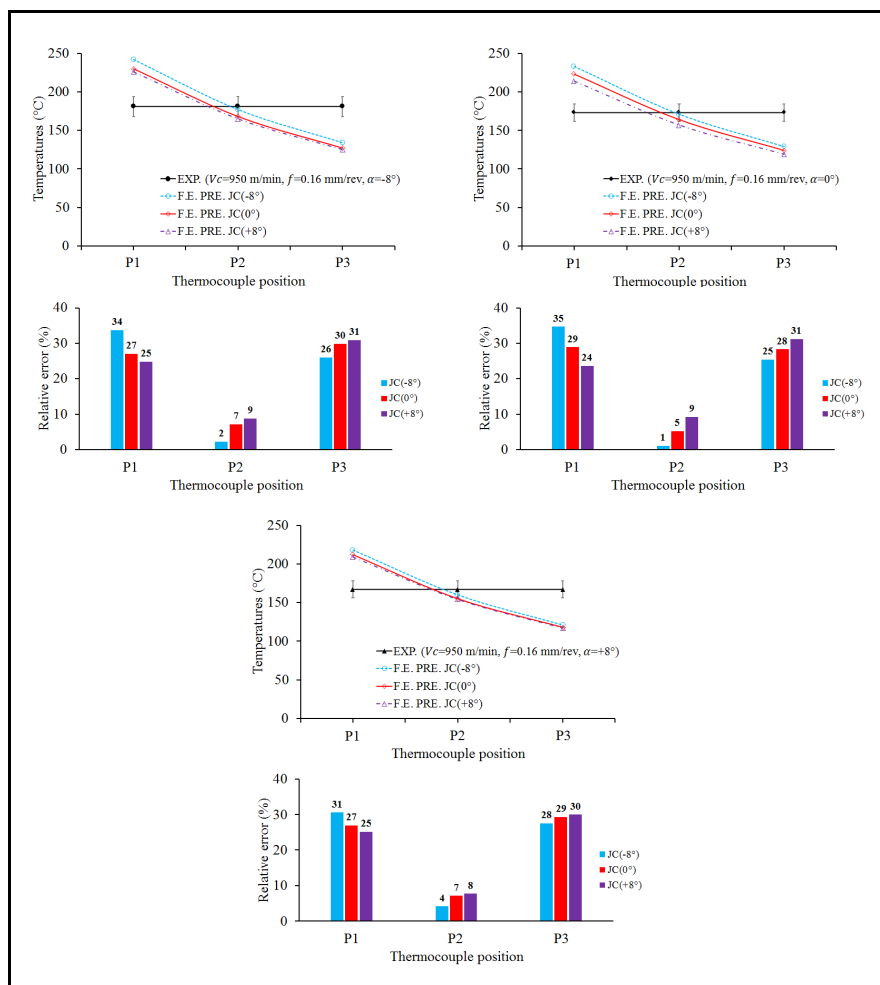


Figure 6-21 Comparison between experimental (EXP.) and predicted (F.E. PRE.) temperatures



It is clear that the best prediction for the three sets of JC is obtained at the thermocouple position P2; the percentage difference with experimental values are 1-4%, 5-7%, and 8-9% for JC(-8°), JC(0°), and JC(+8°), respectively. For the first position of the thermocouple (P1), JC(+8°) gives the better prediction while for the other thermocouple positions (P2 and P3) the better prediction is obtained with JC(-8°). Despite JC(0°) predicts temperatures within [5-30%] for all positions, it still predicts temperatures with reasonable accuracy and closer to the other sets. Obviously, JC(-8°) predicts always higher temperatures despite its lower prediction steady state temperature at tool-chip interface (see Figure 6-17). This could be explained by the fact that the material constants set JC(-8°) exhibits higher tool-chip contact length followed by JC(0°) and JC(+8°), respectively, as reported in (Daoud *et al.*, 2015a). In fact, the longer the tool-chip contact length is, the more heat transfer to the cutting tool is, leading to higher temperature.

Additional comparisons are also made by taking the mean temperature of the three positions (P1, P2, and P3). As can be seen in Table 6-8, differences in average temperatures between the three sets of JC are 0.4-2.7% for JC(-8°), 1.5-3.3% for JC(0°), and 4.1-5.5% for JC(+8°). Based on the average values, the effect of the three sets of JC did not show a significant influence since the difference between the measured temperatures and the predicted average ones are smaller than 5.5% in the three cutting conditions.

Table 6-8 Experimental (EXP.) and predicted (F.E. PRE.) average temperatures

$V_c=950$ m/min, $f=0.16$ mm/rev						
	$\alpha = -8^\circ$		$\alpha = 0^\circ$		$\alpha = +8^\circ$	
	Temperature (°C)	Error (%)	Temperature (°C)	Error (%)	Temperature (°C)	Error (%)
EXP. (tool no. 1)	181	-	173	-	167	-
F.E. PRE. JC(-8°)	184	1.8	178	2.7	166	0.4
F.E. PRE. JC(0°)	175	3.3	170	1.5	162	3.1
F.E. PRE. JC(+8°)	172	4.9	163	5.5	160	4.1

Finally, it is worth noting that the work in (Shi *et al.*, 2010b) showed that the rake angle of 0° results in the largest extent of the central portion of the primary shear zone. Consequently, the cutting with the rake angle of 0° seems to be the most appropriate cutting condition for

the analytical model used in the identification step to better estimate the physical quantities  $(\bar{\epsilon}_{AB}, \dot{\bar{\epsilon}}_{AB}, \bar{\sigma}_{AB}, T_{AB})$  in the primary shear zone.

## 6.7 Conclusions

In this work, a FEM study, using SFTC-Deform 2D and 3D finite element software, was carried out in order to underline the effect of three sets of JC constants on the residual stresses and cutting temperatures. These sets of JC were determined by machining tests at three different rake angles and were then implemented in the commercial FE software DEFORM-2D. Dry machining tests under orthogonal machining conditions were conducted on Al2024-T3 using a disk-Shaped workpiece and an uncoated sharp carbide cutting insert under high cutting speed. Residual stresses and thermal fields predicted by FEM with the three JC sets were compared to those measured by X-ray diffraction and thermocouples techniques, respectively. Results show that a better prediction of the residual stresses is obtained when using JC(0°) while the other sets JC(-8°) and JC(+8°) tend to overestimate and underestimate the predicted ones, respectively. Prediction of cutting temperatures is not much affected by the sets of JC especially when a mean value is considered.

In the light of the obtained results, the material set obtained at 0° rake angle, JC(0°), shows an overall more accurate prediction of residual stresses and temperatures. It is believed that the cutting with the rake angle of 0°, during the identification step, gives a good estimate of the physical quantities  $(\bar{\epsilon}_{AB}, \dot{\bar{\epsilon}}_{AB}, \bar{\sigma}_{AB}, T_{AB})$  in the primary shear zone. This can explain the more accurate predicted results obtained with the material constants identified at 0° rake angle.

## CONCLUSION

This research work was achieved to comprehensively understand the effect of the material constitutive law model on the numerically predicted machining parameters such as cutting forces, chip morphology, tool-chip contact length, machining-induced residual stresses, and cutting tool temperature under high speed machining. Orthogonal dry machining tests using a disk-shaped workpiece and uncoated carbide cutting tool have been carried out to validate the finite element methodology, by comparing the numerically predicted machining parameters to the experimental ones conducted under similar cutting conditions. The Johnson-Cook constitutive law model developed to represent the material behavior at high ranges of strains, strain rates, and temperatures was adopted in this research work.

In the first article, an inverse approach based on response surface methodology (IABRSM) was developed to determine the constants of the selected constitutive law model for finite element simulation of high speed machining. Three aluminum alloys (Al2024-T3, Al6061-T6, and Al7075-T6) commonly used in aircraft applications were considered in the experiments. Since the rake angle is regarded as one of the most critical parameter in metal cutting, specific attention was given to its effect on the constitutive law model. The obtained constitutive law models were then validated and employed to simulate the machining behavior of the three aluminum alloys using DEFORM-2D software.

In the light of the above considerations, it was found that the rake angle has a significant effect on the constitutive models when the inverse approach is considered; therefore, the effect of the obtained constitutive law models at different rake angles on the numerically predicted results requires an investigation. Therefore, in the second article, five sets of Johnson-Cook constitutive law model were determined at five different rake angles based on IABRSM developed in the first article. These five sets were then used in numerical machining model using DEFORM-2D software to investigate their effect on the predicted cutting forces, chip morphology, and tool-chip contact length in the machining simulation of Al2024-T3 alloy.

In the third article, two numerical approaches, namely a 2D thermo-mechanical simulation (DEFORM-2D) and a 3D pure thermal analysis (DEFORM-3D), were used to underline the effect of different rake angle-based Johnson-Cook material constant sets on the machined workpiece residual stresses and the cutting tool temperature distributions. During this investigation, an analysis of the physical quantities within the workpiece has fostered a rich discussion of the physical mechanisms governing the generation of the results.

In the light of obtained results, the major conclusions are summarized as follows:

*a) An inverse approach has been proposed to identify material constitutive law for finite element simulation*

An experimental procedure to determine the constants of the Johnson-Cook constitutive law model was developed based on the response surface methodology coupled to the central composite design.

It was proved that the developed second-order polynomial models can adequately predict the cutting forces and chip thickness for all studied alloys thanks to the high regression coefficients (the coefficient of determination is found to be between 87.77 and 99.30% while the adjusted coefficient of determination is found to be between 69.43 and 98.25%). Based on these results, the effect of the rake angle on the material constants was then investigated. It was shown that the material constant sets obtained with a rake angle of  $0^\circ$  gives the smallest relative difference of flow stress for the three studied alloys as compared to other rake angles. This could be explained by the fact that the rake angle of  $0^\circ$  is the most suitable cutting condition for the analytical cutting models to better estimate the physical quantities in the primary shear zone.

A comparative investigation showed that the material constants sets obtained from the proposed approach predict flow stresses with better accuracy than the ones determined by other methods; the differences between the predicted and experimental flow stresses are

found to be within [14-27%], [16-18%], and [ $\sim$ 27%] for Al2024-T3, Al6061-T6, and Al7075-T6 alloys, respectively. The differences could be attributed to some error sources. The measurements of cutting forces and the chip thickness, the weakness of JC constitutive model to fully predict the complexity of material behaviour, the independent effect of strain, strain rate and temperature on the flow stress assumed in the JC constitutive model, and the analytical cutting models, used to calculate the physical quantities, are based on simplifications and some assumptions which may decrease the models' accuracies.

FEM investigation has also shown an overall good prediction of the measured cutting forces and chip morphologies when using material constants obtained by IABRSM. The predicted tangential forces are within 8.2, 20.5, and 19.3% of experimental values for Al2024-T3, Al6061-T6, and Al7075-T6 alloys, respectively. For the thrust forces, the best prediction is obtained with the Al7075-T6 alloy. However, higher difference is found with the two other alloys. This underestimate of the thrust force could be attributed to the reduced number of elements in the secondary shear zone. Regarding the chip thickness, the predictions are within 1, 23.6% of the measured ones for Al2024-T3 and Al6061-T6 alloys, respectively. In the case of serrated chip geometry for Al7075-T6 alloy, the chip peak thickness and serration frequency are also modeled very well. A possible reason that explains the better results is that the material constants sets obtained from the proposed approach were identified directly from the cutting tests, but this is not the case for the other sets of the material constants

*b) The effects of material constant sets used in JC constitutive law on the numerically predicted cutting forces, chip morphology, tool-chip contact length, residual stresses, and the temperatures*

A comparative investigation using a series of finite element modeling simulation showed that the cutting forces, chip morphology, tool-chip contact length, residual stresses distributions are more sensitive to the different rake angle-based Johnson-Cook material constants than the temperature distributions are.

Referring to the tangential force, better prediction was obtained with JC(0°), JC(+5°), and JC(+8°). The predicted tangential forces are within 7.9, 9, and 10.2 % of the experimental values with JC(0°), JC(+5°), and JC(+8°), respectively. By taking the average relative error for each set of JC, the JC(0°), JC(+5°), and JC(+8°) have practically similar values of 4, 4.36, and 4.93 %, respectively. This could be attributed to the fact that the flow stresses predicted by these sets of JC are close to each other. Oppositely, higher relative errors are encountered when evaluating the tangential forces with material constant set JC(-8°) and JC(-5°) because of the lower flow stress predictions. The predictions of thrust force are less accurate for all five JC sets due to numerical issues. The average error related to the thrust force was found to be 27.8 and 30.6 % for JC(-8°) and JC (0°), respectively.

Results showed that the chip thickness is more sensitive to the JC material constant sets than the cutting forces. JC(0°) gives a best prediction with smaller average relative error of 4.26 % and large discrepancies in chip thickness were obtained using material constant set JC(-8°) with a maximum relative error of about 66 %. The worst prediction of chip thickness with JC(-8°) could be explained by the fact that JC(-8°) has lower yield strength coefficient A and higher hardening coefficient n. In the case of a serrated chip formation, the chip thickness and the serration frequency are relatively well predicted when the material constants JC(0°) is used.

Best prediction of tool-chip contact length was obtained with JC(-5), followed by the material constants set JC(0°), JC(+5°), and JC(+8°), respectively, with an average difference of 3.64, 9.38, 13.4, and 12.52 %, respectively.

The results show that a better prediction of the residual stresses is obtained with JC(0°) while the other sets of JC(-8°) and JC(+8°) tend to overestimate or underestimate the measured residual stresses, respectively. This trend could be mainly attributed to the different equivalent plastic energies obtained with these sets of JC.

As far as the temperature of the cutting tool is concerned, the average values of the temperatures of the cutting tool for each studied set of JC was considered in order to evaluate the best prediction. Based on these average values, the effect of the three sets of JC was not significant influence since the difference between the measured temperatures and the predicted average ones are less than 5.5% with the three cutting conditions.

According to the above results, the set of material constants obtained at  $0^\circ$  rake angle gives an overall more accurate prediction while higher relative errors are encountered with the set obtained at  $-8^\circ$  rake angle.





## **CONTRIBUTIONS**

The main contributions accomplished by the current research work can be summarized as follows:

- Development of an efficient approach to identify Johnson-Cook constitutive law model for finite element simulation of aluminum alloys. The developed approach is an improvement of the existing inverse method with a reduced number of experiments. The developed approach was then used to investigate a large number of cutting conditions within fixed ranges of cutting parameters such as cutting speed, feed rate, and rake angle.
- Study the influence of the rake angle on the constitutive model when the inverse approach is considered.
- Demonstration of the role of different material constant sets in controlling the predicted results. This will be highly useful to understand how different sets of material constants identified at different rake angles lead to distinguishable results for the same cutting conditions.
- Accomplishment of an understanding of the physical mechanisms involved in the generation of the cutting parameters and how the different sets of material constants alter the plastic deformation and heat generation within the workpiece.



## RECOMMENDATIONS

Future research work related to this study will concentrate on the following aspects:

- The identification approach developed in the first part of this study is based on orthogonal machining experiments. Other cutting tool materials can be investigated. A study of their impact on the constitutive law material constants and consequently on the numerically predicted cutting process parameters will lead to general guidelines on machining simulations.
- The examination of the effect of rake angle on other constitutive law materials. It is desirable to conduct similar investigation on the other constitutive law models other than that of JC used in this study.
- Extension of the identification approach to a wider range of strains and strain rates, by combining the proposed approach with dynamic tests. This may provide more reliable predictions of finite element modeling of machining.
- Extension of the current finite element simulation to model complex machining processes such as milling, turning, drilling, and distortion of aerospace components. This will help to verify the reliability of the obtained material constants.
- Using the different sets of JC constitutive law obtained in this work in an analytical model to predict cutting forces, temperatures, residual stresses, etc.
- The developed models need to be improved in order to be more representative.
- Conduction of residual stress measurements, we suggest to use high speed camera coupled with circularity profile measurement in order to confirm the machined surface zone which is representative of the cutting test.



## APPENDIX I

### **Finite element correction method for in-depth residual stress measurement obtained by XRD**

As mentioned in chapter 2, corrections to the residual stress measurements due to the removed volume of material were made using finite element method. Correction was carried out in three steps. First, the correction matrix was determined using a 3D finite element model and based on a very low gradient tensile stress profile induced numerically by artificial thermal loads. Then, the obtained correction matrix was used to correct a high in-depth gradient stress profile induced numerically for validation purpose. Finally, the correction matrix was employed to calculate the needed corrections on the real residual stress measurements by XRD.

#### *Obtaining the correction matrix*

An elastic finite element of the studied workpiece (disk in shape having an outer diameter of 75 mm and an inner diameter of 16 mm with a thickness of 3.14 mm) was modeled using ANSYS software. The 3D model was meshed using 8 nodes solid brick elements with a dense mesh in the polishing zone. We used the symmetry properties for both the workpiece and polishing zone in order to reduce CPU time, as shown in Figure A I-1.

The proper element size is determined based on convergence study as shown in Figure A I-2. Based on this investigation the element size of 5  $\mu\text{m}$  in depth and 200  $\mu\text{m}$  in width are retained.

In order to minimise the errors in the determination of the correction matrix, initial rotationally symmetric and quasi-uniform stresses were induced using thermal loads (see Figure A I-3). Elastic properties of the workpiece (Al2024-T3) presented in Table 4-2 were used. The polishing zone dimensions are 2 mm (width) by 2.5 mm (length).

The equivalent Von Mises stress was used to verify the purely elastic relaxation assumption after each polishing step, as shown in Figure A I-4. In is clear that the equivalent Von Mises stress remains below the tensile yield strength of the studied aluminum alloy and consequently there is no risk of plasticity.

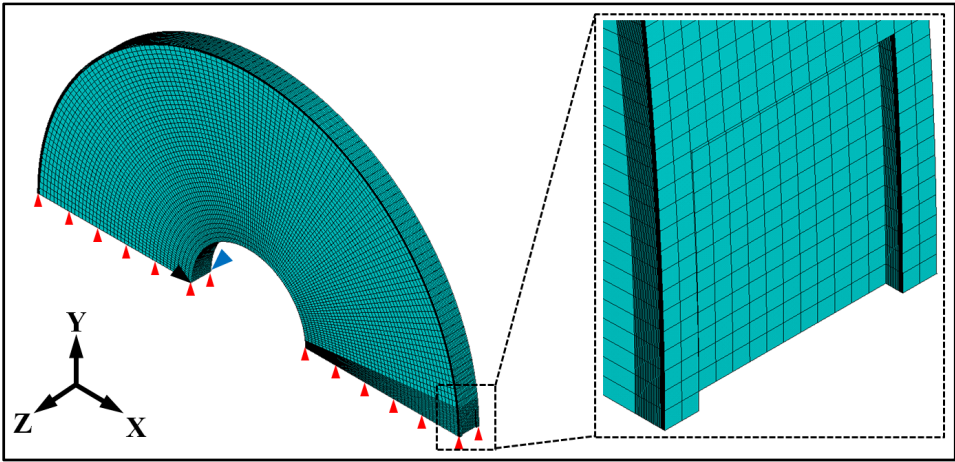


Figure A I-1 Boundary conditions with polishing zone visualization

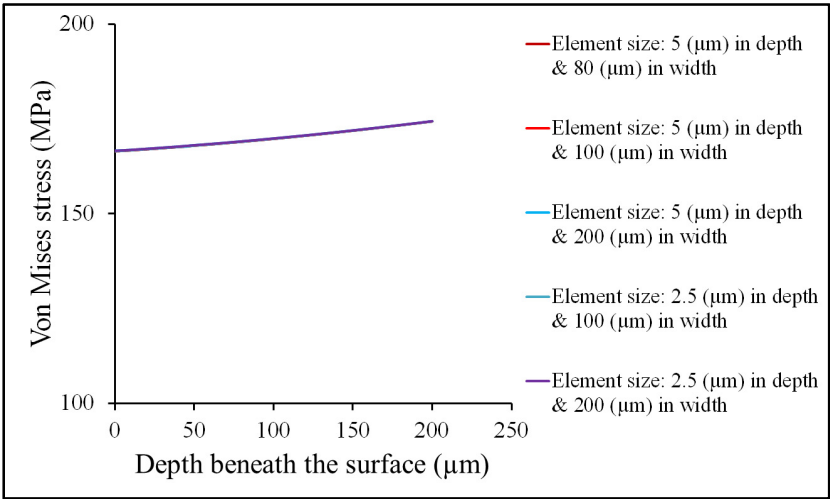


Figure A I-2 Mesh convergence within the disk-shaped workpiece

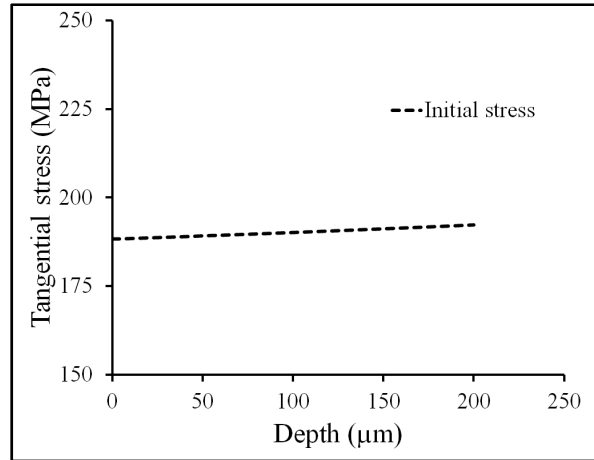


Figure A I-3 Residual stress profile used to determine the correction matrix

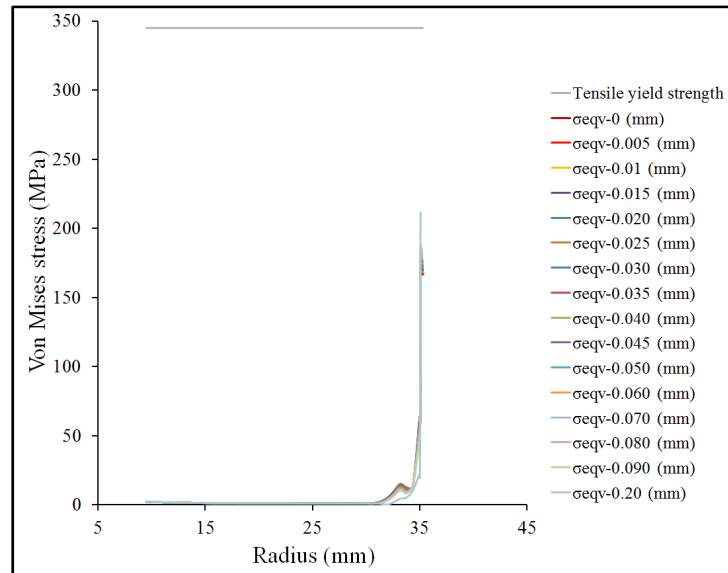


Figure A I-4 Distribution of the equivalent Von Mises stress for the nodes situated in the center of rectangular polishing zone

#### *Validation of the correction matrix*

As mentioned above, the correction matrix is independent of the stress distribution used to calculate it. This is because the redistribution of residual stresses is assumed to remain elastic after the material removal; therefore, the obtained correction matrix was used to calculate the

corrections on a numerically stress profile induced in the same geometry of workpiece. Figure A I-5 shows the results of the initial (before layer removal), the measured (after each layer removal), and the corrected tangential stress profiles. This figure shows also a comparison between the analytical method (Moore and Evans) and the finite element one.

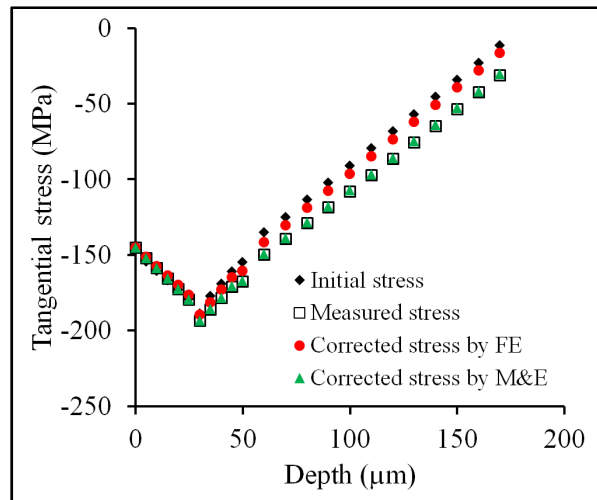


Figure A I-5 Correction of high in-depth gradient stress profile

#### *Correction residual stress profile obtained by XRD*

Figure A I-6 shows the residual stresses obtained experimentally with the cutting conditions listed in Table 6-4 and the corrected ones. It is clear that there is no difference between the measured profiles and the corrected ones. In fact, removing thin layers from a large workpiece or from any workpiece geometry in which no significant stresses are present, correction will be insignificant (Prevey, 1986).



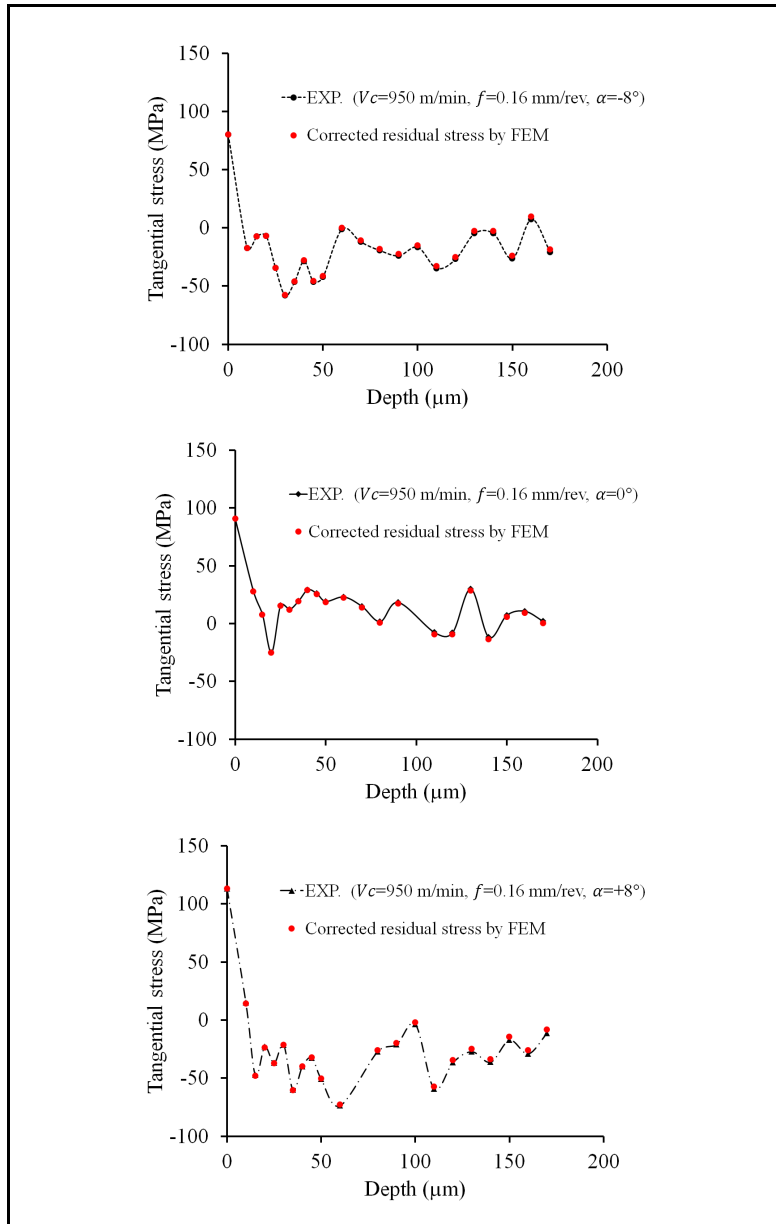


Figure A I-6 Correction of the residual stress measurements



## APPENDIX II

### Determination of the physical quantities in the primary shear zone

In this section, we present the relationships allowing the determination of the strain, the strain rate, the flow stress, and the temperature generated in the primary shear zone during the machining tests.

By rewriting the Oxley model in the function of input data (cutting forces and chip thickness), the physical quantities in the primary shear zone can be described as follows:

The shear angle can be estimated from Equation (A II-1):

$$\phi = \tan^{-1} \left[ \frac{\frac{f}{t_c} \cos \alpha}{1 - \frac{f}{t_c} \sin \alpha} \right] \quad (\text{A II-1})$$

The expression of the equivalent strain on the primary deformation zone is given by:

$$\bar{\epsilon}_{AB} = \frac{\cos \alpha}{2 \sqrt{3} \sin \phi \cos(\phi - \alpha)} \quad (\text{A II-2})$$

The equivalent strain rate on the primary deformation zone is expressed as follows

$$\dot{\bar{\epsilon}}_{AB} = \frac{2 V_c \cos \alpha}{h \sqrt{3} \cos(\phi - \alpha)} \quad (\text{A II-3})$$

where  $h$  is the thickness of the primary shear zone and can be estimated as (Tounsi *et al.*, 2002):

$$h = 0.5 \times f \quad (\text{A II-4})$$

The equivalent flow stress at the primary zone is given by

$$\bar{\sigma}_{AB} = \sqrt{3} \sin \phi \frac{(F_c \cos \phi - F_f \sin \phi)}{f w} \quad (\text{A II-5})$$

Hastings *et al.* (1980) used in their analysis a modified Boothroyd's temperature model (Boothroyd et al., 1963). In this model, the average temperature at the primary shear zone can be obtained by:

$$T_{AB} = T_{room} + \frac{(1 - \eta)(F_c \cos \phi - F_f \sin \phi) \cos \alpha}{\rho C_p f w \cos(\phi - \alpha)} \quad (\text{A II-6})$$

where  $\eta$  the ratio of the heat flowing into the workpiece and can be obtained as:

$$\eta = 0.5 - 0.35 \times \log(R_T \tan \phi) \text{ for } 0.04 \leq R_T \tan \phi \leq 10 \quad (\text{A II-7a})$$

$$\eta = 0.3 - 0.15 \times \log(R_T \tan \phi) \text{ for } R_T \tan \phi > 10 \quad (\text{A II-7b})$$

$$R_T = \frac{\rho C_p f V_c}{K_{th}} \quad (\text{A II-8})$$

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